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A PROCEDURE FOR USING THE COMPUTER CODE OF
NASA CR-710 TO OBTAIN THE TWO-D AXISYMMETRIC
FLOWS BEHIND BLADE ROWS IN WET VAPOR TURBINES

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FOREWORD

The work described in this report was performed under NASA Contract NAS 7-390, "Basic Investigation of Turbine Erosion Phenomena". The work was done under the supervision of Mr. W. D. Pouchot of the Systems and Technology Department of the Westinghouse Astronuclear Laboratory. Mr. L. G. Hays of Jet Propulsion Laboratory is the NASA Program Manager.

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I. INTRODUCTION

This report is designed to be used in conjunction with NASA CR-710 (Reference 1) to give the user sufficient information so as to be capable of utilizing the NASA Performance Computer Code for Axial Flow Turbines as modified at WANL. The modified code is written entirely in FORTRAN IV for the CDC 6600 computer. But the code should be capable of being used with appropriate control cards on any computer having at least 32 K of core storage.

The following sections of the report give: the applicability and modifications made from the original code, definitions of the input and output nomenclature, a method for making the code input applicable for wet vapor turbines, suggestions for further possible future modifications, three sample problems illustrating the usage of the code, a FORTRAN listing of the entire code, and control cards showing proper deck setup. No attempt is made to discuss the method of calculation of turbine performance or to give computer flow diagrams since these topics are adequately covered in Reference (1). The modifications made to the code do not significantly change the original program logic or capability. These modifications for the most part were necessary to enable the code to accurately calculate wet vapor turbine performance. Ideal gas turbines can still be analyzed as well as air breathing fossil fuel burning turbines for which the code was originally primarily designed.

II. INTENT OF CODE

A. Applicability of Code and Limiting Assumptions

The principal purpose of the original code as written by E. E. Flagg⁽¹⁾ is to provide a complete performance map of axial flow turbines suitable for use in air breathing fossil fuel fired jet engines. In the process of accomplishing this end, the code calculates the two-dimensional bulk flow conditions fore and aft of the turbine rows.

1. Description and Scope of Modified Code

- (a) Axial flow turbines.
- (b) Up to 8 stages.
- (c) Up to 6 radial sectors (although only 5 are usually used for reasons of symmetry).
- (d) Each sector is a quasi-one-dimensional element with the properties at the radial centers of these sectors being joined; utilizing simple radial equilibrium at the stator and rotor exits.
- (e) Semi-perfect gas properties (gas constant and specific heat ratio) are assumed and are input at the entrance and exit of each blade row. Provision is also made to simulate changes in gas flow rates at the entrance and exit of each blade row. Energy balance effects are simulated by changing the values of the gas constant and specific heat ratio.
- (f) The turbine geometry may be either input as a passage distributed area (SPA and RPA)* or as effective exit vector flow angles (SDEA and RDEA).

* Nomenclature defined in Section III-A of this report.

The assumption that the effective exit flow angles are approximately equal to the design blade exit angles is usually valid. Mandatory inputs are the diameters of the root (DR) and tip (DT) for the entrance and exit of each blade row and the stator and rotor design inlet angles (SDIA and RDIA) for each of the radial sectors.

- (g) Even though there are two subroutines (LØSS 1 and LØSS 2) which are capable of calculating losses by a total pressure loss coefficient method, the values for the coefficients of the series expansion are not generally known. See page 11 of NASA CR-710. The standard method is to input the values of optimum recovery coefficients for stator and rotor (SREC and RREC) together with exponents to be used in the event of both negative and positive (EXPN and EXPP) incidence. See page 10 of NASA CR-710 for equations used.
- (h) Separate cases may be run for various turbine speeds by merely changing the RPM and indicating that is a change case (STGCH = 0.0).
- (i) The FORTRAN IV code calculates a performance map for the case of a given turbine at a particular RPM by in effect varying the exit back pressure. The output for each "iteration" (i.e., value of back pressure) gives flow rates, velocities, flow angles, temperatures, pressures, densities, Mach numbers, efficiencies, and work done both for an overall stage output and also row-by-row output for each of the radial sectors. An exact choke point is found during the calculation of the performance map and the turbine back pressure is effectively further reduced until the discharge annulus area is choked at the pitchline sector (assuming AACCS = 1.0). A single performance point can be obtained by simply setting all pressure ratio increments (DELC, DELL, and DELA) to zero. This is the usual case when fixed operating conditions are known at design.

- (j) The gas flow at the entrance to the first stator is assumed to have uniform radial temperature, pressure, and velocity. The flow is further assumed to be exactly aligned with the turbine axial direction (i.e., no tangential velocity component).

B. Modifications to Code

As stated previously, as originally programmed the code was principally intended for analysis of JP-4 burning, air breathing jet engines. Internal to the code is a subroutine for calculating the thermodynamic properties of reacted JP-4-air mixtures. It also had a capability to input thermodynamic properties which was extended as required by the method used in determining the performance of wet vapor turbines. It was decided that the thermodynamic properties fore and aft of each blade row would be inputted in terms of representative values for the particular working fluid and its state. The variables to be input would be the ratio of specific heats at constant pressure to that at constant volume and Boyles and Charles Law gas constant. The internals of the program are then used to calculate effective specific heat and various other effective thermodynamic properties.

The following modifications were made in the code:

1. Wherever the Boyles and Charles gas law constant R_G appeared in the code, it was replaced by a two-dimensional variable $RV(I, K)$ with proper choice of axial blade position I and stage number K to correspond to the location in the turbine for which the calculation is being performed.
2. A change was made in the input NAMELIST format to allow reading in of a variable RV . Also a modification was made to read in reference values for the gas constant, temperature, pressure, and specific heat ratio all at standard sea level conditions. Formerly the code contained these values for air internally in a DATA statement. But since gases other than air will be used, it was thought useful to include a capability for inputting these values for each case rather than requiring a recompilation whenever a different working fluid was used.

3. The output was expanded to print out the values for the flow, γ (ratio of specific heats), gas constant, and RWG (the ratio of the flow at a particular station to turbine inlet flow). To insure that these variables were being properly handled within the code, decreasing values of γ , RV, and RWG were fed in. The output was found to be consistent after a slight change in the logic.
4. Since values for γ and RV are now fed in for all cases, the subroutines to calculate γ , RG, and C_p 's are superfluous since they would never be called upon. If by inadvertently omitting the inputting of γ and/or RV and subsequently a subroutine for calculating its value is entered, then an error message was added which would print out the words "SUBROUTINE (____) HAS BEEN CALLED UPON" followed by a string of asterisks so that attention would be immediately drawn to the error. The (____) is filled in by the name of the subroutine being called. After the error message is printed out, the calculation is allowed to proceed using properties for air, water, and JP-4 fuel.
5. On page 193 of NASA CR-710 the statement:

$$21 \quad PTP(I, K + 1) = PTBAR(K) * ((TTRA(I, K)/TTBAR(K))^{**E3} \quad ST2A \ 153$$
 was found to be incorrect and should read:

$$21 \quad PTP(I, K + 1) = PTBAR(K) * (TT2A(I, K)/TTBAR(K))^{**E3}$$
6. On page 208 of NASA CR-710 the statement:

$$AS0H = SQRT(GAM(I, K) * G * RG * STTS0(L)) \quad INST \ 175$$
 was found to be incorrect and should read:

$$AS0H = SQRT(GAM(1, K) * G * RG * STTS0(L))$$
7. Any cards from the original code which had to be removed rather than modified were denoted by a comment card with the words "CARD DELETED" followed by a string of asterisks.

8. As an aid in debugging a computer run, an option was added to allow the printout of when entry and exit was made from each subroutine. This enables the user to examine the program logic as an aid in determining where discrepancies occur. This option is not recommended for other than debugging runs since a large amount of output results.

III. NOMENCLATURE FOR INPUT AND OUTPUT OF MODIFIED CODE

A. Input Definitions*

1. "TRUE" or "FALSE" card depending on whether or not a listing of when an entrance and exit is made from each subroutine is desired. This card is inputted only once per case.
2. Two heading cards of 60 characters each inputted only once per case.
3. Constants inputted only once per case:

<u>Code Name</u>	<u>Definition</u>	<u>Units</u>
STAGE **	Stage identification number	---
STGCH	Flag indicating whether following data is for the basic case (1.0) or for a change case (0.0)	---
TTIN	Turbine inlet total temperature	$^{\circ}\text{R}$
PTIN	Turbine inlet total pressure	psia
WAIR	Water to air ratio (not used in modified code); should be input as 0.0	---
FAIR	Fuel to air ratio (not used in modified code); should be input as 0.0	---
PTPS	Pitchline pressure ratio (total to static) across first stator for 0 th calculation. This ratio is incremented by DELC, DELL, or DELA for next calculation	---
DELC	First try at increment to PTPS	---
DELL	Increment to PTPS after first stator has critical flow and also when choke iteration is complete	---
DELA	Increment to PTPS when last rotor is choked	---
STG	Number of stages in turbine (8 maximum)	---
SECT	Number of radial sectors (6 maximum)	---

* Refer to Standard Option Input Sheet (page 11).

** Must be input every time new stage data is read in.

<u>Code Name</u>	<u>Definition</u>	<u>Units</u>
EXPN	Exponent of cosine term for negative incidence used in calculating an inlet recovery factor (see page 10 of Reference 1)	---
EXPP	Exponent of cosine term for positive incidence used in calculating an inlet recovery factor (see page 10 of Reference 1)	---
PAF	Profile averaging fork (either 0.0, 1.0, or 2.0); gives the next stage inlet conditions for either: uniform (0.0) at the average value of the preceding stage, or the radial sector profiles (1.0) of pressure and temperature of the preceding stage, or a third option which keeps the exit total temperature radial profile and "smooths" (2.0) the exit total pressure profile from the preceding stage	---
SLI	Stage loss indicator (0.0 means that recovery, efficiency, and flow coefficients are inputed for each stage; 1.0 means that they are inputed only once and are assumed constant throughout the turbine)	---
AACS	Discharge annulus area choke stop which is the maximum limit for the turbine exit axial Mach number at the pitch-line sector. This code will continue to decrease the back pressure until this limit is reached (assuming DELC, DELL, and DELA \neq 0.0)	---
RPM	Turbine speed	RPM
VCTD	Vector diagram interstage output (either 0.0 for overall stage performance output only or 1.0 for row-by-row sector performance in addition to overall stage output printout)	---
RSL	Gas constant at sea level standard conditions	ft lb/lb °R
TSL	Standard temperature at sea level = 518.688	°R
PSL	Standard pressure at sea level = 14.696	psia
GAMSL	Specific heat ratio at sea level standard conditions	---
ENDSTG	0.0 if more stage data to follow; 1.0 if last stage data has been read in	---
ENDJØB	0.0 if more cases to follow; 1.0 if all data for all cases has been input	---
PCNH	Percent station height distribution (example: if 5 equal (in height) radial sectors were desired, then PCNH = 0.2, 0.2, 0.2, 0.2, 0.2)	---

4. Axial station input for each stage (stations 0, 1, 1A, 2, and 2A)

<u>Code Name</u>	<u>Definition</u>	<u>Units</u>
RG	Gas constant	ft lb/lb °R
GAMG	Specific heat ratio	---
DR	Diameter of root or hub of turbine	in
DT	Diameter of tip of turbine	in
RWG	Ratio of station flow to turbine inlet flow	---

5. Stator radial distributions for each stage (hub to tip sectors)

<u>Code Name</u>	<u>Definition</u>	<u>Units</u>
SDIA	Stator design inlet angle	(° from axis)
SDEA	Stator effective exit flow angle — should not be input if SPA is input	(° from axis)
SREC	Stator optimum recovery coefficient ($\eta_{sr\ opt}$)	---
SETA	Stator efficiency coefficient (η_s)	---
SCF	Stator flow coefficient (C_{fs})	---
SPA	Stator passage area per unit height — should not be input if SDEA is input	in ² /in
SESTH *	Stator ratio of exit blade height to throat height	---

* Only a single value is input.

6. Rotor radial distributions for each stage (hub to tip sectors)

<u>Code Name</u>	<u>Definition</u>	<u>Units</u>
RDIA	Rotor design inlet angle	(^o from axis)
RDEA	Rotor effective exit flow angle — should not be input if RPA is input	(^o from axis)
RREC	Rotor optimum recovery coefficient ($\eta_{rr\text{opt}}$)	---
RETA	Rotor efficiency coefficient (η_r)	---
RCF	Rotor flow coefficient (C_{fr})	---
RPA	Rotor passage area per unit height — should not be input if DREA is input	in ² /in
RTF	Rotor test factor used to represent the non-uniform work extraction due to blade end effects	---
RERTH *	Rotor ratio of exit blade height to throat height	---

* Only a single value is input.

WANL MODIFIED
TURBINE COMPUTER PROGRAM
STANDARD OPTION
INPUT SHEET

Start All Input Cards in Column 2

Subroutine Entry and Exit Listing Option (TRUE or FALSE)

Name (Comment Information)

Title (Comment Information)

\$DATAIN STAGE = ,

STGCH=

TTIN= , PTIN= , WAIR= , FAIR= ,

PTPS= , DELC= , DELL= , DELA= ,

STG= , SECT= , EXPN= , EXPP= ,

PAF= , SLI= , AACS= , RPM= ,

VCTD= , RSL= , TSL= , PSL= ,

GAMSL= , ENDSTG= , ENDJØB= ,

INLET RADIAL PROFILE

PCNH(1)= , , , , ,

AXIAL STATIONS

	STA. 0	STA. 1	STA. 1A	STA. 2	STA. 2A
RG(1)=	/	/	/	/	/
GAMG(1)=	/	/	/	/	/
DR(1)=	/	/	/	/	/
DT(1)=	/	/	/	/	/
RWG(1)=	/	/	/	/	/

STATOR RADIAL DISTRIBUTIONS

	ROOT	PITCH	TIP
SDIA(1)=	/	/	/
SDEA(1)=	/	/	/
SREC(1)=	/	/	/
SETA(1)=	/	/	/
SCF(1)=	/	/	/
SPA(1)=	/	/	/
SESTH=	/		

ROTOR RADIAL DISTRIBUTIONS

	ROOT	PITCH	TIP
RDIA(1)=	/	/	/
RDEA(1)=	/	/	/
RREC(1)=	/	/	/
RETA(1)=	/	/	/
RCF(1)=	/	/	/
RPA(1)=	/	/	/
RTF(1)=	/	/	/
RERTH=	/		

ENDSTG= , ENDSTG=1.0 IF LAST CASE
ENDJØB= \$ ENDJØB=1.0 IF LAST STAGE

B. Output Definitions

1. Station Nomenclature

The axial station numbers (0, 1, 1A, 2, and 2A) following a parameter refer to the following designations:

Station Number	0	1	1A	2	2A
Definition	Stator Inlet	Stator Exit	Rotor Inlet	Rotor Exit	Next Stage Stator Inlet

Also see Figure IV-1 on page 21 for further clarification of terminology.

In the stage and overall performance output printout several parameters are given in terms of the equivalent parameter referenced to standard sea level conditions. This provides a common basis for comparison of performance maps for different turbine cases.

2. Stage Performance Parameters

<u>Symbol</u>	<u>Definition</u>	<u>Units</u>
TTBAR 0	Stage average inlet total temperature	$^{\circ}\text{R}$
PTBAR 0	Stage average inlet total pressure	psia
WG 0	Stage inlet total weight flow	lb/sec
DEL H	Stage enthalpy drop (energy output)	BTU/lb
WRT/P	Stage corrected weight flow function	$(\text{lb/sec}) (^{\circ}\text{R}/\text{psia})^{1/2}$
DH/TTBAR0	Stage energy function	$\text{BTU/lb } ^{\circ}\text{R}$
N/RT	Stage corrected speed	$\text{RPM}/(^{\circ}\text{R})^{1/2}$
ETA TT	Stage total to total efficiency	---
ETA TS	Stage total to static efficiency	---
ETA AT	Stage total to axial total efficiency	---
PT0/PSI	Stator total to static pressure ratio at pitchline	---

<u>Symbol</u>	<u>Definition</u>	<u>Units</u>
PTBAR0/PTBAR2	Stage average total to total pressure ratio	---
PTBAR0/PS2	Stage average total to pitchline static pressure ratio	---
PTR2/PS2	Rotor exit relative total to static pressure ratio at pitchline	---
TTBAR2/TTBAR0	Stage average total to total temperature ratio	---
TTR1A/TTBAR0	Rotor inlet pitchline relative total to stage inlet average total temperature ratio	---
WG 1	Stator exit total weight flow	lb/sec
PS 1A	Rotor inlet static pressure at pitchline	psia
TTR 1A	Rotor inlet relative total temperature at pitchline	°R
PTR 1A	Rotor inlet relative total pressure at pitchline	psia
WG 1A	Rotor inlet total weight flow	lb/sec
PS 2	Rotor exit static pressure at pitchline	psia
TTBAR 2	Stage exit average total temperature	°R
PTBAR 2	Stage exit average total pressure	psia
WG 2	Rotor exit total weight flow	lb/sec
WG 2A	Next stage stator inlet total weight flow	lb/sec
UP/VI	Wheel speed to isentropic velocity ratio at pitchline	---
UR/VI	Root wheel speed to pitchline isentropic velocity ratio	---
PSI P	Kinetic energy loading parameter at pitchline	---
PSI R	Kinetic energy loading parameter at root	---
RX P	Reaction ratio at pitchline	---
RX R	Reaction ratio at root	---
ALPHA 0	Stator inlet gas angle at pitchline	°
I STATOR	Stator inlet incidence angle at pitchline	°
BETA 1A	Rotor inlet gas angle at pitchline	°

<u>Symbol</u>	<u>Definition</u>	<u>Units</u>
I R Ø T Ø R	Rotor inlet incidence angle at pitchline	°
ALPHA 2A	Next stage stator inlet gas angle at pitchline	°
DBETA R	Rotor root turning angle	°
M 1	Stator exit Mach number at pitchline	---
M1 RT	Stator exit Mach number at root	---
MR 1A	Rotor inlet relative Mach number at pitchline	---
MR1A RT	Rotor inlet relative Mach number at root	---
MR 2	Rotor exit relative Mach number at pitchline	---
MR2 TIP	Rotor exit relative Mach number at tip	---
E/TH CR	Stage equivalent energy, corrected to standard inlet critical conditions	BTU/lb
N/RTH CR	Stage equivalent speed, corrected to standard inlet critical conditions	RPM
WRTHCRE/D	Stage equivalent flow, correct to standard inlet critical conditions	lb/sec

3. Overall Turbine Performance Parameters

<u>Symbol</u>	<u>Definition</u>	<u>Units</u>
PSI P	Overall kinetic energy loading parameter at pitchline	---
PSI R	Overall kinetic energy loading parameter at root	---
DEL H	Overall enthalpy drop (energy output)	BTU/lb
WRT/P	Turbine inlet corrected weight flow function	(lb/sec) ($^{\circ}\text{R}/\text{psia}$) ^{1/2}
N/RT	Turbine inlet corrected speed	RPM/($^{\circ}\text{R}$) ^{1/2}
DELH/TTIN	Overall energy function	BTU/lb $^{\circ}\text{R}$
PT0/PTBAR2	Overall average total pressure ratio	---
PT0/PS2	Overall total to static pressure ratio at pitchline	---
PT0/PAT2A	Overall total to axial total pressure ratio at pitchline	---

<u>Symbol</u>	<u>Definition</u>	<u>Units</u>
ETA TT	Overall total to total efficiency	---
ETA TS	Overall total to static efficiency	---
ETA TAT	Overall total to axial total efficiency	---
WNE/60D	Turbine inlet equivalent flow-speed parameter	lb/sec ²
N/RTH CR	Turbine inlet equivalent speed, corrected to standard inlet critical conditions	RPM
E/TH CR	Overall equivalent energy, corrected to standard inlet critical conditions	BTU/lb

4. Inter-Stage Radial Sector Performance Parameters

<u>Symbol</u>	<u>Definition</u>	<u>Units</u>
DIAM 0	Diameter of mid-points of radial sectors at stator inlet	in
TT 0	Total temperature at stator inlet	°R
PT 0	Total pressure at stator inlet	psia
ALPHA 0	Gas angle (with respect to axial direction) at stator inlet	°
I STATØR	Incidence angle at stator inlet	°
V 0	Gas velocity (composed of tangential and axial components) at stator inlet	ft/sec
VU 0	Tangential gas velocity at stator inlet	ft/sec
VZ 0	Axial gas velocity at stator inlet	ft/sec
TS 0	Static temperature at stator inlet	°R
PS 0	Static pressure at stator inlet	psia
DENS 0	Static density at stator inlet	lb/ft ³
M 0	Mach number at stator inlet	---
CP 0	Specific heat at constant pressure at station inlet	BTU/lb °R
RG 0	Gas constant at stator inlet	ft lb/lb °R

<u>Symbol</u>	<u>Definition</u>	<u>Units</u>
GAMG 0	Ratio of specific heats at stator inlet	---
RWG 0	Ratio of station flow to turbine inlet flow (by definition this must be 1.0 at the first stator inlet of turbine)	---
WG 0	Weight flow at stator inlet	lb/sec
DIAM 1	Diameter of mid-points of radial sectors at stator exit	in
ALPHA 1	Gas angle (with respect to axial direction) at stator exit	°
DEL A	Gas turning angle ($\alpha_0 + \alpha_1$)	°
V 1	Gas velocity (composed of tangential and axial components (at stator exit	ft/sec
VU 1	Tangential gas velocity at stator exit	ft/sec
VZ 1	Axial gas velocity at stator exit	ft/sec
TS 1	Static temperature at stator exit	°R
PS 1	Static pressure at stator exit	psia
DENS 1	Static density at stator exit	lb/ft ³
M 1	Mach number at stator exit	---
ZWI INC	Zweifel parameter, incompressible	---
CP S	Stator pressure coefficient, incompressible	---
CP 1	Specific heat at constant pressure at stator exit	BTU/lb °R
RG 1	Gas constant at stator exit	ft lb/lb °R
GAMG 1	Ratio of specific heats at stator exit	---
RWG 1	Ratio of stator exit flow to turbine inlet flow	---
WG 1	Weight flow at stator exit	lb/sec
DIAM 1A	Diameter of mid-points of radial sectors at root inlet	in
PTR 1A	Relative total pressure at rotor inlet	psia
TTR 1A	Relative total temperature at rotor inlet	°R
BETA 1A	Relative gas angle at rotor inlet	°

<u>Symbol</u>	<u>Definition</u>	<u>Units</u>
I R Ø T Ø R	Incidence angle at rotor inlet	°
R 1A	Relative gas velocity at rotor inlet	ft/sec
RU 1A	Relative gas tangential velocity at rotor inlet	ft/sec
MR 1A	Relative Mach number at rotor inlet	---
U 1A	Wheel speed at rotor inlet	ft/sec
PS 1A	Static pressure at rotor inlet	psia
TS 1A	Static temperature at rotor inlet	°R
CP 1A	Specific heat at constant pressure at rotor inlet	BTU/lb °R
RG 1A	Gas constant at rotor inlet	ft lb/lb °R
GAMG 1A	Ratio of specific heats at rotor inlet	---
RWG 1A	Ratio of rotor inlet flow to turbine inlet flow	---
WG 1A	Weight flow at rotor inlet	lb/sec
DIAM 2	Diameters of mid-points of radial sectors at rotor exit	in
PTR 2	Relative total pressure at rotor exit	psia
TTR 2	Relative total temperature at rotor exit	°R
BETA 2	Relative gas angle at rotor exit	°
DBETA	Gas turning angle ($\beta_{1A} + \beta_2$)	°
R 2	Relative gas velocity at rotor exit	ft/sec
RU 2	Relative tangential gas velocity at rotor exit	ft/sec
MR 2	Relative Mach number at rotor exit	---
U 2	Wheel speed at rotor exit	ft/sec
RX	Reaction	---
DELH	Enthalpy drop (energy output)	BTU/lb
PSI P	Kinetic energy loading parameter	---
ETA TT	Total to total efficiency	---
ETA TS	Total to static efficiency	---
ETA AT	Total to axial total efficiency	---

<u>Symbol</u>	<u>Definition</u>	<u>Units</u>
ZWI INC	Zweifel parameter, incompressible	---
CP R	Rotor pressure coefficient, incompressible	---
PS 2	Static pressure at rotor exit	psia
TS 2	Static temperature at rotor exit	$^{\circ}\text{R}$
CP 2	Specific heat at constant pressure at rotor exit	BTU/lb $^{\circ}\text{R}$
RG 2	Gas constant at rotor exit	ft lb/lb $^{\circ}\text{R}$
GAMG 2	Ratio of specific heats at rotor exit	---
RWG 2	Ratio of rotor exit flow to turbine inlet flow	---
WG 2	Weight flow at rotor exit	lb/sec
PT 2A	Total pressure at inlet to next stator	psia
TT 2A	Total temperature at inlet to next stator	$^{\circ}\text{R}$
V 2A	Gas velocity (composed of tangential and axial components) at inlet to next stator	ft/sec
VU 2A	Tangential gas velocity at inlet to next stator	ft/sec
ALPHA 2A	Gas angle (with respect to axial direction) at inlet to next stator	$^{\circ}$
MF 2A	Axial Mach number at inlet to next stator	---
VZ 2A	Axial gas velocity at inlet to next stator	ft/sec
TS 2A	Static temperature at inlet to next stator	$^{\circ}\text{R}$
PS 2A	Static pressure at inlet to next stator	psia
DENS 2A	Static density at inlet to next stator	lb/ft ³
M 2A	Mach number at inlet to next stator	---
CP 2A	Specific heat at constant pressure at inlet to next stator	BTU/lb $^{\circ}\text{R}$
RG 2A	Gas constant at inlet to next stator	ft lb/lb $^{\circ}\text{R}$
GAMG 2A	Ratio of specific heats at inlet to next stator	---
RWG 2A	Ratio of flow at inlet to next stator to turbine inlet flow	---
WG 2A	Weight flow at inlet to next stator	lb/sec

IV. METHOD FOR CALCULATION OF MODIFIED PARAMETERS FOR WET VAPOR TURBINES

A. Assumptions Used and Development of Equations for Modified Parameters

In wet vapor turbines since there exists two distinct phases (gas and liquid), the usual ideal thermodynamic relationships which are valid for gas turbines are not directly applicable. The approach used to determine the performance of wet vapor turbines involved making a minimum of changes in the code but required modifying the input data appropriately to closely simulate the thermodynamic processes of a turbine operating within the saturation dome of a T-S (temperature-entropy) diagram. The following method was derived and gives good agreement with the results from the WSD 2-D code as run by Fentress⁽²⁾.

In order to arrive at a consistent set of relatively simple relationships, the following assumptions were made:

1. The inlet hub and tip diameters for a given blade row are assumed equal to the exit hub and tip diameters from the preceding blade row. The same assumption holds true for the modified parameters γ^* , η^* , and R^* . The superscript * indicates that it is a modified value for specific heat ratio, blade efficiency, and gas constant.
2. All inefficiencies are assumed to be lumped into the single blade efficiency parameter η^* . This includes such items as incidence and exit losses and flow coefficients. Consequently $EXPP = EXPN = 0.0$, $SREC = RREC = 1.0$, $SCF = RCF = 1.0$, $RTF = 1.0$, and $SESTH = RERTH = 1.0$. The definitions of these computer code terms may be found in Section III-A.
3. The exit gas flow angle from each blade row is taken to be equal to the exit blade angle. Therefore, actual blade exit angles (SDEA and RDEA) are input rather than distributed passage areas (SPA and RPA).

4. Since all energy changes are accounted for in the calculation of the modified parameters, there is no need to take into consideration the decrease in the gas flow rate due to condensation effects. Consequently $RWG = 1.0$.
5. Radial variations in γ^* , η^* , and R^* are assumed to be negligible.

In applying the following formulas to determine the modified values of R^* , γ^* , and η^* , care must be exercised to obtain the proper relative velocity either entering or leaving a blade row. See Figure IV-1 for clarification of the station terminology used in the example potassium turbine. The initial values for static temperatures, pressures, specific volumes, and velocities are obtained from previous 1-D calculations. Definitions of the nomenclature used are given in Section IV-B.

FIFTH STAGE

$$R_0^* = \frac{144 P_{S0} v_{S0}}{T_{S0}} \quad (1)$$

$$\gamma_0^* = \frac{1}{1 - \frac{2 g R_0^* (T_{T0} - T_{S0})}{V_0^2}} \quad (2)$$

$$P_{T0}^* = P_{S0} \left(\frac{T_{T0}}{T_{S0}} \right)^{\frac{\gamma_0^*}{\gamma_0^* - 1}} \quad (3)$$

$$PTPS = \frac{P_{T0}^*}{P_{S1}} \quad (4)$$

$$R_1^* = \frac{144 P_{S1} v_{S1}}{T_{S1}} \quad (5)$$

$$\gamma_1^* = \frac{1}{1 - \frac{2 g R_1^* (T_{T0} - T_{S1})}{V_1^2}} \quad (6)$$

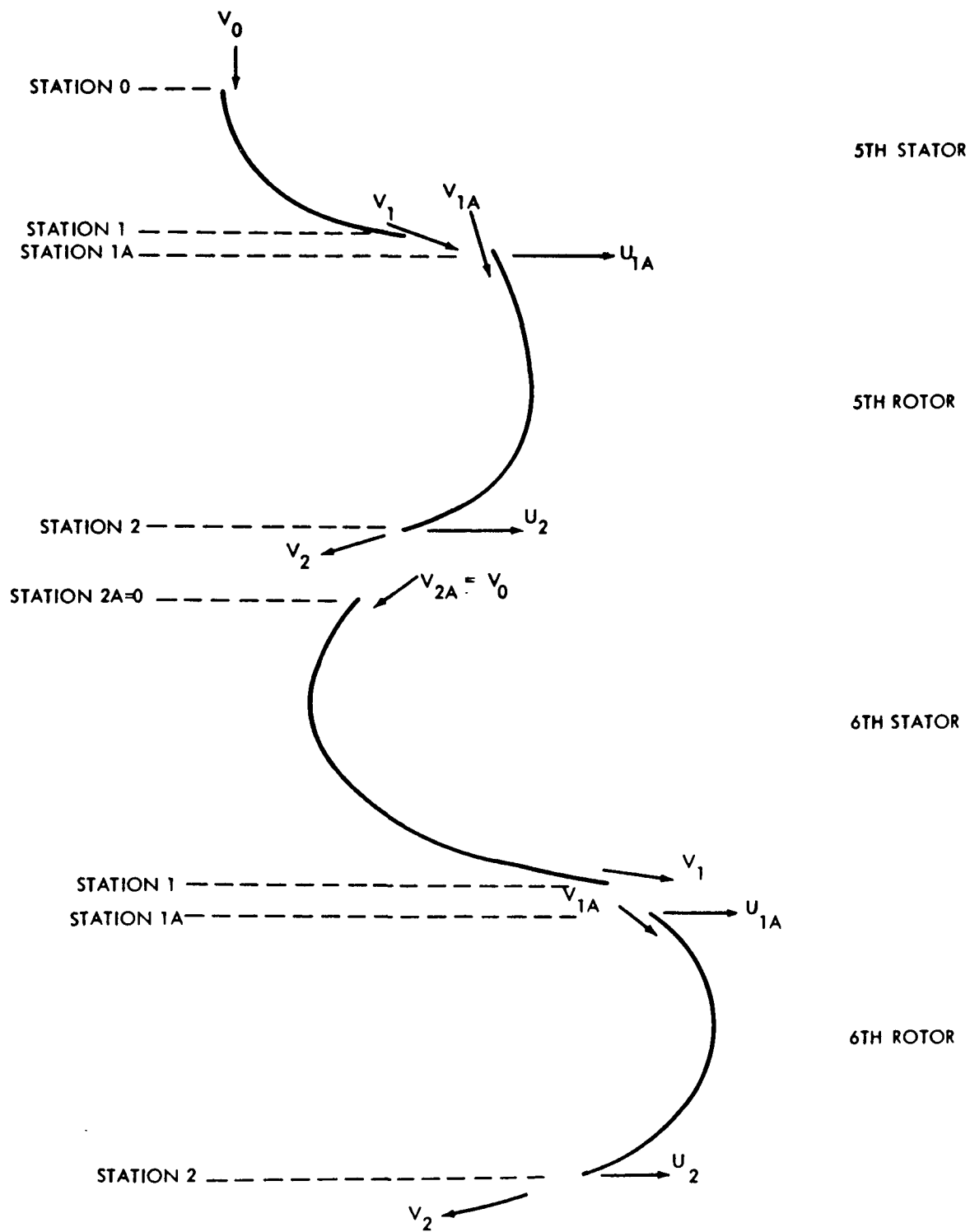


Figure IV-1. Axial Station Velocity Nomenclature

$$\eta_1^* = \frac{T_{T0} - T_{S1}}{T_{T0} \left[1 - \left(\frac{P_{S1}}{P_{T0}^*} \right)^{\frac{\gamma_1^* - 1}{\gamma_1^*}} \right]} \quad (7)$$

$$D_{R1A}^* = D_{R1} \quad (8)$$

$$D_{T1A}^* = D_{T1} \quad (9)$$

$$R_{1A}^* = R_1^* \quad (10)$$

$$\gamma_{1A}^* = \gamma_1^* \quad (11)$$

$$R_2^* = \frac{144 P_{S2} v_{S2}}{T_{S2}} \quad (12)$$

$$T_{T2g} = T_{S1A} + \frac{(\gamma_{1A}^* - 1) (V_{1A}^2 + U_2^2 - U_{1A}^2)}{2g \gamma_{1A}^* R_{1A}^*} \quad (13)$$

$$P_{T2g} = P_{S1A} \left[1 + \frac{(\gamma_{1A}^* - 1) (V_{1A}^2 + U_2^2 - U_{1A}^2)}{2g \gamma_{1A}^* R_{1A}^* T_{S1A}} \right]^{\frac{\gamma_{1A}^*}{\gamma_{1A}^* - 1}} \quad (14)$$

$$\gamma_2^* = \frac{1}{1 - \frac{2g R_2^* (T_{T2g} - T_{S2})}{V_2^2}} \quad (15)$$

$$\eta_2^* = \frac{T_{T2g} - T_{S2}}{T_{T2g} \left[1 - \left(\frac{P_{S2}}{P_{T2g}} \right)^{\frac{\gamma_2^* - 1}{\gamma_2^*}} \right]} \quad (16)$$

$$D_{R2A}^* = D_{R2} \quad (17)$$

$$D_{T2A}^* = D_{T2} \quad (18)$$

$$R_{2A}^* = R_2^* \quad (19)$$

$$\gamma_{2A}^* = \gamma_2^* \quad (20)$$

$$D_{R0}^* = D_{R2A}^* \quad (21)$$

$$D_{T0}^* = D_{T2A}^* \quad (22)$$

$$R_0^* = R_{2A}^* \quad (23)$$

$$\gamma_0^* = \gamma_{2A}^* \quad (24)$$

$$R_1^* = \frac{144 P_{S1} v_{S1}}{T_{S1}} \quad (25)$$

$$T_{T0g} = T_{S2A} + \frac{v_{2A}^2}{2g \gamma_{2A}^* R_{2A}^*} \frac{\gamma_{2A}^*}{\gamma_{2A}^* - 1} \quad (26)$$

$$P_{T0g} = P_{S2A} \left(\frac{T_{T0g}}{T_{S2A}} \right)^{\frac{\gamma_{2A}^*}{\gamma_{2A}^* - 1}} \quad (27)$$

$$\gamma_1^* = \frac{1}{1 - \frac{2g R_1^* (T_{T0g} - T_{S1})}{v_1^2}} \quad (28)$$

$$\eta_1^* = \frac{T_{T0g} - T_{S1}}{T_{T0g} \left[1 - \left(\frac{P_{S1}}{P_{T0g}} \right)^{\frac{\gamma_1^* - 1}{\gamma_1^*}} \right]} \quad (29)$$

The remainder of the expressions for the modified parameters for the rest of the sixth stage are the same as those in Equations (8) through (20). For turbines with more than two stages, the same relationships are repeated for each succeeding stage. Since there is a significant amount of hand calculations involved in obtaining the modified parameters, a small computer program could be written to punch out these values in a format compatible with the input to the modified NASA turbine code.

B. Nomenclature Used in Calculation of Modified Parameters

<u>Symbol</u>	<u>Definition</u>	<u>Units</u>
D_R	Root diameter	in
D_T	Tip diameter	in
g	Gravitational acceleration (32.2)	ft/sec ²
P_S	Static pressure	psia
P_T	Total pressure	psia
PTPS	Total-to-static pressure ratio across first stator	---
R	Gas constant	ft/ ^o R
T_S	Static temperature	^o R
T_T	Total temperature	^o R
U	Wheel speed	ft/sec
V	Gas velocity	ft/sec
v_S	Specific volume	ft ³ /lb
γ	Ratio of specific heats	---
η	Overall effective blade efficiency	---

V. POSSIBLE FUTURE MODIFICATION TO CODE

1. With the advent of the CDC 6600 computer and its 65 K core (as compared to the IBM 7094 and its core of 32 K), it is possible to expand the maximum number of radial sectors to greater than 6 and the maximum number of stages to exceed 8. Of course computer run times would be longer and a different method of printing out data would have to be used.
2. The code could be changed so as to iterate to a desired exit pressure condition automatically by comparing the average turbine exit total pressure with that desired. If the difference between the exit total pressures were not within some given tolerance, the first stator pressure ratio PTPS would be adjusted accordingly.
3. Non-uniform turbine inlet radial distributions in pressure, temperature, and velocity could be achieved by inputting such quantities. The assumption in the code as presently programmed is that the inlet radial distributions are uniform.

REFERENCES

- (1) E. E. Flagg, "Analytical Procedure and Computer Program for Determining the Off-Design Performance of Axial Flow Turbines", NASA CR-710, February 1967.
- (2) Westinghouse Electric Corporation, Astronuclear Laboratory, Report WANL-PR(DD)-017, January 1967, Contract NAS 7-390.

APPENDIX I

SAMPLE PROBLEMS ILLUSTRATING USE OF CODE

A. NASA Reference Two-Stage Gas Turbine (5 Radial Sectors)

1. Comparison of Results

The sample problem given in NASA CR-710 was run both on the IBM 7094 (II) and CDC 6600 computer. The data output from both machines was in exact agreement to at least the sixth significant figure. The minor discrepancies noted were thought to be due to the difference in the number of significant places carried in the respective machines. It was found that the sample problem data output given in NASA CR-710 did not exactly correspond to that report's data input. When the data input was appropriately changed, the subsequent output was in substantial agreement (at least to the fourth significant place) with that given in NASA CR-710. No explanation can be given at this time as to why there was not agreement to at least the sixth place. But it is felt that the agreement is more than adequate to satisfy engineering criteria.

2. Data Input

```

                                TURBINE COMPTER PROGRAM
NASA TWO STAGE REFERENCE TURBINE
1.00 5041 -8 DEG. LOSS PROFILE .98 .946, .977 .90,
$DATAIN
STGCH=      1.000
TTIN=      700.000  PTIN=      17.140  WAIR=      0.000  FAIR=      0.000
PTPS=      1.600  DELC=      0.000  DELL=      0.000  DELA=      0.000
STG=       2.000  SECT=      5.000  EXPN=      3.000  EXPP=      3.000
PAF=       0.000  SLI=       0.000  AACs=      1.000  RPM=     5041.000
VCTD=      1.000  RSL=     53.350  TSL=     518.688  PSL=     14.696
GAMSL=     1.400  ENDSTG=     0.000  ENDJOB=     0.000

                                INLET RADIAL PROFILES
PCNH=      .200      .200      .200      .200      .200      0.000

```

2. Data Input (continued)

		STANDARD OPTION					
		AXIAL STATIONS					
STAGE=	1	STA. 0	STA. 1	STA. 1A	STA. 2	STA. 2A	
RG=	53.350	53.350	53.350	53.350	53.350	53.350	0.000
GAMG=	1.400	1.400	1.400	1.400	1.400	1.400	0.000
DR=	19.110	19.110	18.969	18.406	18.265		0.000
DT=	28.000	28.000	28.141	28.704	28.845		0.000
RWG=	1.000	1.000	1.000	1.000	1.000		0.000

		STATOR RADIAL DISTRIBUTIONS					
		ROOT	PITCH			TIP	
SDIA=	0.000	0.000	0.000	0.000	0.000	0.000	0.000
SDEA=	0.000	0.000	0.000	0.000	0.000	0.000	0.000
SREC=	1.000	1.000	1.000	1.000	1.000	1.000	0.000
SETA=	.970	.980	.980	.980	.970		0.000
SCF=	.977	.977	.977	.977	.977		0.000
SPA=	22.140	26.035	30.135	34.194	38.499		0.000
SESTH=	1.000						

		ROTOR RADIAL DISTRIBUTIONS					
RDIA=	50.600	44.900	38.100	30.200	20.900		0.000
RDEA=	0.000	0.000	0.000	0.000	0.000		0.000
RREC=	1.000	1.000	1.000	1.000	1.000		0.000
RETA=	.919	.946	.946	.946	.919		0.000
RCF=	.950	.950	.950	.950	.950		0.000
RPA=	33.408	36.352	38.976	41.280	43.008		0.000
RTF=	1.000	1.000	1.000	1.000	1.000		0.000
RERTH=	1.010						

		STANDARD OPTION					
		AXIAL STATIONS					
STAGE=	2	STA. 0	STA. 1	STA. 1A	STA. 2	STA. 2A	
RG=	53.350	53.350	53.350	53.350	53.350	53.350	0.000
GAMG=	1.400	1.400	1.400	1.400	1.400	1.400	0.000
DR=	18.265	17.814	17.673	17.110	17.110		0.000
DT=	28.845	29.296	29.437	30.000	30.000		0.000
RWG=	1.000	1.000	1.000	1.000	1.000		0.000

		STATOR RADIAL DISTRIBUTIONS					
		ROOT	PITCH			TIP	
SDIA=	25.000	22.400	20.200	18.300	16.600		0.000
SDEA=	0.000	0.000	0.000	0.000	0.000		0.000
SREC=	1.000	1.000	1.000	1.000	1.000		0.000
SETA=	.970	.980	.980	.980	.970		0.000
SCF=	.925	.925	.925	.925	.925		0.000
SPA=	30.420	36.855	43.485	50.765	58.240		0.000
SESTH=	1.010						

		ROTOR RADIAL DISTRIBUTIONS					
RDIA=	36.600	26.900	16.100	4.600	-6.700		0.000
RDEA=	0.000	0.000	0.000	0.000	0.000		0.000
RREC=	1.000	1.000	1.000	1.000	1.000		0.000
RETA=	.919	.946	.946	.946	.919		0.000
RCF=	.900	.900	.900	.900	.900		0.000
RPA=	43.350	48.150	52.350	55.750	58.550		0.000
RTF=	1.000	1.000	1.000	1.000	1.000		0.000
RERTH=	1.010						

3. Listing of Data Output

```

      NASA TURBINE COMPUTER PROGRAM
NASA TWO STAGE REFERENCE TURBINE
1.00 5041 -R DEG. LOSS PROFILE .98 .946. .977 .90.
      CASE 1. 0
      STAGE PERFORMANCE
            STAGE 1      STAGE 2      STAGE 3      STAGE 4

TTPAR 0      700.0      608.5
PTPAR 0      17.140     10.140
  WG 0      43.612     43.612
  CEL F      21.960     11.370
  WRT/P      67.320     106.303
DH/TTPAR0    .03137     .01869
  N/RT      190.532     204.358
  ETA TT      .93545     .93046
  ETA TS      .82312     .74101
  ETA AT      .92064     .92376
  PT0/PS1     1.600     1.347
PTPAR0/PTPAR2 1.694     1.358
  PTPAR0/PS2   1.840     1.475
  PTR2/PS2     1.340     1.216
TTPAR2/TTPAR0 .86926     .92212
TTP1A/TTPAR0 .91710     .94753
  WG 1      43.612     43.612
  PS 1A      10.770     7.659
  TTR 1A      642.0     576.8
  PTR 1A      12.478     8.343
  WG 1A      43.612     43.612
  PS 2       9.314     6.860
  TTPAR 2     608.5     561.1
  PTPAR 2     10.120     7.452
  WG 2      43.612     43.612
  WG 2A      43.612     43.612
  UP/VI      .44821     .59055
  UR/VI      .35559     .43632
  PSI P      1.02409     .53026
  PSI R      1.62705     .97270
  RX P       .21420     .26054
  RX R      -.08793     -.07253
  ALPHA 0     0.000     20.347
  I STATOR     0.000     .127
  BETA 1A     46.336     15.343
  I ROTOR      8.236     -.757
  ALPHA 2A     20.327     -9.259
  DBETA R     116.216     86.338
  M 1         .83798     .64215
  M1 RT      1.01118     .78439
  MR 1A      .47064     .35156
  MR1A RT    .69181     .50438
  MR 2       .64048     .52077
  MR2 TIP    .69787     .61846
  E/TH CR     16.272     9.652
  N/RTH CR    4339.3     4654.2
WRTHCRE/C    43.440     68.554

      OVERALL PERFORMANCE
PSI P      .77717      PSI R      1.32335      DEL H      33.33004
WRT/P      67.31951    N/RT      190.53189    DELH/TTIN   .04761
PT0/PTPAR2 2.29991    PT0/PS1   2.49847      PT0/PAT2A   2.30903
ETA TT      .93700      ETA TS     .86213      FTA TAT     .93477
WNE/60C    3141.641    N/RTH CR   4339.329    E/TH CR     24.69720

```

3. Output Data (continued)

NASA TURBINE COMPUTER PROGRAM
NASA TWO STAGE REFERENCE TURBINE
1.00 5041 -8 DEG. LOSS PROFILE .98 .946, .977 .90,
CASE 1. 0
INTER-STAGE PERFORMANCE

STA 0	STATOR INLET		STAGE 1.					
DIAM 0	19.999	21.777	23.555	25.333	27.111			
TT 0	700.0	700.0	700.0	700.0	700.0			
PT 0	17.140	17.140	17.140	17.140	17.140			
ALPHA 0	0.000	0.000	0.000	0.000	0.000			
I STATOR	0.000	0.000	0.000	0.000	0.000			
V 0	299.463	299.463	299.463	299.463	299.463			
VI 0	0.000	0.000	0.000	0.000	0.000			
VZ 0	299.463	299.463	299.463	299.463	299.463			
TS 0	692.5	692.5	692.5	692.5	692.5			
PS 0	16.509	16.509	16.509	16.509	16.509			
DENS 0	.06434	.06434	.06434	.06434	.06434			
M 0	.23213	.23213	.23213	.23213	.23213			
CP 0	.23996	.23996	.23996	.23996	.23996			
RG 0	53.350	53.350	53.350	53.350	53.350			
GAMG 0	1.40000	1.40000	1.40000	1.40000	1.40000			
RWG 0	1.00000	1.00000	1.00000	1.00000	1.00000			
WG 0	6.58435	7.70666	8.80273	9.78081	10.73712	43.61168	TOTAL FLOW	
STA 1	STATOR EXIT							
DIAM 1	19.999	21.777	23.555	25.333	27.111			
ALPHA 1	69.539	67.940	66.303	64.911	63.359			
DEL A	69.539	67.940	66.303	64.911	63.359			
V 1	1147.972	1080.202	1017.726	954.148	895.217			
VO 1	1075.549	1001.125	931.914	864.123	800.175			
VZ 1	401.291	405.692	409.026	404.586	401.413			
TS 1	590.3	602.9	613.8	624.2	633.3			
PS 1	9.252	10.046	10.712	11.379	11.936			
DENS 1	.04230	.04498	.04711	.04920	.05087			
M 1	.96384	.89743	.83798	.77904	.72567			
ZWI INC	-.65502	-.69615	-.73603	-.76804	-.80159			
CP S	.93195	.92314	.91342	.90150	.88810			
CP 1	.23996	.23996	.23996	.23996	.23996			
RG 1	53.350	53.350	53.350	53.350	53.350			
GAMG 1	1.40000	1.40000	1.40000	1.40000	1.40000			
RWG 1	1.00000	1.00000	1.00000	1.00000	1.00000			
WG 1	6.58435	7.70666	8.80273	9.78081	10.73712	43.61168	TOTAL FLOW	

3. Output Data (continued)

NASA TURBINE COMPUTER PROGRAM
NASA TWO STAGE REFERENCE TURBINE
1.00 5041 -8 DEG. LOSS PROFILE .98 .946. .977 .90.
CASE 1. 0
INTER-STAGE PERFORMANCE

STA 1A	ROTOR INLET		STAGE 1.				
DIAM 1A	19.886	21.721	23.555	25.389	27.224		
PTR 1A	11.994	12.251	12.478	12.778	13.083		
TTR 1A	637.2	639.2	642.0	645.8	650.4		
BETA 1A	58.685	53.188	46.336	37.964	27.212		
I Rotor	8.085	8.288	8.236	7.764	6.312		
R 1A	754.102	656.963	572.034	493.769	433.126		
RU 1A	644.243	525.969	413.810	303.751	198.059		
MR 1A	.63340	.54564	.47064	.40274	.35068		
U 1A	437.407	477.755	518.104	558.452	598.801		
PS 1A	9.225	10.067	10.770	11.461	12.035		
TS 1A	589.8	603.3	614.7	625.5	634.8		
CP 1A	.23996	.23996	.23996	.23996	.23996		
RG 1A	53.350	53.350	53.350	53.350	53.350		
GAMG 1A	1.40000	1.40000	1.40000	1.40000	1.40000		
RWG 1A	1.00000	1.00000	1.00000	1.00000	1.00000		
WG 1A	6.58435	7.70666	8.80273	9.78081	10.73712	43.61168	TOTAL FLOW
STA 2	ROTOR EXIT						
DIAM 2	19.436	21.495	23.555	25.615	27.674		
PTR 2	11.947	12.225	12.478	12.810	13.154		
TTR 2	636.5	638.8	642.0	646.3	651.4		
BETA 2	57.531	58.629	59.379	60.258	60.964		
UBETA	116.216	111.817	105.715	98.223	88.175		
R 2	700.060	738.948	764.765	797.556	818.486		
RU 2	590.630	630.922	658.123	692.496	715.611		
MR 2	.58512	.61884	.64048	.66793	.68411		
U 2	427.500	472.802	518.104	563.406	608.708		
RX	-.00402	.11699	.21420	.30756	.38295		
DELTA	21.683	22.139	22.182	22.137	21.658		
PST P	2.90236	2.45376	2.06895	1.76143	1.48743		
ETA TT	.91418	.94184	.94518	.94752	.92554		
ETA TS	.80596	.82681	.83127	.83236	.81639		
ETA AT	.89419	.92233	.92980	.93438	.91677		
ZWI INC	-1.85326	-1.61317	-1.42033	-1.24554	-1.09099		
CP R	-.16035	.20959	.44051	.61671	.71997		
PS 2	9.277	9.298	9.314	9.330	9.342		
TS 2	595.7	593.3	593.3	593.3	595.7		
CP 2	.23996	.23996	.23996	.23996	.23996		
RG 2	53.350	53.350	53.350	53.350	53.350		
GAMG 2	1.40000	1.40000	1.40000	1.40000	1.40000		
RWG 2	1.00000	1.00000	1.00000	1.00000	1.00000		
WG 2	6.89815	7.85826	8.73519	9.66391	10.45615	43.61166	TOTAL FLOW
PT 2A	10.061	10.112	10.122	10.148	10.138		
TT 2A	609.6	607.7	607.6	607.7	609.7		
V 2A	402.148	406.590	403.078	404.007	398.169		
VU 2A	164.083	158.536	140.019	128.806	106.469		
ALPHA 2A	24.080	22.949	20.327	18.592	15.509		
MF 2A	.30674	.31338	.31636	.32046	.32044		
VZ 2A	367.151	374.409	377.978	382.924	383.670		
TS 2A	596.2	594.0	594.0	594.2	596.5		
PS 2A	9.304	9.334	9.355	9.376	9.391		
DFNS 2A	.04213	.04241	.04251	.04259	.04249		
M 2A	.33598	.34032	.33736	.33811	.33255		
CP 2A	.23996	.23996	.23996	.23996	.23996		
RG 2A	53.350	53.350	53.350	53.350	53.350		
GAMG 2A	1.40000	1.40000	1.40000	1.40000	1.40000		
RWG 2A	1.00000	1.00000	1.00000	1.00000	1.00000		
WG 2A	6.89815	7.85826	8.73519	9.66391	10.45615	43.61166	TOTAL FLOW

3. Output Data (continued)

NASA TURBINE COMPUTER PROGRAM
NASA TWO STAGE REFERENCE TURBINE
1.00 5041 -8 DEG. LOSS PROFILE .98 .946. .977 .90.
CASE 1. 0
INTER-STAGE PERFORMANCE

STA 0	STATOR INLET		STAGE 2.					
DIAM 0	19.323	21.439	23.555	25.671	27.787			
TT 0	608.5	608.5	608.5	608.5	608.5			
PT 0	10.120	10.120	10.120	10.120	10.120			
ALPHA 0	24.080	22.949	20.327	18.592	15.509			
I STATOR	-.920	.549	.127	.292	-1.090			
V 0	402.148	406.590	403.078	404.007	398.169			
VU 0	164.083	158.536	140.019	128.806	106.469			
VZ 0	367.151	374.409	377.978	382.924	383.670			
TS 0	596.2	594.0	594.0	594.2	596.5			
PS 0	9.304	9.334	9.355	9.376	9.391			
DENS 0	.04213	.04241	.04251	.04259	.04249			
M 0	.33598	.34032	.33736	.33811	.33255			
CP 0	.23996	.23996	.23996	.23996	.23996			
RG 0	53.350	53.350	53.350	53.350	53.350			
GAMG 0	1.40000	1.40000	1.40000	1.40000	1.40000			
RWG 0	1.00000	1.00000	1.00000	1.00000	1.00000			
WG 0	6.89815	7.85826	8.73519	9.66391	10.45615	43.61166	TOTAL FLOW	
STA 1	STATOR EXIT							
DIAM 1	18.962	21.259	23.555	25.851	28.148			
ALPHA 1	61.651	59.301	57.068	54.670	52.234			
DEL A	85.731	82.250	77.395	73.262	67.744			
V 1	852.196	795.576	746.339	695.126	650.709			
VU 1	749.990	684.085	626.415	567.111	514.398			
VZ 1	404.665	406.164	405.742	401.978	398.519			
TS 1	548.0	555.8	562.1	568.3	573.2			
PS 1	6.934	7.321	7.624	7.926	8.158			
DENS 1	.03415	.03556	.03661	.03765	.03841			
M 1	.74259	.68839	.64215	.59484	.55441			
ZWI INC	-1.03734	-1.09869	-1.13154	-1.16854	-1.17646			
CP S	.77731	.73881	.70832	.66221	.62558			
CP 1	.23996	.23996	.23996	.23996	.23996			
RG 1	53.350	53.350	53.350	53.350	53.350			
GAMG 1	1.40000	1.40000	1.40000	1.40000	1.40000			
RWG 1	1.00000	1.00000	1.00000	1.00000	1.00000			
WG 1	6.56380	7.69027	8.76394	9.79979	10.79383	43.61164	TOTAL FLOW	

3. Output Data (continued)

NASA TURBINE COMPUTER PROGRAM
NASA TWO STAGE REFERENCE TURBINE
1.00 5041 -8 DEG. LOSS PROFILE .98 .946, .977 .90,
CASE 1, 0
INTER-STAGE PERFORMANCE

STA 1A ROTOR INLET			STAGE 2.		
DIAM 1A	18.849	21.202	23.555	25.908	28.261
PTR 1A	7.983	8.161	8.343	8.600	8.871
TTR 1A	570.7	573.3	576.8	581.8	587.6
BETA 1A	40.535	28.960	15.343	-.585	-15.815
I ROTOR	3.935	2.060	-.757	-5.185	-9.115
R 1A	522.962	453.435	405.333	389.963	400.905
KU 1A	339.876	219.552	108.311	-3.979	-109.261
MR 1A	.45573	.39220	.35196	.33344	.34128
U 1A	414.602	466.353	518.104	569.855	621.606
PS 1A	6.930	7.341	7.659	7.970	8.209
TS 1A	548.0	556.2	562.9	569.2	574.3
CP 1A	.23996	.23996	.23996	.23996	.23996
RG 1A	53.350	53.350	53.350	53.350	53.350
GAMG 1A	1.40000	1.40000	1.40000	1.40000	1.40000
RWG 1A	1.00000	1.00000	1.00000	1.00000	1.00000
WG 1A	6.56380	7.69027	8.76394	9.79979	10.79383
					43.61164 TOTAL FLOW
STA 2 ROTOR EXIT					
DIAM 2	18.399	20.977	23.555	26.133	28.711
PTR 2	7.950	8.142	8.343	8.625	8.926
TTR 2	570.0	573.0	576.8	582.3	588.7
BETA 2	45.803	47.980	49.600	51.528	52.888
URETA	86.338	76.940	64.943	50.944	37.072
R 2	510.818	554.390	597.132	646.661	685.002
RI 2	366.232	414.834	454.736	506.279	546.256
MR 2	.44500	.48703	.52077	.56377	.59605
I 2	404.695	461.399	518.104	574.808	631.513
RX	.02767	.16069	.26054	.38780	.43067
DELH	11.872	11.918	11.652	11.307	10.570
EST P	1.77102	1.38665	1.08674	.84417	.67406
ETA TT	.92499	.94980	.94776	.94026	.89652
ETA TS	.77200	.77551	.75903	.73739	.69018
ETA AT	.92288	.94655	.94164	.93302	.88571
ZWI INC	-1.83075	-1.49053	-1.21766	-.94425	-.75608
CR R	-.04411	.34059	.53009	.67634	.65747
PS 2	6.855	6.857	6.860	6.863	6.867
TS 2	548.3	547.0	547.1	547.5	549.6
CP 2	.23996	.23996	.23996	.23996	.23996
RG 2	53.350	53.350	53.350	53.350	53.350
GAMG 2	1.40000	1.40000	1.40000	1.40000	1.40000
RWG 2	1.00000	1.00000	1.00000	1.00000	1.00000
WG 2	6.21734	7.46048	8.67619	10.00385	11.25376
PT 2A	7.334	7.389	7.439	7.491	7.537
TT 2A	559.0	558.8	559.9	561.4	564.4
V 2A	358.174	376.670	392.171	408.102	422.019
VU 2A	-38.463	-46.565	-63.368	-64.529	-85.256
ALPHA 2A	-6.165	-7.101	-9.299	-9.667	-11.655
MF 2A	.31022	.32601	.33752	.35074	.35964
VZ 2A	356.102	373.780	387.017	402.308	413.318
TS 2A	548.3	547.0	547.1	547.5	549.6
PS 2A	6.855	6.857	6.860	6.863	6.867
DENS 2A	.03374	.03384	.03384	.03384	.03372
M 2A	.31702	.32853	.34202	.35579	.36721
CP 2A	.23996	.23996	.23996	.23996	.23996
RG 2A	53.350	53.350	53.350	53.350	53.350
GAMG 2A	1.40000	1.40000	1.40000	1.40000	1.40000
RWG 2A	1.00000	1.00000	1.00000	1.00000	1.00000
WG 2A	6.21734	7.46048	8.67619	10.00385	11.25376
					43.61162 TOTAL FLOW

B. Wet-Vapor Potassium Turbine* (5 Radial Sectors)

1. Calculation of Modified Parameters

Using the equations given in Section IV, the values for the modified parameters (given in Table I B-1) were calculated by hand and used as data input to the modified NASA turbine code. Only the 5th and 6th stages are analyzed and correspond to stages 1 and 2 in the output listing.

2. Comparison of Results from Modified NASA Code and WSD Code

Table I B-2 shows a comparison of the results between the 1-D and 2-D codes from WSD and the NASA code using the modified parameters. The total-to-static pressure ratio (PTPS) across the first stator was adjusted until the turbine exit conditions were identical to those obtained in the Steam Division codes. The modified parameters were assumed to remain constant during the small changes in PTPS. Unfortunately, a completely consistent set of input data was impossible to be obtained from either Table I or Table II of Reference (2) or a combination of the two. The difference in the 2-D blade angle distribution from that used in the 1-D calculation is most likely the primary reason that the jet velocities at the mean diameters are not in better agreement.

Figures I B-1 and I B-2 show the slight differences in the angles used in Fentress 2-D calculations and those used as input to the NASA code 2-D analysis. Figure I B-3 shows the good agreement between the turbine exit jet velocities as calculated by both codes. In Figure I B-4, there is also good agreement with the static pressure distributions from the 5th stator exit.

It is therefore concluded that if one performs a hand solution (or uses an appropriate computer code) for a 1-D turbine analysis, then this method of using modified γ , R , and η parameters with the NASA code will give a valid and thermodynamically consistent two-dimensional analysis of a turbine operating in the wet vapor region.

* Described in Reference (2).

TABLE I B-1
MODIFIED PARAMETERS FOR POTASSIUM TURBINE

Station	D_R^*	D_T^*	R^*	γ^*	η^*	
0	5.29	7.51	31.158	1.1825	---	$P_{T0}^* = 38.828$; $PTPS = 1.3619$
1	5.15	7.83	30.842	1.1437	0.92577	
1A	5.15	7.83	30.842	1.1437	---	
2	5.04	8.28	30.689	1.16607	0.81662	
2A	5.04	8.28	30.689	1.16607	---	
0	5.04	8.28	30.689	1.16607	---	
1	4.88	8.62	30.828	1.1447	0.94752	
1A	4.88	8.62	30.828	1.1447	---	
2	4.60	9.10	30.763	1.1637	0.8155	
2A	4.60	9.10	30.763	1.1637	---	

TABLE IB-2
COMPARISON OF POTASSIUM TURBINE DATA AT MEAN DIAMETER

Blade Row Exit Conditions	Fourth Rotor			Fifth Stator			Fifth Rotor			Sixth Stator			Sixth Rotor		
	⊙ 1-D Code (1)	NASA Code (2)	% Difference	⊙ 1-D Code (1)	NASA Code (2)	% Difference	⊙ 1-D Code (1)	NASA Code (2)	% Difference	⊙ 1-D Code (1)	NASA Code (2)	% Difference	⊙ 1-D Code (1)	NASA Code (2)	% Difference
Blade height (inch)	1.11	1.11 *	0.0	1.34	1.34 *	0.0	1.62	1.62 *	0.0	1.87	1.87 *	0.0	2.25	2.25 *	0.0
Mean diameter (inch)	6.40	6.40 *	0.0	6.49	6.49 *	0.0	6.66	6.66 *	0.0	6.75	6.75 *	0.0	6.85	6.85 *	0.0
Flow angle (degree)	64.37	--	--	64.37(65.03)	65.03 *	+1.03(0.0)	64.37(63.65)	63.65 *	-1.12(0.0)	57.32(57.57)	57.57 *	+0.436(0.0)	60.30(58.98)	58.98 *	-2.19(0.0)
Static pressure (psia)	37.00	--	--	28.51	28.198	-1.09	22.04	21.963	-0.349	19.69	19.495	-0.990	16.90	16.892	-0.047
Static temperature (°R)	2052	--	--	1994	1991.9	-0.105	1937	1936.7	-0.015	1914	1911.9	-0.110	1882	1882.0	0.0
Flow rate (lb/sec)	5.76	--	--	5.76	5.75951	0.0	5.76	5.75951	0.0	5.76	5.75951	0.0	5.76	5.75951	0.0
Jet velocity (ft/sec)	1034	--	--	1049(1076.5)	1091.3	+4.03(+1.37)	1075(1033.5)	1028.4	-4.33(-0.496)	815(811.7)	823.0	+0.982(-1.39)	822(790.6)	779.9	-4.77(-1.35)
Gamma	1.211	1.1825*	--	1.203	1.1437*	--	1.196	1.16607*	--	1.195	1.1447*	--	1.194	1.1637*	--
Gas constant (ft/°R)	31.51	31.158*	--	31.23	30.842*	--	30.93	30.689*	--	30.80	30.828*	--	30.65	30.763*	--
Efficiency Coefficient for Blade Row	--	--	--	--	0.92577	--	--	0.81662*	--	--	0.94752*	--	--	0.8155*	--

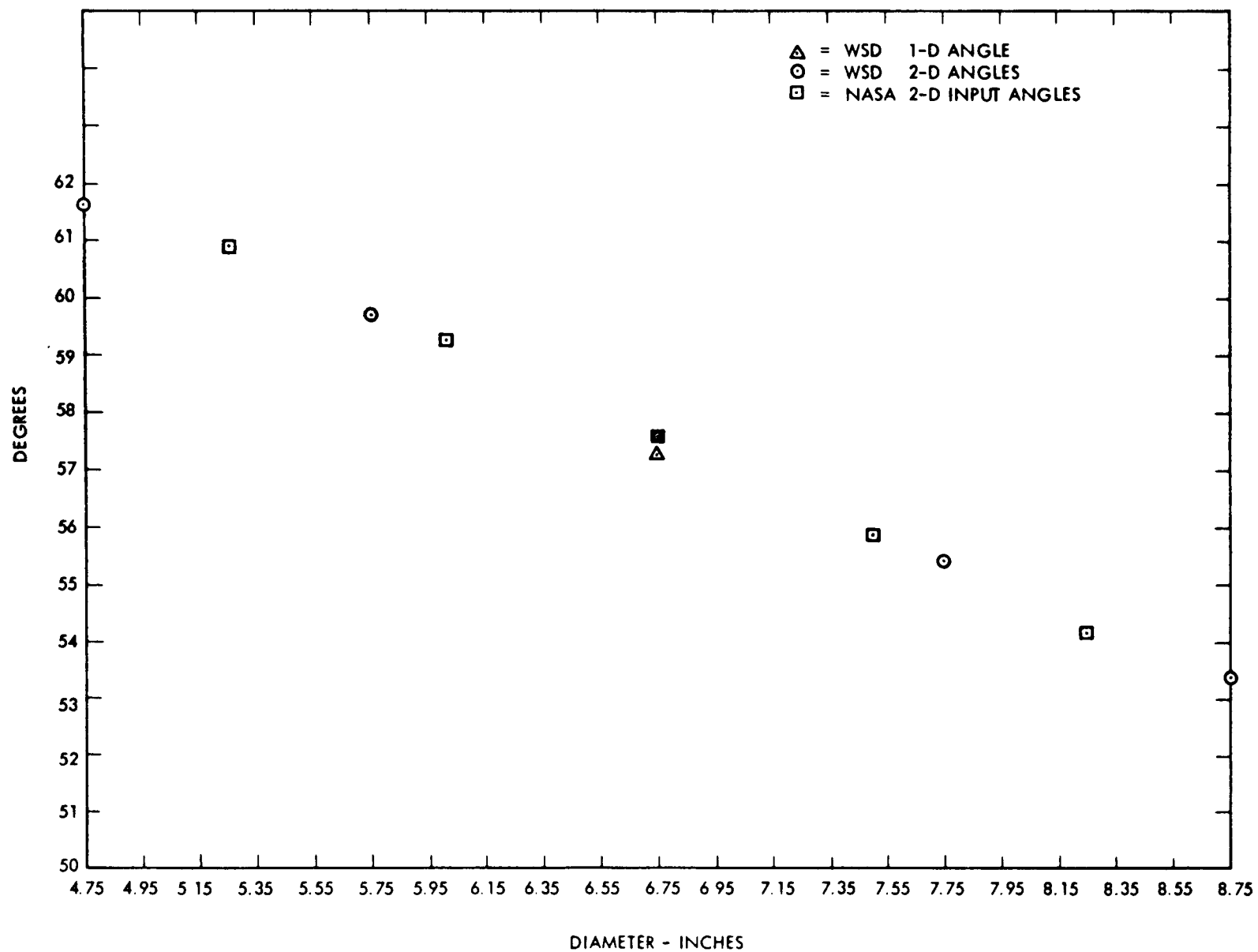
(1) From Reference (2)

(2) Using modified NASA Code (5 radial sectors)

Terms in parentheses are from ⊙ 2-D code. See Reference (2)

Flow angles are with respect to axial direction

* Indicates NASA code input data.



TD 963

Figure 1 B-1. 6th Stator Blade Exit Angles

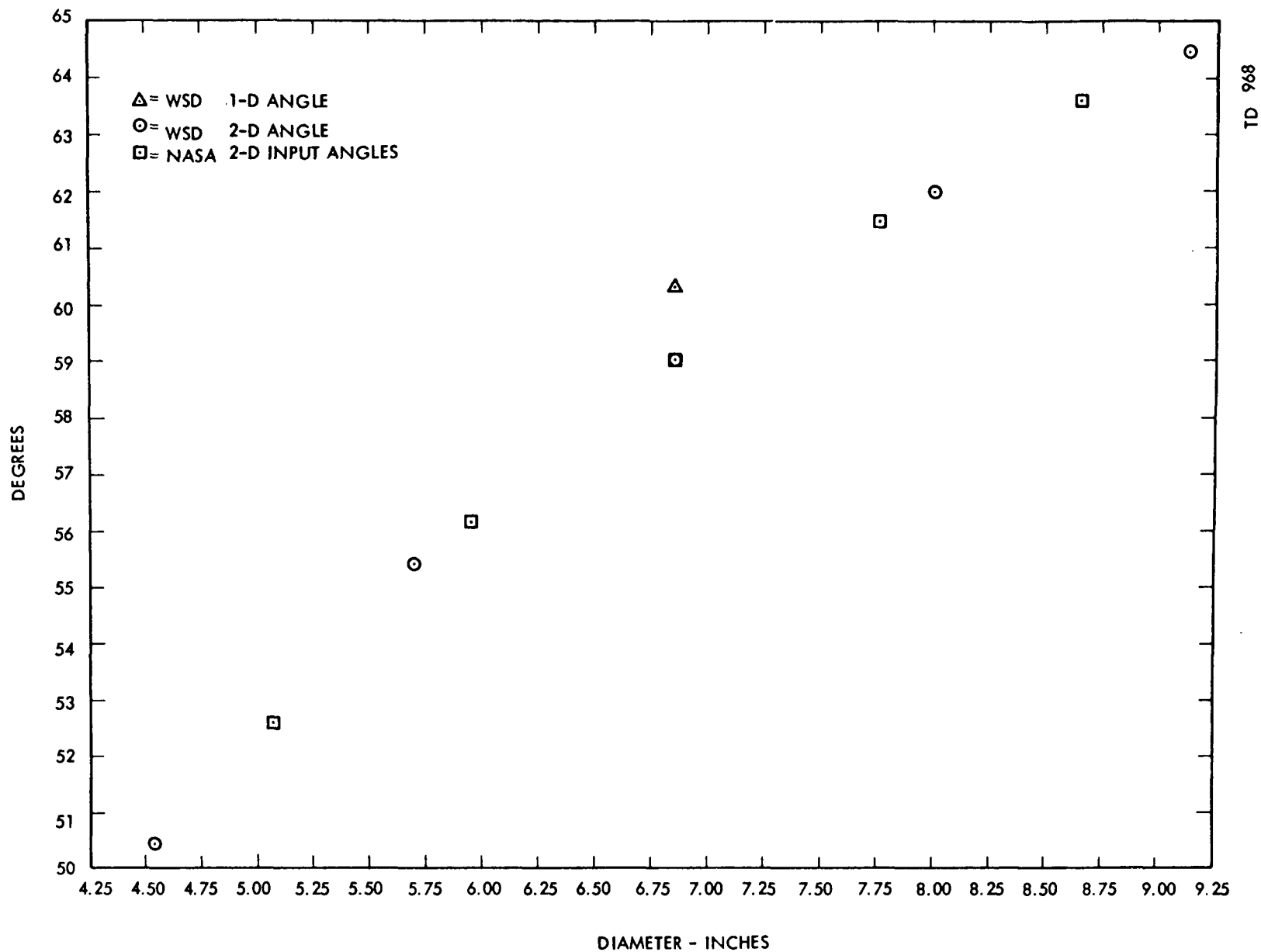


Figure I B-2. 6th Rotor Blade Exit Angles

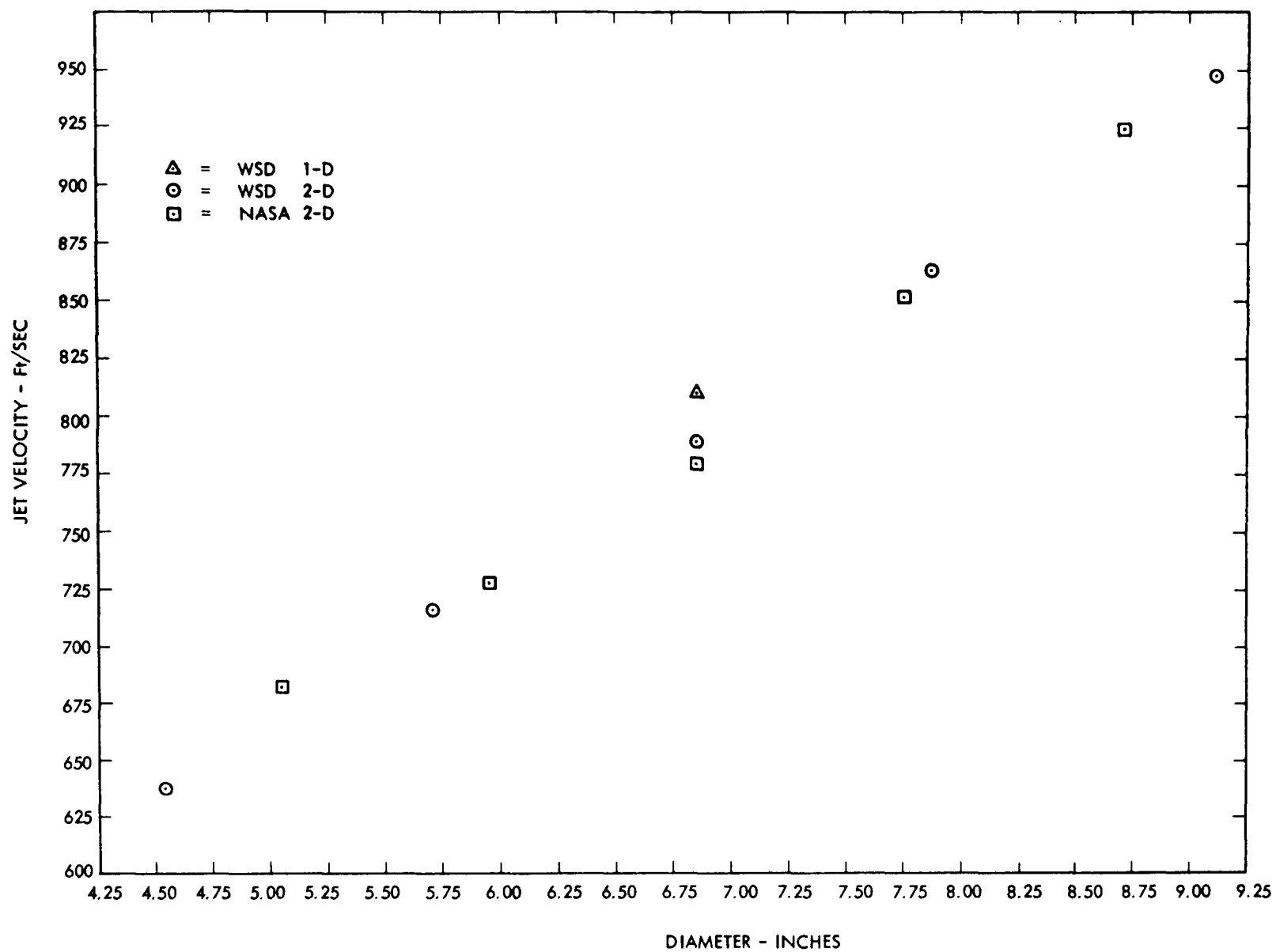


Figure I B-3. 6th Rotor Exit Jet Velocity

TD 966

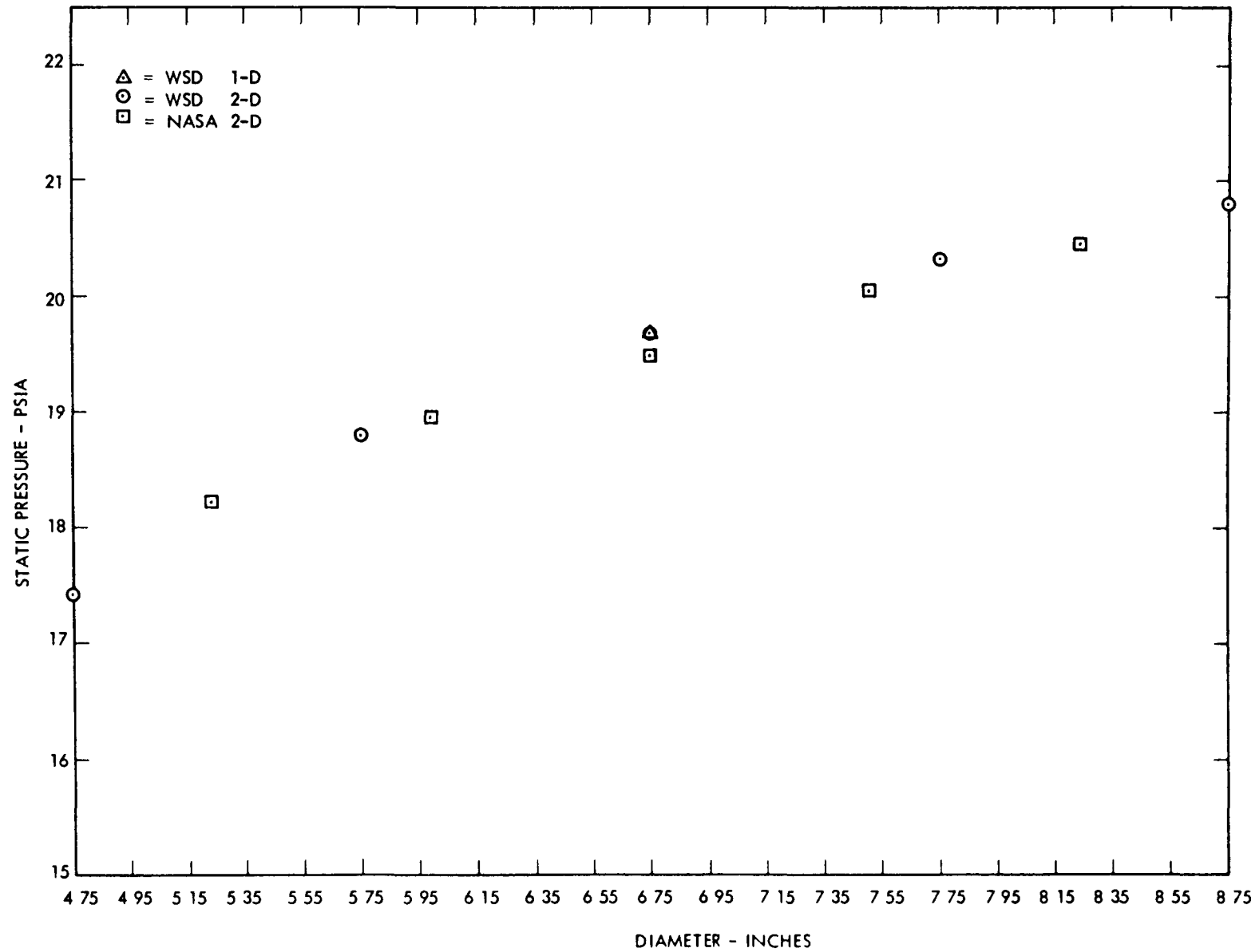


Figure I B-4. 5th Stator Exit Static Pressure

3. Data Input

```

          TURBINE COMPUTER PROGRAM
    TWO STAGE POTASSIUM TURBINE
    FIVE RADIAL SECTORS
$DATAIN
STGCM= 1.000
TTIN= 2067.300 PTIN= 38.828 WAIR= 0.000 FAIR= 0.000
PTPS= 1.377 DELC= 0.000 DELL= 0.000 DELA= 0.000
STG= 2.000 SECT= 5.000 EXPN= 0.000 EXPP= 0.000
PAF= 1.000 SLI= 0.000 AAC= 1.000 RPM= 24000.000
VCTD= 1.000 HSL= 37.600 TSL= 1800.000 PSL= 11.200
GAMSL= 1.618 ENDSTG= 0.000 ENDJOR= 0.000

```

```

          INLET RADIAL PROFILES
PCNM= .200 .200 .200 .200 .200 0.000

```

```

          STANDARD OPTION
          AXIAL STATIONS
STAGE= 1
STA. 0 STA. 1 STA.1A STA. 2 STA.2A
RG= 31.158 30.842 30.842 30.689 30.689 0.000
GAMG= 1.182 1.144 1.144 1.166 1.166 0.000
DR= 5.290 5.150 5.150 5.040 5.040 0.000
DT= 7.510 7.830 7.830 8.280 8.280 0.000
RWG= 1.000 1.000 1.000 1.000 1.000 0.000

```

```

          STATOR RADIAL DISTRIBUTIONS
          ROOT PITCH TIP
SDIA= 0.000 0.000 0.000 0.000 0.000 0.000
SDEA= 66.100 65.600 65.030 64.350 63.650 0.000
SREC= 1.000 1.000 1.000 1.000 1.000 0.000
SETA= .926 .926 .926 .926 .926 0.000
SCF= 1.000 1.000 1.000 1.000 1.000 0.000
SPA= 0.000 0.000 0.000 0.000 0.000 0.000
SESTH= 1.000

```

```

          ROTOR RADIAL DISTRIBUTIONS
ROIA= 48.850 41.500 33.060 22.000 8.500 0.000
RDEA= 61.600 62.650 63.650 64.550 65.350 0.000
RREC= 1.000 1.000 1.000 1.000 1.000 0.000
RETA= .817 .817 .817 .817 .817 0.000
RCF= 1.000 1.000 1.000 1.000 1.000 0.000
RPA= 0.000 0.000 0.000 0.000 0.000 0.000
RTF= 1.000 1.000 1.000 1.000 1.000 0.000
RERTH= 1.000

```

```

          STANDARD OPTION
          AXIAL STATIONS
STAGE= 2
STA. 0 STA. 1 STA.1A STA. 2 STA.2A
RG= 30.689 30.828 30.828 30.763 30.763 0.000
GAMG= 1.166 1.145 1.145 1.164 1.164 0.000
DR= 5.040 4.880 4.880 4.600 4.600 0.000
DT= 8.280 8.620 8.620 9.100 9.100 0.000
RWG= 1.000 1.000 1.000 1.000 1.000 0.000

```

```

          STATOR RADIAL DISTRIBUTIONS
          ROOT PITCH TIP
SDIA= 32.300 29.700 26.540 23.400 20.000 0.000
SDEA= 60.900 59.250 57.570 55.850 54.150 0.000
SREC= 1.000 1.000 1.000 1.000 1.000 0.000
SETA= .948 .948 .948 .948 .948 0.000
SCF= 1.000 1.000 1.000 1.000 1.000 0.000
SPA= 0.000 0.000 0.000 0.000 0.000 0.000
SFSTH= 1.000

```

```

          ROTOR RADIAL DISTRIBUTIONS
ROIA= 32.800 16.000 -2.860 -20.500 -35.000 0.000
RDEA= 52.600 56.100 58.980 61.450 63.600 0.000
RREC= 1.000 1.000 1.000 1.000 1.000 0.000
RETA= .816 .816 .816 .816 .816 0.000
RCF= 1.000 1.000 1.000 1.000 1.000 0.000
RPA= 0.000 0.000 0.000 0.000 0.000 0.000
RTF= 1.000 1.000 1.000 1.000 1.000 0.000
RERTH= 1.000

```


4. Listing of Data Output

NASA TURBINE COMPUTER PROGRAM
TWO STAGE POTASSIUM TURBINE
FIVE RADIAL SECTORS

		CASE 2. 0			
		STAGE PERFORMANCE			
	STAGE 1	STAGE 2	STAGE 3	STAGE 4	
TTBAR 0	2067.3	1955.8			
PTBAR 0	38.828	23.524			
WG 0	5.758	5.758			
DEL H	32.871	17.954			
WRT/P	6.743	10.825			
DH/TTBAR0	.01590	.00918			
N/RT	527.849	542.687			
ETA TT	.82986	.82761			
ETA TS	.73329	.71650			
ETA AT	.80877	.82531			
PT0/PS1	1.377	1.205			
PTBAR0/PTBAR2	1.651	1.331			
PTBAR0/PS2	1.768	1.353			
PTR2/PS2	1.396	1.219			
TTBAR2/TTBAR0	.94606	.96900			
TTR1A/TTBAR0	.97296	.94389			
WG 1	5.758	5.758			
PS 1A	26.198	19.495			
TTR 1A	2011.4	1924.3			
PTR 1A	30.472	20.520			
WG 1A	5.758	5.758			
PS 2	21.963	16.892			
TTBAR 2	1955.8	1895.2			
PTBAR 2	23.524	17.676			
WG 2	5.758	5.758			
WG 2A	5.758	5.758			
UP/VI	.45953	.63653			
UR/VI	.35610	.44398			
PSI P	.86799	.44324			
PSI R	1.44550	.91223			
RX P	.42914	.42908			
RX R	.20904	.12500			
ALPHA 0	0.000	26.147			
I STATOR	0.000	-.393			
BETA 1A	33.910	-1.554			
I ROTOR	.850	1.276			
ALPHA 2A	26.147	-6.951			
OBETA R	109.973	84.948			
M 1	.72584	.55860			
M1 RT	.86583	.70312			
MR 1A	.36920	.29968			
MR1A RT	.55090	.42459			
MR 2	.68866	.52969			
MR2 TIP	.75376	.65843			
E/TH CR	39.397	23.242			
N/RTM CR	26274.6	27306.5			
WRTMCR/D	1.806	2.853			
OVERALL PERFORMANCE					
PSI P	.64847	PSI R	1.19809	DEL H	50.82569
WRT/P	6.74307	N/RT	527.84874	DELH/TTIN	.02459
PT0/PTBAR2	2.19659	PT0/PS2	2.29857	PT0/PAT2A	2.20133
ETA TT	.83219	ETA TS	.78924	ETA TAT	.83138
WNE/60C	791.002	N/RTM CR	26274.585	E/TH CR	60.91617

4. Output Data (continued)

NASA TURBINE COMPUTER PROGRAM
TWO STAGE POTASSIUM TURBINE
FIVE RADIAL SECTIONS

CASE 2. 0
INTER-STAGE PERFORMANCE

STA 1A ROTOR INLET		STAGE 1.				
DIAM 1A	5.418	5.954	6.490	7.026	7.562	
PTR 1A	29.654	30.037	30.472	31.095	31.763	
TTR 1A	2006.4	2008.6	2011.4	2015.7	2020.5	
BETA 1A	48.373	42.053	33.910	22.748	9.389	
I ROTOR	-.477	.553	.850	.748	.889	
R 1A	754.950	644.729	555.112	479.687	433.031	
RU 1A	564.315	431.849	309.689	185.482	70.646	
MR 1A	.50486	.42985	.36920	.31830	.28681	
U 1A	567.371	623.501	679.631	735.760	791.890	
PS 1A	25.666	27.044	28.198	29.351	30.308	
TS 1A	1970.3	1982.3	1991.9	2001.2	2008.6	
CP 1A	.31545	.31545	.31545	.31545	.31545	
RG 1A	30.842	30.842	30.842	30.842	30.842	
GAMG 1A	1.14370	1.14370	1.14370	1.14370	1.14370	
RWG 1A	1.00000	1.00000	1.00000	1.00000	1.00000	
WG 1A	.96621	1.06158	1.15544	1.24444	1.33073	5.75839 TOTAL FLOW
STA 2 ROTOR EXIT						
DIAM 2	5.364	6.012	6.660	7.308	7.956	
PTR 2	29.607	30.095	30.660	31.441	32.298	
TTR 2	2006.0	2009.1	2013.0	2018.6	2024.8	
BETA 2	61.600	62.650	63.650	64.550	65.350	
UBETA	109.973	104.703	97.560	87.298	74.739	
R 2	982.663	1002.932	1028.350	1063.272	1100.854	
RU 2	864.397	890.820	921.504	960.092	1000.536	
MR 2	.65412	.67166	.68866	.71198	.73709	
U 2	561.716	629.575	697.433	765.291	833.150	
RX	.27374	.35918	.42914	.49716	.55252	
DELH	32.437	32.851	33.097	33.027	32.852	
PSI P	2.54805	2.09519	1.74763	1.46741	1.24504	
ETA IT	.82775	.83218	.83334	.83074	.82585	
ETA TS	.71574	.72958	.73822	.73985	.73811	
ETA AT	.79117	.80466	.81300	.81534	.81449	
ZWI INC	-1.34588	-1.19694	-1.06034	-.93090	-.81566	
CP R	.40976	.58675	.70861	.79647	.84527	
PS 2	21.823	21.907	21.963	22.019	22.058	
TS 2	1936.3	1936.5	1936.7	1937.0	1937.4	
CP 2	.27692	.27692	.27692	.27692	.27692	
RG 2	30.689	30.689	30.689	30.689	30.689	

4. Output Data (continued)

NASA TURBINE COMPUTER PROGRAM
TWO STAGE POTASSIUM TURBINE
FIVE RADIAL SECTORS

CASE 2.0
INTER-STAGE PERFORMANCE

STA 0 STATOR INLET		STAGE 1.				
DIAM 0	5.512	5.956	6.400	6.844	7.288	
TT 0	2067.3	2067.3	2067.3	2067.3	2067.3	
PT 0	38.828	38.828	38.828	38.828	38.828	
ALPHA 0	0.000	0.000	0.000	0.000	0.000	
I STATOR	0.000	0.000	0.000	0.000	0.000	
V 0	447.396	447.396	447.396	447.396	447.396	
VU 0	0.000	0.000	0.000	0.000	0.000	
VZ 0	447.396	447.396	447.396	447.396	447.396	
TS 0	2051.9	2051.9	2051.9	2051.9	2051.9	
PS 0	36.991	36.991	36.991	36.991	36.991	
DENS 0	.08332	.08332	.08332	.08332	.08332	
M 0	.28686	.28686	.28686	.28686	.28686	
CP 0	.25944	.25944	.25944	.25944	.25944	
RG 0	31.158	31.158	31.158	31.158	31.158	
GAMG 0	1.18250	1.18250	1.18250	1.18250	1.18250	
RWG 0	1.00000	1.00000	1.00000	1.00000	1.00000	
WG 0	.96621	1.06158	1.15544	1.24444	1.33073	5.75839 TOTAL FLOW
STA 1 STATOR EXIT						
DIAM 1	5.418	5.954	6.490	7.026	7.562	
ALPHA 1	66.100	65.600	65.030	64.350	63.650	
DEL A	66.100	65.600	65.030	64.350	63.650	
V 1	1237.825	1158.855	1091.327	1021.951	962.546	
VU 1	1131.686	1055.350	989.319	921.243	862.536	
VZ 1	501.495	478.729	460.698	442.375	427.230	
TS 1	1970.3	1982.3	1991.9	2001.2	2008.6	
PS 1	25.666	27.044	28.198	29.351	30.308	
DENS 1	.06082	.06370	.06609	.06848	.07045	
M 1	.82778	.77262	.72584	.67812	.63751	
ZWI INC	-.74081	-.75242	-.76537	-.78043	-.79547	
CP S	.86936	.85095	.83194	.81834	.80396	
CP 1	.31545	.31545	.31545	.31545	.31545	
RG 1	30.842	30.842	30.842	30.842	30.842	
GAMG 1	1.14370	1.14370	1.14370	1.14370	1.14370	
RWG 1	1.00000	1.00000	1.00000	1.00000	1.00000	
WG 1	.96621	1.06158	1.15544	1.24444	1.33073	5.75839 TOTAL FLOW
GAMG 2	1.16607	1.16607	1.16607	1.16607	1.16607	
RWG 2	1.00000	1.00000	1.00000	1.00000	1.00000	
WG 2	.93714	1.03939	1.14342	1.25898	1.37946	5.75839 TOTAL FLOW
PT 2A	23.655	23.566	23.492	23.480	23.473	
TT 2A	1957.3	1955.9	1955.0	1955.3	1955.9	
V 2A	556.829	529.680	508.471	496.706	488.699	
VU 2A	302.681	261.246	224.071	194.801	167.386	
ALPHA 2A	32.928	29.552	26.147	23.091	20.030	
MF 2A	.31302	.30858	.30566	.30596	.30742	
VZ 2A	467.379	460.772	456.437	456.913	459.139	
TS 2A	1936.3	1936.5	1936.7	1937.0	1937.4	
PS 2A	21.823	21.907	21.963	22.019	22.058	
DENS 2A	.05288	.05308	.05321	.05334	.05342	
M 2A	.37293	.35473	.34051	.33260	.32721	
CP 2A	.27692	.27692	.27692	.27692	.27692	
RG 2A	30.689	30.689	30.689	30.689	30.689	
GAMG 2A	1.16607	1.16607	1.16607	1.16607	1.16607	
RWG 2A	1.00000	1.00000	1.00000	1.00000	1.00000	
WG 2A	.93714	1.03939	1.14342	1.25898	1.37946	5.75839 TOTAL FLOW

4. Output Data (continued)

NASA TURBINE COMPUTER PROGRAM
TWO STAGE POTASSIUM TURBINE
FIVE RADIAL SECTORS

CASE 2. 0
INTER-STAGE PERFORMANCE

STA 0 STATOR INLET			STAGE 2.			
DIAM 0	5.364	6.012	6.660	7.308	7.956	
TT 0	1957.3	1955.9	1955.0	1955.3	1955.9	
PT 0	23.655	23.566	23.492	23.480	23.473	
ALPHA 0	32.928	29.552	26.147	23.091	20.030	
I STATOR	.628	-.148	-.393	-.309	.030	
V 0	556.829	529.680	508.471	496.706	488.699	
VU 0	302.681	261.246	224.071	194.801	167.386	
VZ 0	467.379	460.772	456.437	456.913	459.139	
TS 0	1936.3	1936.5	1936.7	1937.0	1937.4	
PS 0	21.823	21.907	21.963	22.019	22.058	
DENS 0	.05288	.05308	.05321	.05334	.05342	
M 0	.37293	.35473	.34051	.33260	.32721	
CP 0	.27692	.27692	.27692	.27692	.27692	
RG 0	30.689	30.689	30.689	30.689	30.689	
GAMG 0	1.16607	1.16607	1.16607	1.16607	1.16607	
RWG 0	1.00000	1.00000	1.00000	1.00000	1.00000	
WG 0	.93714	1.03939	1.14342	1.25898	1.37946	5.75839 TOTAL FLOW
STA 1 STATOR EXIT						
DIAM 1	5.254	6.002	6.750	7.498	8.246	
ALPHA 1	60.900	59.250	57.570	55.850	54.150	
DEL A	93.828	88.802	83.717	78.941	74.180	
V 1	972.507	889.612	823.007	758.647	707.705	
VU 1	849.749	764.538	694.656	627.834	573.632	
VZ 1	472.965	454.853	441.353	425.875	414.479	
TS 1	1897.0	1905.4	1911.9	1918.6	1923.9	
PS 1	18.215	18.945	19.495	20.045	20.459	
DENS 1	.04485	.04644	.04763	.04880	.04967	
M 1	.66265	.60483	.55860	.51402	.47883	
ZwI INC	-1.15624	-1.17526	-1.18763	-1.19784	-1.19952	
CP S	.67216	.64549	.61830	.57133	.52315	
CP 1	.31340	.31340	.31340	.31340	.31340	
RG 1	30.828	30.828	30.828	30.828	30.828	
GAMG 1	1.14470	1.14470	1.14470	1.14470	1.14470	
RWG 1	1.00000	1.00000	1.00000	1.00000	1.00000	
WG 1	.90940	1.03454	1.15778	1.27150	1.38518	5.75840 TOTAL FLOW

4. Output Data (continued)

NASA TURBINE COMPUTER PROGRAM TWO STAGE POTASSIUM TURBINE MEAN DIAMETER CALCULATION

CASE 3. 0
INTER-STAGE PERFORMANCE

STA 0	STATOR INLET		STAGE 1.
DIAM 0	5.290	6.400	7.510
TT 0	2067.3	2067.3	2067.3
PT 0	38.828	38.828	38.828
ALPHA 0	0.000	0.000	0.000
I STATOR	0.000	0.000	0.000
V 0	447.396	447.396	447.396
VU 0	0.000	0.000	0.000
VZ 0	447.396	447.396	447.396
TS 0	2051.9	2051.9	2051.9
PS 0	36.991	36.991	36.991
DENS 0	.08332	.08332	.08332
M 0	.28686	.28686	.28686
CP 0	.25944	.25944	.25944
RG 0	31.158	31.158	31.158
GAMG 0	1.18250	1.18250	1.18250
RWG 0	1.00000	1.00000	1.00000
STA 1	STATOR EXIT		
DIAM 1	5.150	6.490	7.830
ALPHA 1	69.720	65.030	60.672
DEL A	69.720	65.030	60.672
V 1	1329.131	1091.327	940.563
VU 1	1246.734	989.319	820.011
VZ 1	460.698	460.698	460.698
TS 1	1955.5	1991.9	2011.3
PS 1	24.344	28.198	30.458
DENS 1	.05812	.06609	.07070
M 1	.89220	.72584	.62254
ZWI INC	-.65026	-.76537	-.85406
CP S	.88670	.83194	.77374
CP 1	.31545	.31545	.31545
RG 1	30.842	30.842	30.842
GAMG 1	1.14370	1.14370	1.14370
RWG 1	1.00000	1.00000	1.00000

C. Wet-Vapor Potassium Turbine * (Mean Diameter Calculation)

1. Comparison of Results

The same modified parameters given in Table I B-1 are used in the one radial sector (mean diameter) calculation. The results are in good agreement with the 5 radial sector calculation as can be seen by comparing the calculated parameters at the mean diameter. In the single sector case the hub and tip values are calculated assuming a free vortex distribution.**

There is a slight inconsistency in the results in that P_S , T_S , ρ , and M for station 0 of the second stage are not identical to those at station 2A of the first stage. The discrepancies are small and thought not to be significant. At this time there is no explanation for this anomaly. The output format for the mean diameter case is slightly different from that using 5 radial sectors.

2. Data Input

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                                TURBINE COMPUTER PROGRAM
                TWO STAGE POTASSIUM TURBINE
                MEAN DIAMETER CALCULATION
SDATAIN
  STGCH=      1.000
  TTIN= 2067.300  PTIN=   38.828  WAIR=      0.000  FAIR=      0.000
  PTPS=      1.377  DELC=      0.000  DELL=      0.000  DELA=      0.000
  STG=       2.000  SECT=      1.000  EXPN=      0.000  EXPP=      0.000
  PAF=       1.000  SLI=      0.000  AACs=      1.000  RPM= 24000.000
  VCTD=      1.000  RSL=   37.600  TSL= 1000.000  PSL=   11.200
  GAMSL=      1.618  ENDSTG=      0.000  ENDJOB=      0.000

                                INLET RADIAL PROFILES
  PCNH=      1.000      0.000      0.000      0.000      0.000      0.000

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* Described in Reference (2)

** Assumes a constant axial velocity component

2. Data Input (continued)

		STANDARD OPTION				
		AXIAL STATIONS				
STAGE=	1	STA. 0	STA. 1	STA. 1A	STA. 2	STA. 2A
RG=	31.158	30.842	30.842	30.689	30.689	0.000
GAMG=	1.182	1.144	1.144	1.166	1.166	0.000
DR=	5.290	5.150	5.150	5.040	5.040	0.000
DT=	7.510	7.830	7.830	8.280	8.280	0.000
RWG=	1.000	1.000	1.000	1.000	1.000	0.000
STATOR RADIAL DISTRIBUTIONS						
	ROOT		PITCH		TIP	
SDIA=	0.000	0.000	0.000	0.000	0.000	0.000
SDEA=	65.030	0.000	0.000	0.000	0.000	0.000
SREC=	1.000	0.000	0.000	0.000	0.000	0.000
SETA=	.926	0.000	0.000	0.000	0.000	0.000
SCF=	1.000	0.000	0.000	0.000	0.000	0.000
SPA=	0.000	0.000	0.000	0.000	0.000	0.000
SESTH=	1.000					
ROTOR RADIAL DISTRIBUTIONS						
RDIA=	33.060	0.000	0.000	0.000	0.000	0.000
RDEA=	63.650	0.000	0.000	0.000	0.000	0.000
RREC=	1.000	0.000	0.000	0.000	0.000	0.000
RETA=	.817	0.000	0.000	0.000	0.000	0.000
RCF=	1.000	0.000	0.000	0.000	0.000	0.000
RPA=	0.000	0.000	0.000	0.000	0.000	0.000
RTF=	1.000	0.000	0.000	0.000	0.000	0.000
RERTH=	1.000					
STANDARD OPTION						
AXIAL STATIONS						
STAGE=	2	STA. 0	STA. 1	STA. 1A	STA. 2	STA. 2A
RG=	30.689	30.828	30.828	30.763	30.763	0.000
GAMG=	1.166	1.145	1.145	1.164	1.164	0.000
DR=	5.040	4.880	4.880	4.600	4.600	0.000
DT=	8.280	8.620	8.620	9.100	9.100	0.000
RWG=	1.000	1.000	1.000	1.000	1.000	0.000
STATOR RADIAL DISTRIBUTIONS						
	ROOT		PITCH		TIP	
SDIA=	26.540	0.000	0.000	0.000	0.000	0.000
SDEA=	57.570	0.000	0.000	0.000	0.000	0.000
SREC=	1.000	0.000	0.000	0.000	0.000	0.000
SETA=	.948	0.000	0.000	0.000	0.000	0.000
SCF=	1.000	0.000	0.000	0.000	0.000	0.000
SPA=	0.000	0.000	0.000	0.000	0.000	0.000
SESTH=	1.000					
ROTOR RADIAL DISTRIBUTIONS						
RDIA=	-2.860	0.000	0.000	0.000	0.000	0.000
RDEA=	58.980	0.000	0.000	0.000	0.000	0.000
RREC=	1.000	0.000	0.000	0.000	0.000	0.000
RETA=	.816	0.000	0.000	0.000	0.000	0.000
RCF=	1.000	0.000	0.000	0.000	0.000	0.000
RPA=	0.000	0.000	0.000	0.000	0.000	0.000
RTF=	1.000	0.000	0.000	0.000	0.000	0.000
RERTH=	1.000					

3. Listing of Data Output

NASA TURBINE COMPUTER PROGRAM
TWO STAGE POTASSIUM TURBINE
MEAN DIAMETER CALCULATION

	CASE 3. 0				
	STAGE PERFORMANCE				
	STAGE 1	STAGE 2	STAGE 3	STAGE 4	
TTBAR 0	2067.3	1952.6			
PTBAR 0	38.828	23.187			
WG 0	5.777	5.777			
DEL H	33.818	19.562			
WRT/P	6.765	11.010			
DM/TTBAR0	.01636	.01002			
N/RT	527.849	543.133			
ETA TT	.83062	.83613			
ETA TS	.73030	.72533			
ETA AT	.80615	.83602			
PT0/PS1	1.377	1.212			
PTBAR0/PTBAR2	1.675	1.362			
PTBAR0/PS2	1.803	1.430			
PTR2/PS2	1.423	1.248			
TTBAR2/TTBAR0	.94451	.96618			
TTT1A/TTBAR0	.97296	.98377			
WG 1	5.777	5.777			
PS 1A	28.198	19.126			
TTR 1A	2011.4	1920.9			
PTR 1A	30.472	20.164			
WG 1A	5.777	5.777			
PS 2	21.540	16.219			
TTBAR 2	1952.6	1886.5			
PTBAR 2	23.187	17.022			
WG 2	5.777	5.777			
WG 2A	5.777	5.777			
UP/VI	.45217	.61276			
UR/VI	.35039	.42713			
PSI P	.89299	.48294			
PSI R	1.48713	.99392			
RX P	.44730	.45539			
RX R	.18072	.10147			
ALPHA 0	0.000	28.042			
I STATOR	0.000	1.502			
HETA 1A	33.910	-.209			
I ROTOR	.850	2.651			
ALPHA 2A	28.042	-1.642			
DBETA R	97.560	58.771			
M 1	.72584	.56766			
M1 RT	.89220	.73490			
MR 1A	.36920	.30442			
MR1A RT	.56669	.44181			
MR 2	.70879	.56012			
MR2 TIP	.78183	.70431			
E/TH CR	40.532	25.366			
N/RTH CR	26274.6	27329.0			
WRTHCRE/D	1.812	2.942			
	OVERALL PERFORMANCE				
PSI P	.68107	PSI R	1.25831	DEL H	53.38050
WRT/P	6.76508	N/RT	527.84874	DELH/TTIN	.02582
PT0/PTBAR2	2.28102	PT0/PS2	2.39404	PT0/PAT2A	2.28112
ETA TT	.83611	ETA TS	.79231	ETA TAT	.83607
WNE/60C	793.584	N/RTH CR	26274.585	E/TH CR	63.97818

3. Output Data (continued)

NASA TURBINE COMPUTER PROGRAM
TWO STAGE POTASSIUM TURBINE
FIVE RADIAL SECTIONS

CASE 2. 0
INTER-STAGE PERFORMANCE

STA 1A ROTOR INLET		STAGE 2.				
DIAM 1A	5.254	6.002	6.750	7.498	8.246	
PTR 1A	19.788	20.105	20.520	21.156	21.871	
TTR 1A	1917.0	1919.8	1924.3	1931.7	1940.2	
HETA 1A	32.348	16.648	-1.584	-20.279	-34.969	
I ROTOR	-.452	.648	1.276	.221	.031	
R 1A	559.846	474.753	441.522	454.016	505.793	
RU 1A	299.552	136.010	-12.202	-157.354	-289.886	
MR 1A	.38147	.32277	.29968	.30767	.34222	
U 1A	550.197	628.527	706.858	785.188	863.518	
PS 1A	18.215	18.945	15.495	20.045	20.459	
TS 1A	1897.0	1905.4	1911.9	1918.6	1923.9	
CP 1A	.31340	.31340	.31340	.31340	.31340	
RG 1A	30.828	30.828	30.828	30.828	30.828	
GAMG 1A	1.14470	1.14470	1.14470	1.14470	1.14470	
RWG 1A	1.00000	1.00000	1.00000	1.00000	1.00000	
WG 1A	.90940	1.03454	1.15778	1.27150	1.38518	5.75840 TOTAL FLOW
STA 2 ROTOR EXIT						
DIAM 2	5.050	5.950	6.850	7.750	8.650	
PTR 2	19.669	20.069	20.600	21.390	22.300	
TTR 2	1915.5	1919.4	1925.2	1934.4	1945.0	
HETA 2	52.600	56.100	58.980	61.450	63.600	
DBETA	84.948	72.748	57.396	41.171	28.631	
R 2	682.758	726.616	775.865	851.208	924.085	
RU 2	542.393	603.100	668.334	747.701	827.713	
MR 2	.46369	.49354	.52969	.57800	.62726	
U 2	528.834	623.082	717.330	811.577	905.825	
RX	.22043	.33978	.42908	.51389	.57638	
DELM	18.960	18.696	18.208	17.619	16.959	
PSI P	1.63020	1.19519	.89898	.60187	.54219	
ETA TT	.85797	.85186	.83888	.81776	.79218	
ETA TS	.74297	.74135	.72943	.70720	.68159	
ETA AT	.85783	.85155	.83705	.81470	.78772	
ZWI INC	-1.43231	-1.11191	-.86859	-.67083	-.51999	
CP R	.32764	.57310	.67947	.71551	.70041	
PS 2	16.889	16.889	16.892	16.895	16.899	
TS 2	1882.4	1881.8	1882.0	1882.9	1884.3	
CP 2	.28103	.28103	.28103	.28103	.28103	
RG 2	30.763	30.763	30.763	30.763	30.763	
GAMG 2	1.16370	1.16370	1.16370	1.16370	1.16370	
RWG 2	1.00000	1.00000	1.00000	1.00000	1.00000	
WG 2	.86346	.99454	1.13553	1.30005	1.46480	5.75839 TOTAL FLOW
PT 2A	17.685	17.650	17.650	17.679	17.708	
TT 2A	1893.3	1892.7	1893.6	1895.8	1898.6	
V 2A	414.913	405.759	404.869	411.799	418.240	
VU 2A	13.559	-14.982	-48.996	-63.877	-78.112	
ALPHA 2A	1.873	-2.823	-6.951	-8.924	-10.764	
MF 2A	.28163	.27527	.27297	.27624	.27890	
VZ 2A	414.691	405.267	401.894	406.814	410.881	
TS 2A	1882.4	1881.8	1882.0	1882.9	1884.3	
PS 2A	16.889	16.889	16.892	16.895	16.899	
DENS 2A	.04200	.04201	.04201	.04200	.04198	
M 2A	.28178	.27561	.27499	.27963	.28390	
CP 2A	.28103	.28103	.28103	.28103	.28103	
RG 2A	30.763	30.763	30.763	30.763	30.763	
GAMG 2A	1.16370	1.16370	1.16370	1.16370	1.16370	
RWG 2A	1.00000	1.00000	1.00000	1.00000	1.00000	
WG 2A	.86346	.99454	1.13553	1.30005	1.46480	5.75839 TOTAL FLOW

3. Output Data (continued)

NASA TURBINE COMPUTER PROGRAM
TWO STAGE POTASSIUM TURBINE
MEAN DIAMETER CALCULATION

CASE 3. 0
INTER-STAGE PERFORMANCE

STA 1A	ROTOR INLET		STAGE 1.
DIAM 1A	5.150	6.490	7.830
PTR 1A	29.190	30.472	32.116
TTR 1A	2000.6	2011.4	2024.7
BETA 1A	56.927	33.910	.007
I ROTOR	.291	.850	1.473
R 1A	844.214	555.112	460.698
RU 1A	707.428	309.689	.056
MR 1A	.56669	.36920	.30493
U 1A	539.306	679.631	819.955
PS 1A	24.344	28.198	30.458
TS 1A	1955.5	1991.9	2011.3
CP 1A	.31545	.31545	.31545
RG 1A	30.842	30.842	30.842
GAMG 1A	1.14370	1.14370	1.14370
RWG 1A	1.00000	1.00000	1.00000
STA 2	ROTOR EXIT		
DIAM 2	5.040	6.660	8.280
BETA 2	61.327	63.650	66.283
DBETA	118.254	97.560	66.290
R 2	977.999	1057.223	1166.653
RU 2	858.070	947.377	1064.121
MR 2	.65625	.70879	.78183
U 2	527.787	697.433	867.079
RX	.21685	.44730	.57436
DELH	33.818	33.818	33.818
PSI P	2.97392	1.78568	1.18905
ETA TT	.83062	.83062	.83062
ETA TS	.73030	.73030	.73030
ETA AT	.80615	.80615	.80615
ZWI INC	-1.54897	-1.06034	-.73654
CP R	.25488	.72431	.84406
PS 2	21.278	21.540	21.664
TS 2	1929.0	1932.4	1933.9
CP 2	.27692	.27692	.27692
RG 2	30.689	30.689	30.689
GAMG 2	1.16607	1.16607	1.16607
RWG 2	1.00000	1.00000	1.00000
PT 2A	23.187	23.187	23.187
TT 2A	1952.6	1952.6	1952.6
V 2A	573.834	531.667	510.506
VU 2A	330.283	249.944	201.042
ALPHA 2A	35.140	28.042	23.192
MF 2A	.31487	.31460	.31447
VZ 2A	469.253	469.253	469.253
TS 2A	1929.0	1932.4	1933.9
PS 2A	21.278	21.540	21.664
DENS 2A	.05176	.05230	.05256
M 2A	.38505	.35644	.34212
CP 2A	.27692	.27692	.27692
RG 2A	30.689	30.689	30.689
GAMG 2A	1.16607	1.16607	1.16607
RWG 2A	1.00000	1.00000	1.00000

3. Output Data (continued)

NASA TURBINE COMPUTER PROGRAM
TWO STAGE POTASSIUM TURBINE
MEAN DIAMETER CALCULATION
CASE 3. 0
INTER-STAGE PERFORMANCE

STA 0	STATOR INLET		STAGE 2.
DIAM 0	5.040	6.660	8.280
TT 0	1952.6	1952.6	1952.6
PT 0	23.187	23.187	23.187
ALPHA 0	35.140	28.042	23.192
I STATOR	1.715	1.502	1.305
V 0	573.834	531.667	510.506
VL 0	330.283	244.944	201.042
VZ 0	469.253	469.253	469.253
TS 0	1928.8	1932.4	1933.8
PS 0	21.266	21.540	21.652
DENS 0	.05173	.05230	.05254
M 0	.38506	.35644	.34213
CP 0	.27692	.27692	.27692
RG 0	30.689	30.689	30.689
GAMG 0	1.16607	1.16607	1.16607
RWG 0	1.00000	1.00000	1.00000
STA 1	STATOR EXIT		
DIAM 1	4.880	6.750	8.620
ALPHA 1	65.329	57.570	50.945
DEL A	100.469	85.612	74.137
V 1	1073.451	835.528	711.146
VL 1	975.465	705.225	552.235
VZ 1	448.068	448.068	448.068
TS 1	1879.2	1908.1	1920.4
PS 1	16.948	14.126	20.120
DENS 1	.04213	.04682	.04894
M 1	.73490	.56766	.48161
ZWI INC	-1.00388	-1.21163	-1.31870
CP S	.71424	.59509	.48467
CP 1	.31340	.31340	.31340
RG 1	30.828	30.828	30.828
GAMG 1	1.14470	1.14470	1.14470
RWG 1	1.00000	1.00000	1.00000

3. Output Data (continued)

NASA TURBINE COMPUTER PROGRAM
TWO STAGE POTASSIUM TURBINE
MEAN DIAMETER CALCULATION
CASE 3. 0
INTER-STAGE PERFORMANCE

STA 1A	ROTOR INLET		STAGE 2.
DIAM 1A	4.880	6.750	8.620
PTR 1A	18.936	20.164	21.894
TTR 1A	1905.7	1920.9	1941.0
BETA 1A	46.027	-1.209	-38.030
I ROTOR	.940	2.651	2.043
R 1A	645.339	448.071	568.840
RU 1A	464.433	-1.633	-350.448
MR 1A	.44181	.30442	.38524
U 1A	511.032	706.858	902.684
PS 1A	16.948	19.126	20.120
TS 1A	1879.2	1908.1	1920.4
CP 1A	.31340	.31340	.31340
RG 1A	30.828	30.828	30.828
GAMG 1A	1.14470	1.14470	1.14470
RWG 1A	1.00000	1.00000	1.00000
STA 2	ROTOR EXIT		
DIAM 2	4.600	6.850	9.100
BETA 2	47.552	58.980	65.806
DBETA	93.579	58.771	27.776
R 2	628.292	822.853	1034.685
RU 2	463.610	705.175	943.799
MR 2	.42768	.56012	.70431
U 2	481.710	717.330	952.949
RX	.12058	.45539	.59732
DELH	19.562	19.562	19.562
PSI P	1.98611	.96582	.56853
ETA TT	.83613	.83613	.83613
ETA TS	.72533	.72533	.72533
ETA AT	.83602	.83602	.83602
ZWI INC	-1.94035	-.88134	-.48493
CP R	-.05500	.70348	.69775
PS 2	16.218	16.219	16.219
TS 2	1873.7	1873.7	1873.7
CP 2	.28103	.28103	.28103
RG 2	30.763	30.763	30.763
GAMG 2	1.16370	1.16370	1.16370
RWG 2	1.00000	1.00000	1.00000
PT 2A	17.022	17.022	17.022
TT 2A	1886.5	1886.5	1886.5
V 2A	424.434	424.222	424.146
VU 2A	-18.101	-12.155	-9.150
ALPHA 2A	-2.444	-1.642	-1.236
MF 2A	.28865	.28865	.28865
VZ 2A	424.047	424.047	424.047
TS 2A	1873.7	1873.7	1873.7
PS 2A	16.218	16.219	16.219
DENS 2A	.04052	.04052	.04052
M 2A	.28892	.28877	.28872
CP 2A	.28103	.28103	.28103
RG 2A	30.763	30.763	30.763
GAMG 2A	1.16370	1.16370	1.16370
RWG 2A	1.00000	1.00000	1.00000

APPENDIX II

LISTING OF CODE

The asterisks in the identification columns (73—80) indicate that the card has been changed from the original listing given in NASA CR-710. Most of the changes are in format statements so as to make the output nomenclature agree with the names of program variables used in the computer code.

```

      PROGRAM JIM(INPUT,OUTPUT,TAPE5=INPUT,TAPE6=OUTPUT)
CNTCP                                     NTCP 001
C      NASA TURBINE PROGRAM                                     NTCP 002
C                                                                 NTCP 003
      REAL MFSTOP                                               NTCP 004
      LOGICAL PREVER,SRFLAG                                     *****
      COMMON SRFLAG                                           *****
      COMMON /SNTCP/G,AJ,PRFC,ICASE,PREVER,MFSTOP,JUMP,LOPIN,ISCASE, NTCP 006
      1KN,GAMF,IP,SCHIT,PTRN,ISECT,KSTG,WTOL,RHOTOL,PRTOL,TRLOOP,LSTG, NTCP 007
      2LBRC,IBWC,ICHOKE,ISORN,CHOKE,PTOPSL(6,8),PTRS2(6,8),TRDIAG,SC,RC, NTCP 008
      3DELPH,PASS,IPC,LOPC,ISS                                  NTCP 009
C                                                                 NTCP 010
      COMMON /SINPUT/ HSL,TSL,PSL,GAMSL,                      *****
      1PTPS,PTIN,TTIN,WAIR,FAIR,DELC,DELL,DELA,AACS,VCTD,STG,SECT,EXPN, NTCP 012
      2EXPP,EXPRE, RPM,PAF,SLI,STGCH,FNDJOR,NAME(10),TITLE(10),PCNH(6),*****
      3RV(6,8),GAM(6,8),DR(6,8),DT(6,8),RWG(6,8),ALPHAS(6,8),ALPHA1(6,8),*****
      4ETARS(6,8),ETAS(6,8),CFS(6,8),ANNO(6,8),BETA1(6,8),BETA2(6,8),ETARNTCP 015
      5R(6,8),ETAR(6,8),CFR(6,8),TFR(6,8),ANDCR(6,8),OMEGAS(6,8),AS0(6,8)NTCP 016
      6,ASMP0(6,8),ACMNO(6,8),A1(6,8),A2(6,8),A3(6,8),A4(6,8),A5(6,8),A6(NTCP 017
      76,8),OMEGAR(6,8),HSIA(6,8),RSMPIA(6,8),HCMNIA(6,8),B1(6,8),B2(6,8)NTCP 018
      8,B3(6,8),B4(6,8),H5(6,8),B6(6,8),SESTHI(8),RERTHI(8)    NTCP 019
C                                                                 NTCP 020
      REAL MR2,M2, MF2                                           NTCP 021
      COMMON /SFLOW2/TS2(6,8),CP2(8),R2(6,8),RHOS2(6,8),BETZE(6,8),RU2(6,NTCP 022
      1,8),VU2(6,8),DPUR2(6,8),VZ2(6,8),MR2(6,8),MF2(6,8),M2(6,8) NTCP 023
C                                                                 NTCP 024
C                                                                 NTCP 025
      DIMENSION CS(8),CR(8)                                     NTCP 026
C                                                                 NTCP 027
C                                                                 NTCP 028
      CALL SLITE(0)                                             NTCP 029
      WAIR=0.0                                                  NTCP 030
      FAIR=0.0                                                  NTCP 031
      PTPS=1.02                                                 NTCP 032
      DELC=0.0                                                  NTCP 033
      DELL=0.0                                                  NTCP 034
      DELA=0.0                                                  NTCP 035
      EXPN=2.0                                                  NTCP 036
      EXPP=2.0                                                  NTCP 037
      EXPRE=0.0                                                 *****
      RV(1,1)=0.0                                              NTCP 039
      PAF=0.0                                                  NTCP 040
      SLI=0.0                                                  NTCP 041
      AACS=1.0                                                  NTCP 042
      SECT=1.0                                                  NTCP 043
      VCTD=0.0                                                  NTCP 044
      WTOL=1.E-04

```

Listing of Code (continued)

RHOTOL=1.E-04	NTCP 045
PRTOL=1.E-06	NTCP 046
PCNM(1)=1.0	NTCP 047
GAM(1,1)=0.0	NTCP 048
RWG(1,1)=1.0	NTCP 049
ETAS(1,1)=0.0	NTCP 050
ALPHA1(1,1)=0.0	NTCP 051
ETAH(1,1)=0.0	NTCP 052
ETA2(1,1)=0.0	NTCP 053
THLOOP=0.	NTCP 054
TRDIAG=0.0	NTCP 055
G=32.17405	NTCP 056
AJ=778.161	NTCP 057
ICASE=0	NTCP 058
1 PREVER=.FALSE.	NTCP 059
READ(5,100) SHFLAG	*****
100 FORMAT(1X,L1)	*****
IF(SHFLAG) WRITE(6,10000)	*****
10000 FORMAT(1H1,39H AN ENTRY HAS BEEN MADE IN MAIN PROGRAM)	*****
CALL INIT	NTCP 060
ISCASE=0	NTCP 061
IF (PREVER) GO TO 1	NTCP 062
DO 25 I=1,8	NTCP 063
CS(I)=0.0	NTCP 064
25 CR(I)=0.0	NTCP 065
PASS=0	NTCP 066
2 PRPC=CS(KN)	NTCP 067
CALL STA01	NTCP 068
IF (PREVER) GO TO 40	NTCP 069
IF(ICHOKE.NF.0) GO TO 3	NTCP 070
IF(SCRIT.EQ.1.) SC=SC+1.	NTCP 071
3 CALL STA1A	NTCP 072
IF (PREVER) GO TO 40	NTCP 073
LOPIN=0	NTCP 074
4 JUMP=0	NTCP 075
PRPC=CR(KN)	NTCP 076
CALL STA2	NTCP 077
CR(KN)=PRPC	NTCP 078
IF (PREVER) GO TO 40	NTCP 079
IF (1.-MF2(1,KN))24,5,5	NTCP 080
5 IF (JUMP)6,6,20	NTCP 081
6 CALL STA2A	NTCP 082
IF (PREVER) GO TO 40	NTCP 083
IF (KN-KSTG)7,9,9	NTCP 084
7 KN=KN+1	NTCP 085
LOPIN=0	NTCP 086
8 JUMP=0	NTCP 087

Listing of Code (continued)

PRPC=CS(KN)	NTCP 088
CALL STAI	NTCP 089
CS(KN)=PRPC	NTCP 090
IF (PREVER) GO TO 40	NTCP 091
IF (JUMP)3,3,20	NTCP 092
9 CALL OVRALL	NTCP 093
IF (VCTD)11,11,10	NTCP 094
10 CALL INSTG	NTCP 095
11 PASS=1.	NTCP 096
IF (TRDIAG)13,13,12	NTCP 097
12 CALL DIAGT(0)	NTCP 098
13 IF (1.-MFSTOP)24,24,14	NTCP 099
14 IF (DELC)24,24,15	NTCP 100
15 IF (DELL)17,17,16	NTCP 101
16 IF (DELPR)24,24,18	NTCP 102
17 IF (CHOKL)24,18,24	NTCP 103
18 ISCASE=ISCASE+1	NTCP 104
19 JL=(ISORH-1)*8+LSTG	NTCP 105
IF (SC.EQ.1.) DELPR=DELL	NTCP 106
PT0PS1(IP,JL)=PT0PS1(IP,JL)+DELPR	NTCP 107
20 LOPIN=1	NTCP 108
KN=LSTG	NTCP 109
IBRC=LBRK	NTCP 110
IPC=0	NTCP 111
IF (KN-1)21,21,22	NTCP 112
21 IF (ISORH-1)2,2,4	NTCP 113
22 IF (ISORH-1)8,8,4	NTCP 114
40 WRITE(6,106)	NTCP 115
24 IF (ENDJOB=1.)1,23,23	NTCP 116
23 IF (SRFLAG) WRITE(6,20000)	*****
20000 FORMAT(1H1,40H AN EXI! HAS BEEN MADE FROM MAIN PROGRAM)	*****
CALL EXIT	*****
106 FORMAT(//3X65H THE PREVIOUS CASE HAS BEEN TERMINATED DUE TO ERRORS	NTCP 118
1- CHECK DUMP.)	NTCP 119
STOP	NTCP 120
END	NTCP 121

Listing of Code (continued)

```

SUBROUTINE INIT                                INIT 001
CINIT                                           INIT 002
C  SUBROUTINE FOR INITIALIZATION OF INPUT DATA  INIT 003
C                                           INIT 004
REAL MFSTOP                                    INIT 005
LOGICAL PREVER,SRFLAG                          *****
COMMON SRFLAG                                *****
COMMON /SNTCP/G,AJ,PRFC,ICASE,PREVER,MFSTOP,JUMP,LOPIN,ISCASE,  INIT 007
1KN,GAMF,IP,SCHIT,PTRN,ISECT,KSTG,WIOL,WHOTOL,PTOL,TRLOOP,LSTG,  INIT 008
2LBRC,IBRC,ICHOKE,ISORH,CHOKE,PTOPSI(6,8),PTRS2(6,8),TRDIAG,SC,RC,  INIT 009
3DELPH,PASS,IPC,LOPC,ISS                      INIT 010
C                                           INIT 011
COMMON /SINIT/H1(6,8),H2(6,8),DP0(6,8),DP1(6,8),DP1A(6,8),DP2(6,8)  INIT 012
1,DP2A(6,8),CSALF1(6,8),ALF1(6,8),CSHET2(6,8),HET2(6,8),RADSU(6,8),  INIT 013
2RADRD(6,8),ANN1(6,8),ANN2(6,8),ANN2A(6,8),ANN1A(6,8),U1A(6,8),  INIT 014
3U2(6,8),ANN0(6,8),PT0(6,8),TT0(6,8),ALPHA0(6,8),PTP(6,8)        INIT 015
C                                           INIT 016
COMMON /SINPUT/ RSL,TSL,PSL,GAMSL,          *****
1PTPS,PTIN,TTIN,WAIR,FAIR,DELC,DELL,DELA,AACS,VCTD,STG,SECT,EXPN,  INIT 018
2EXPP,EXPRE,  RPM,PAF,SLI,STGCH,ENDJOB,NAME(10),TITLE(10),PCNH(6), *****
3RV(6,8),GAM(6,8),DR(6,8),DT(6,8),RWG(6,8),ALPHAS(6,8),ALPHA1(6,8), *****
4ETARS(6,8),ETAS(6,8),CFS(6,8),AND0(6,8),HETA1(6,8),HETA2(6,8),ETAR  INIT 021
5R(6,8),ETAR(6,8),CFR(6,8),TFR(6,8),ANDCH(6,8),OMEGAS(6,8),AS0(6,8)  INIT 022
6,ASMP0(6,8),ACMN0(6,8),A1(6,8),A2(6,8),A3(6,8),A4(6,8),A5(6,8),A6(  INIT 023
7A(6,8),OMEGAR(6,8),BSIA(6,8),HSMPIA(6,8),HCMNIA(6,8),R1(6,8),R2(6,8)  INIT 024
8,B3(6,8),H4(6,8),B5(6,8),B6(6,8),SESTHI(8),RETHI(8)            INIT 025
C                                           INIT 026
C  DIMENSION                                F1A(6,8),H0(6,8),H2A(6,8)  INIT 027
C                                           INIT 028
C  READ INPUT DATA, CHECK FOR ERRORS,        INIT 029
C  SKIP CHANGE CASES IF BASIC CASE HAS AN ERROR  INIT 030
C  IF(SHFLAG) WRITE(6,10000)                *****
10000 FORMAT(44H AN ENTRY HAS BEEN MADE IN SUBROUTINE INIT )    *****
3  CALL INPUT                                INIT 031
ICASE=ICASE+1                                INIT 032
IF(STGCH)5,5,4                                INIT 033
4  IK=1                                        INIT 034
5  CALL CHECK(L)                              INIT 035
GO TO(6,8),L                                INIT 036
6  WRITE(6,100) ICASE                        INIT 037
IF(STGCH)3,3,7                                INIT 038
7  IK=2                                        INIT 039
GO TO 3                                        INIT 040
8  IF (IK-2)9,3,3                            INIT 041
C  INITIALIZE INDEX REGISTERS AND FORKS      INIT 042
9  ISECT=SECT+.0001                          INIT 043

```


Listing of Code (continued)

KSTG= STG+.0001	INIT 044
LUPC=0	INIT 045
CHOKE=0.	INIT 046
ICHOKE=0	INIT 047
ISOHR=1	INIT 048
KN=1	INIT 049
LSTG=1	INIT 050
IHRC=1	INIT 051
LHRC=1	INIT 052
DELPR=DELC	INIT 053
SC=0.0	INIT 054
RC=0.0	INIT 055
PRPC=0.0	INIT 056
IPC=0	INIT 057
ISS=0	INIT 058
PTRN=0.0	INIT 059
C TEST STAGE LOSS INDICATOR	INIT 060
IF(SLI)13,13,11	INIT 061
11 DO 12 I=1,ISECT	INIT 062
DO 12 J=1,KSTG	INIT 063
ETARS(I,J)=ETARS(I,1)	INIT 064
ETAS(I,J)=ETAS(I,1)	INIT 065
CFS(I,J)=CFS(I,1)	INIT 066
ETARR(I,J)=ETARR(I,1)	INIT 067
ETAR(I,J)=ETAR(I,1)	INIT 068
CFR(I,J)=CFR(I,1)	INIT 069
TFR(I,J)=TFR(I,1)	INIT 070
12 CONTINUE	INIT 071
C TEST FOR EQUAL SECTORS	INIT 072
13 IF(PCNH(1)-1.)16,14,14	INIT 073
14 DO 15 I=1,ISECT	INIT 074
15 PCNH(I)= 1./SECT	INIT 075
C SET UP SECTOR HEIGHT, PITCH DIAMETER, ANNULUS AREA,	INIT 076
C PITCHLINE WHEEL SPEED	INIT 077
16 DO 19 K=1,KSTG	INIT 078
SH0=DT(1,K)-DR(1,K)	INIT 079
SH1=DT(2,K)-DR(2,K)	INIT 080
SH1A=DT(3,K)-DR(3,K)	INIT 081
SH2=DT(4,K)-DR(4,K)	INIT 082
SH2A=DT(5,K)-DR(5,K)	INIT 083
DO 18 I=1,ISECT	INIT 084
H0(I,K)=.5*PCNH(I)*SH0	INIT 085
H1(I,K)=.5*PCNH(I)*SH1	INIT 086
H1A(I,K)=.5*PCNH(I)*SH1A	INIT 087
H2(I,K)=.5*PCNH(I)*SH2	INIT 088
H2A(I,K)=.5*PCNH(I)*SH2A	INIT 089
IF(I-1)20,20,17	INIT 090

Listing of Code (continued)

```

20 DP0(I,K)=DH(1,K)+    H0(I,K)          INIT 091
   DP1(I,K)=DH(2,K)+    H1(I,K)          INIT 092
   DP1A(I,K)=DH(3,K)+    F1A(I,K)        INIT 093
   DP2(I,K)=DH(4,K)+    H2(I,K)          INIT 094
   DP2A(I,K)=DH(5,K)+    F2A(I,K)        INIT 095
   GO TO 21                              INIT 096
17 DP0(I,K)=          H0(I-1,K)+    H0(I,K)+DP0(I-1,K)  INIT 097
   DP1(I,K)=          H1(I-1,K)+    H1(I,K)+CP1(I-1,K)  INIT 098
   DP1A(I,K)=          H1A(I-1,K)+    H1A(I,K)+DP1A(I-1,K)  INIT 099
   DP2(I,K)=          H2(I-1,K)+    H2(I,K)+CP2(I-1,K)  INIT 100
   DP2A(I,K)=          H2A(I-1,K)+    H2A(I,K)+DP2A(I-1,K)  INIT 101
21 ANN0(I,K)=.0218166*DP0(I,K)*H0(I,K)      *****
   ANN1(I,K)=.0218166*DP1(I,K)*H1(I,K)      *****
   ANN1A(I,K)=DP1A(I,K)*F1A(I,K)*.0218166    *****
   ANN2(I,K)=.0218166*DP2(I,K)*H2(I,K)      *****
   ANN2A(I,K)=.0218166*DP2A(I,K)*H2A(I,K)    *****
   U1A(I,K)= 3.14159*DP1A(I,K)*RPM/720.      INIT 107
   U2(I,K)= 3.14159*DP2(I,K)*RPM/720.      INIT 108
18 CONTINUE                              INIT 109
19 CONTINUE                              INIT 110
C      DEFINE PITCHLINE INDEX              INIT 111
   IT=ISECT-2*(ISECT/2)                    INIT 112
   IF(IT)22,22,23                          INIT 113
22 IP=ISECT/2                              INIT 114
   GO TO 24                                INIT 115
23 IP=(ISECT+1)/2                          INIT 116
C      CALCULATE INLET AND EXIT ANGLES IN RADIANS  INIT 117
24 IF (ALPHA1(1,1))25,25,27                INIT 118
25 SDEAF=0.                                INIT 119
   DO 26 K=1,KSTG                          INIT 120
   DO 26 I=1,ISECT                          INIT 121
   CSALF1(I,K)=ANDU(I,K)*CFS(I,K)/(SESTHI(K)*3.14159*DP1(I,K)*
1SQRT(ETAS(I,K)))                          INIT 122
26 ALF1(I,K)=ATAN2(SQRT(1.-CSALF1(I,K)*CSALF1(I,K)),CSALF1(I,K))  INIT 124
   GO TO 31                                INIT 125
27 DO 28 K=1,KSTG                          INIT 126
   DO 28 I=1,ISECT                          INIT 127
   ALF1(I,K)=    ALPHA1(I,K)*.01745328      *****
28 CSALF1(I,K)=COS(ALF1(I,K))              INIT 129
31 IF (BETA2(1,1))29,29,32                 INIT 130
29 RDEAF=0.                                INIT 131
   DO 30 K=1,KSTG                          INIT 132
   DO 30 I=1,ISECT                          INIT 133
   CSRET2(I,K)=ANDUR(I,K)*CFR(I,K)/(RERTHI(K)*3.14159*DP2(I,K)*
1SQRT(ETAR(I,K)))                          INIT 134
30 RET2(I,K)=ATAN2(SQRT(1.-CSRET2(I,K)*CSRET2(I,K)),CSRET2(I,K))  INIT 136
   GO TO 34                                INIT 137

```

Listing of Code (continued)

32 DO 33 K=1,KSTG	INIT 138
DO 33 I=1,ISECT	INIT 139
BET2(I,K)= HETA2(I,K)*.01745328	*****
33 CSBET2(I,K)=COS(BET2(I,K))	INIT 141
34 DO 35 K=1,KSTG	INIT 142
DO 35 I=1,ISECT	INIT 143
PTP(I,K)=PTIN	INIT 144
PT0(I,K)=PTIN	INIT 145
TT0(I,K)=TTIN	INIT 146
ALPHA0(I,K)=0.0	INIT 147
PT0PS1(I,K)=PTPS	INIT 148
RADSD(I,K)=ALPHAS(I,K)*.01745328	*****
35 RADHD(I,K)=HETA1(I,K)*.01745328	*****
IF(RV(1,1))36,36,37	*****
36 CALL R(PIN,TTIN,FAIR,wAIR,RV(1,1))	*****
GAMF=0.0	INIT 153
GO TO 38	INIT 154
37 GAMF=1.0	INIT 155
38 CALL CHECK(J)	INIT 156
GO TO (39,40),J	INIT 157
39 GO TO 3	INIT 158
40 IF(SRFLAG) WRITE(6,20000)	*****
20000 FORMAT(45H AN EXIT HAS BEEN MADE FROM SUBROUTINE INIT)	*****
RETURN	*****
100 FORMAT(28X,6HCASE I5,13H HAS AN ERROR)	INIT 160
END	INIT 161

Listing of Code (continued)

```

SUBROUTINE INPUT
CINPUT
C*****
C
    REAL MFSTOP
    LOGICAL PHEVER,SHFLAG
    COMMON SHFLAG
    COMMON /SNTCP/G,AJ,PRFC,ICASE,PREVER,MFSTOP,JUMP,LOPIN,ISCASE,
1KN,GAMF,IP,SCRIT,PTRN,ISECT,KSTG,WIOL,RHOTOL,PRTOL,TRLOOP,LSTG,
2LBRC,IHRC,ICHOKE,ISURH,CHOKE,PTOPSL(6,8),PTRS2(6,8),TRDIAG,SC,RC,
3DELPH,PASS,IPC,LOPC,ISS
C
    COMMON /SINPUT/ RSL,TSL,PSL,GAMSL,
1PTPS,PTIN,TTIN,WAIR,FAIR,DELC,DELL,DELA,AACS,VCTD,STG,SECT,EXPN,
2EXPP,EXPHE, RPM,PAF,SLI,STGCH,ENDJOB,NAME(10),TITLE(10),PCNH(4),
3RV(6,8),GAM(6,8),SR(6,8),ST(6,8),SWG(6,8),ALPHAS(6,8),ALPHA1(6,8),
4ETAS(6,8),ETAS(6,8),CFS(6,8),ANNO(6,8),BETA1(6,8),BETA2(6,8),ETARINPT
5R(6,8),ETAR(6,8),CFR(6,8),TFR(6,8),ANDQR(6,8),OMEGAS(6,8),ASO(6,8),
6ASMP0(6,8),ACMN0(6,8),A1(6,8),A2(6,8),A3(6,8),A4(6,8),A5(6,8),A6(
76,8),OMEGAR(6,8),BSIA(6,8),RSMPIA(6,8),HCMNIA(6,8),B1(6,8),B2(6,8),
8,H3(6,8),R4(6,8),B5(6,8),B6(6,8),SESTHI(8),RERTHI(8)
C
    DIMENSION X(6,8,38),Y(6,38)
C
    EQUIVALENCE (X(1,1,1),RV(1,1)),(Y(1,1),RG(1))
C
    COMMON HG(6),
1GAMG(6),DR(6),DT(6),RWG(6),SDIA(6),SDEA(6),SREC(6),SETA(6),
1SCF(6),SPA(6),RDIA(6),RDEA(6),RREC(6),RETA(6),RCF(6),RTF(6),RPA(6)
2,STPLC(6),SINR(6),SINMP(6),SINMN(6),SCPS(6),SCPC(6),SCPQ(6),SCNS(6)
3,SCNC(6),SCNQ(6),RTPLC(6),RINR(6),RINMP(6),RINMN(6),RCPS(6),RCPC(
46),RCPQ(6),RCNS(6),RCNC(6),RCNQ(6)
C
    NAMFLIST/DATAIN/ RSL,TSL,PSL,GAMSL,
1PTPS,PTIN,TTIN,WAIR,FAIR,DELC,DELL,DELA,AACS,VCTD
1,STG,SECT,STAGE,EXPN,EXPP,EXPHE,RPM,PAF,SLI,ENDSTG,ENDJOB,PCNH,RG,
2GAMG,DR,DT,RWG,SDIA,SLEA,SREC,SETA,SCF,SPA,RDIA,RDEA,RREC,RETA,RCF
3,RTF,RPA,STPLC,SINR,SINMP,SINMN,SCPS,SCPC,SCPQ,SCNS,SCNC,SCNQ,RTPL
4C,RINR,RINMP,RINMN,RCPS,RCPC,RCPO,RCNS,RCNC,RCNQ,SESTH,RERTH,
5WTOL,RHOTOL,PRTOL,TRLOOP,TRDIAG,STGCH
C
    DATA BLANKS/66666666/
C
    READ THE HEADING CARDS EVERY TIME ENTRY IS MADE
    IF (SHFLAG) WRITE(6,10000)

```

Listing of Code (continued)

```

10000 FORMAT(44H AN ENTRY HAS BEEN MADE IN SUBROUTINE INPUT )      *****
10 READ(5,6669) (NAME(I),I=1,10)                                  INPT 043
20 READ(5,6669) (TITLE(I),I=1,10)                                INPT 044
   J=0                                                            INPT 045
   *****
30 DO 25 L=1,38                                                    INPT 047
   DO 25 I=1,6                                                      INPT 048
25 Y(I,L)=BLANKS                                                  INPT 049
   SESTH=BLANKS                                                    INPT 050
   RERTH=BLANKS                                                    INPT 051
   READ(5,DATAIN)                                                  INPT 052
40 K=STAGE+.0001                                                  INPT 053
50 ISECT=SECT+.0001                                              *****
60 DO 80 L=1,38                                                    INPT 055
70 DO 80 I=1,6                                                      INPT 056
   IF (Y(I,L).NE.BLANKS) GO TO 71                                INPT 057
   Y(I,L)=0.0                                                       INPT 058
   GO TO 80                                                         INPT 059
71 X(I,K,L)=Y(I,L)                                                INPT 060
80 CONTINUE                                                       INPT 061
   IF (SESTH.EQ.BLANKS) GO TO 95                                INPT 062
90 SESTHI(K)=SESTH                                                INPT 063
   GO TO 96                                                         INPT 064
95 SESTH=0.                                                         INPT 065
96 IF (RERTH.EQ.BLANKS) GO TO 105                                INPT 066
100 RERTHI(K)=RERTH                                                INPT 067
   GO TO 110                                                       INPT 068
105 RERTH=0.                                                       INPT 069
110 IF (K-1)120,120,130
120 WRITE(6,6670)NAME,TITLE,STGCH,TTIN,PTIN,WAIR,FAIR,PTPS,DELC,DELL, *****
   1DELA,STG,SECT,EXPN,EXPP,   PAF,S,I,AACS,RPM,VCTD,RSL,TSL,PSL,GAMSL *****
   2,ENDSTG,ENDJOB,PCNH                                           *****
   J=J+1                                                            INPT 073
130 WRITE(6,6671) K,RG,GAMG,DR,DT,RWG,SDIA,SDEA,SREC,SETA,SCF,SPA, *****
   1SESTH,                                                         INPT 075
   1RDIA,RDEA,RREC,RETA,RCF,RPA,RTF,RERTH                        *****
140 IF (OMEGAS(1,K))160,160,150                                INPT 077
150 WRITE(6,6672)STPLC,SINR,SINMP,SINMN,SCPS,SCPC,SCPQ,SCNS,SCNC,SCNQ, INPT 078
   1RTPLC,RINR,RINMP,RINMN,RCPS,RCPC,RCPQ,RCNS,RCNC,RCNQ        INPT 079
160 J=J+1                                                         INPT 080
180 AM= J-2*(J/2)                                                  INPT 081
190 IF (AM)200,210,200                                            INPT 082
200 WRITE(6,6673)                                                  INPT 083
210 IF (ENDSTG-1.)30,170,170                                     INPT 084
170 IF (SRFLAG) WRITE(6,20000)                                  *****
20000 FORMAT(1H1,45H AN EXIT HAS BEEN MADE FROM SUBROUTINE INPUT ) *****
   RETURN                                                         *****
6669 FORMAT(10A6)                                                  INPT 086

```

Listing of Code (continued)

```

6670 FORMAT (1H1,24X,24HTURBINE COMPUTER PROGRAM/6X,10A6/6X,10A6/2X, INPT 087
17HSCATAIN/2X,7H STGCH=F10.3/2X7H TTIN=F10.3,1X,7H PTIN=F10.3,2X,*****
16H *AIR=F10.3,2X, *****
25HFAIR=F10.3/2X,7H PIPS=F10.3,1X,7H DELC=F10.3,2X,6H DELL=F10.3,*****
32X,5HDELA=F10.3/2X,7H STG=F10.3,1X,7H SECT=F10.3,2X,6H EXPN=F10.3,*****
4,3,2X,5HEXPP=F10.3/2X, 7H PAF=F10.3,2X,6H SLI=*****
5F10.3,3X,5HAACS=F10.3, 2X,5H RPM=F10.3/2X,7H VCTD=F10.3,4X,4HRS*****
6=F10.3,4X,4HTSL=F10.3,3X,4HPSL=F10.3/2X,7H GAMSL=F10.3,1X,7HENDSTG*****
7=F10.3,1X,7HENDJOB=F10.3//25X,21HINLET RADIAL PROFILES *****
8 /4X,5HPCNH=6(F8.3,2X)/1H1) *****
6671 FORMAT(28X,15HSTANDARD OPTION/3X,6HSTAGE=13,16X,14HAXIAL STATIONS/*****
111X,6HSTA. 04X,6HSTA. 14X,6HSTA. 1A4X,6HSTA. 23X,7H STA.2A/ *****
23X,6H RG=6(F8.3,2X)/ *****
33X,6H GAMG=6(F8.3,2X)/3X,6H DR=6(F8.3,2X)/3X,6H DT=6(F8.3,2X)/INPT 097
33X,6H RWG=6(F8.3,2X)//22X,27HSTATOR RADIAL DISTRIBUTIONS/ INPT 098
413X,4HROOT,15X,5HPITCH,16X,3HTIP/ *****
53X,6H SDIA=6(F8.3,2X)/3X,6H SDEA=6(F8.3,2X)/3X,6H SREC=6(F8.3,2X)/INPT 100
63X,6H SFTA=6(F8.3,2X)/3X,6H SCF=6(F8.3,2X)/3X,6H SPA=6(F8.3,2X)/INPT 101
73X,6HSESTH=F8.3//22X,26HROTOR RADIAL DISTRIBUTIONS/ *****
83X,6H RDIA=6(F8.3,2X)/3X,6H RDEA=6(F8.3,2X)/3X,6H RREC=6(F8.3,2X)/INPT 103
93X,6H RETA=6(F8.3,2X)/3X,6H RCF=6(F8.3,2X)/3X,6H RPA=6(F8.3,2X)/*****
13X,6H RTF=6(F8.3,2X)/3X,6HRETRH=1F8.3/) *****
6672 FORMAT(1/25X,23HLOSS COEFFICIENT OPTION/22X,27HSTATOR RADIAL DISTRINPT 106
1RUTIONS/ INPT 107
23X,6HSTPLC=6(F8.3,2X)/3X,6H SINR=6(F8.3,2X)/3X,6HSINMP=6(F8.3,2X)/INPT 108
33X,6HSINMN=6(F8.3,2X)/3X,6H SCPS=6(F8.3,2X)/3X,6H SCPC=6(F8.3,2X)/INPT 109
43X,6H SCPQ=6(F8.3,2X)/3X,6H SCNS=6(F8.3,2X)/3X,6H SCNC=6(F8.3,2X)/INPT 110
53X,6H SCNQ=6(F8.3,2X)/023X,26HROTOR RADIAL DISTRIBUTIONS/ INPT 111
63X,6HRTPLC=6(F8.3,2X)/3X,6H RJNR=6(F8.3,2X)/3X,6HRJNMP=6(F8.3,2X)/INPT 112
73X,6HRJNMN=6(F8.3,2X)/3X,6H RCPS=6(F8.3,2X)/3X,6H RCPC=6(F8.3,2X)/INPT 113
83X,6H RCPQ=6(F8.3,2X)/3X,6H RCNS=6(F8.3,2X)/3X,6H RCNC=6(F8.3,2X)/INPT 114
93X,6H RCNQ=6(F8.3,2X) INPT 115
6673 FORMAT (1H1) INPT 116
END INPT 117

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Listing of Code (continued)

```

      SUBROUTINE STA01
CSTA01
C      ESTABLISH FIRST STATOR EXIT FLOW, ADJUST FLOWS FOR COOLING
C      AIR INJECTION BETWEEN STATIONS 0 AND 1, FIND INLET
C      MACH NUMBER AND INCIDENCE ANGLE LOSS AT STATION 0,
C      ADJUST PR, GET NEW FLOW AT STATION 1 FOR FINAL RESULT.
C
      REAL MFSTOP
      LOGICAL PREVER,SRFLAG
      COMMON SRFLAG
      COMMON /SNTCP/G,AJ,PRFC,ICASE,PREVER,MFSTOP,JUMP,LOPIN,ISCASE,
1KN,GAMF,IP,SCRIT,PTRN,ISECT,KSTG,WIOL,HOTOL,PRTOL,TRLOOP,LSTG,
2LBRC,IHRC,ICHOKE,ISORN,CHOKF,PTOPSI(6,8),PTRS2(6,8),TRDIAG,SC,RC,
3DELPR,PASS,IPC,LOPC,ISS
C
      COMMON /SINIT/H1(6,8),H2(6,8),DP0(6,8),DP1(6,8),DP1A(6,8),DP2(6,8),
1,DP2A(6,8),CSALF1(6,8),ALF1(6,8),CSHET2(6,8),HET2(6,8),RADSD(6,8),
2RADHD(6,8),ANN1(6,8),ANN2(6,8),ANN2A(6,8),ANN1A(6,8),U1A(6,8),
3U2(6,8),ANN0(6,8),PT0(6,8),TT0(6,8),ALPHA0(6,8),PTP(6,8)
C
      COMMON /SINPUT/ RSL,TSL,PSL,GAMSI,
1PTPS,PTIN,TTIN,WAIR,FAIR,DELC,DEIL,DELA,AACS,VCTD,STG,SECT,EXPN,
2EXPP,EXPRF, RPM,PAF,SLI,STGCH,FNDJOH,NAME(10),TITLE(10),PCNH(6),
3RV(6,8),GAM(6,8),DH(6,8),DT(6,8),RWG(6,8),ALPHAS(6,8),ALPHA1(6,8),
4ETAS(6,8),ETAS(6,8),CFS(6,8),ANN0(6,8),BETA1(6,8),HETA2(6,8),ETARST01
5R(6,8),FTAR(6,8),CFR(6,8),TFR(6,8),ANDCH(6,8),OMEGAS(6,8),AS0(6,8),
6,ASMP0(6,8),ACMN0(6,8),A1(6,8),A2(6,8),A3(6,8),A4(6,8),A5(6,8),A6(
76,8),OMEGAR(6,8),HSIA(6,8),RSMPIA(6,8),HCMNIA(6,8),R1(6,8),R2(6,8),
8,R3(6,8),R4(6,8),R5(6,8),R6(6,8),SESTHI(8),RERTHI(8)
C
      REAL M0
      COMMON /SSTA01/CP0(8),
18),VU0(6,8),VZ0(6,8),RHOS0(6,8),PS1(6,8),WGT1(8),TA1(8),WGL(6,8),
2CPDH1(6,8),SI(6,8), CP1(8),PHI1(6,8),TS1(6,8),V1(6,8),
3,RHOS1(6,8),ALFIE(6,8),VU1(6,8),VZ1(6,8),M0(6,8),WGT0(8),WGL(6,8)
C
      DIMENSION TA0(8), TT0TS0(6,8),PT0PS0(6,8),FFA0(6,8),
1),AAS0(6,8)
C
      IF(SRFLAG) WRITE(6,10000)
10000 FORMAT(44H AN ENTRY HAS BEEN MADE IN SUBROUTINE STA01 )
      K=KN
      SCRIT=0.0
      I=IP
      ID=-1

```

Listing of Code (continued)

WGT1(K)=0.0	ST01 043
JW=1	ST01 044
IF(GAMF)2,2,3	ST01 045
2 TA1(K)=.95*TT0(IP,K)	ST01 046
CALL GAMMA(P TIN,TA1(K),FAIR,WAIR,GAM(2,K))	ST01 047
3 CALL FLOW1(I)	ST01 048
IF(PREVER) GO TO 26	ST01 049
WGT1(K)=WGT1(K)+WG1(I,K)	ST01 050
C TEST FOR TIP SECTION	ST01 051
IF(ISECT-1)5,5,4	ST01 052
4 I=I+ID	ST01 053
IF(I)6,6,22	ST01 054
22 L=I-ID	ST01 055
PS1(I,K)=PS1(L,K)+FLOAT(ID)*DPDR1(L,K)*(ST01 056
1H1(I,K)+H1(L,K))/2.	ST01 057
PTOPSI(I,K)=PT0(I,K)/PS1(I,K)	ST01 058
IF(PTOPSI(I,K)-1.)27,3,3	ST01 059
27 PTRN=-1.	ST01 060
PTOPSI(I,K)=1.0	ST01 061
GO TO 3	ST01 062
6 ID=1	ST01 063
I=IP+ID	ST01 064
GO TO 22	ST01 065
C CALCULATE STA 0 FOR INCIDENCE CURRECTION	ST01 066
5 IF(JW-1)16,16,18	ST01 067
16 IF(GAMF)7,7,17	ST01 068
7 GAM(1,K)=GAM(2,K)	ST01 069
17 EX=(GAM(1,K)-1.)/GAM(1,K)	ST01 070
EXI=1./EX	ST01 071
WGT0(K)=WGT1(K)/HWG(2,K)	ST01 072
I=IP	ST01 073
WG0(I,K)=WG1(I,K)/RWG(2,K)	ST01 074
FFA0(I,K)=WG0(I,K)*SQRT(TT0(I,K))/(144.*PT0(I,K)*	ST01 075
1ANN0(I,K))	ST01 076
19 J=1	ST01 077
8 CALL PRATIO(FFA0(I,K),GAM(1,K),RV(1,K),PTOPSO(I,K),PRTOL)	*****
PS0(I,K)=PTP(I,K)/PTOPSO(I,K)	ST01 079
TT0TS0(I,K)=PTOPSO(I,K)**EX	ST01 080
TS0(I,K)=TT0(I,K)/TT0TS0(I,K)	ST01 081
9 IF(GAMF)10,10,12	ST01 082
10 TAO(K)=.5*(TT0(I,K)+TS0(I,K))	ST01 083
CALL GAMMA(P TIN,TAO(K),FAIR,WAIR,GAM(1,K))	ST01 084
EX=(GAM(1,K)-1.)/GAM(1,K)	ST01 085
EXI=1./EX	ST01 086
IF(J-1)11,11,12	ST01 087
11 J=J+1	ST01 088
GO TO 4	ST01 089

Listing of Code (continued)

```

12 CP0(K)=RV(1,K)*EXI/AJ
DO 14 I=1,ISECT
    WG0(I,K)=WG1(I,K)/RWG(2,K)
    PT0MO= PT0(I,K)
    FFA0(I,K)=WG0(I,K)*SQRT( TT0(I,K)/(144.*PT0(I,K)*
1ANN0(I,K))
    IF(I.EQ.IP) GO TO 28
    PS0(I,K) = PS0(IP,K)
    PT0PS0(I,K) = PTP(I,K)/ PS0(I,K)
28 TT0TS0(I,K)=PT0PS0(I,K)**EX
    TS0(I,K)=TT0(I,K)/TT0IS0(I,K)
13 V0(I,K)=SQRT(2.*G*AJ*CP0(K)*(TT0(I,K)-TS0(I,K)))
    AAS0(I,K)=SQRT(GAM(1,K)*G*RV(1,K)*TS0(I,K))
    M0(I,K)=V0(I,K)/AAS0(I,K)
    SI(I,K)=ALPHA0(I,K)- HADSU(I,K)
    IF(SI(I,K))24,24,20
24 FXPS=EXPX
    GO TO 21
20 FXPS=EXPP
21 PT0PS0(I,K)=(1.+EX*M0(I,K)*ETARS(I,K)*GAM(1,K)*M0(I,K)/2.
1*(CCS(SI(I,K))*EXPS))*EXI
    PT0(I,K)=PS0(I,K)*PT0PS0(I,K)
    WG0(I,K)=WG0(I,K)*PT0(I,K)/PT0MO
    WG1(I,K)=WG1(I,K)*PT0(I,K)/PT0MO
    RHOS0(I,K)=144.*PS0(I,K)/(RV(1,K)*TS0(I,K))
    VU0(I,K)=V0(I,K)*SIN(ALPHA0(I,K))
    VZ0(I,K)=V0(I,K)*COS(ALPHA0(I,K))
14 CONTINUE
C    END OF INCIDENCE LOSS CORRECTION LOOP
    WGT1(K)=0.
    I=IP
    ID=-1
    JW=2
15 GO TO 3
18 CONTINUE
    WGT0(K)=WGT1(K)/RWG(2,K)
    IF(THLOOP.EQ.0.) GO TO 23
    WRITE(6,1000) WGT0(K),WG1(K),(WG0(L,K),L=1,ISECT)
    WRITE(6,1001) (PT0PS0(L,K),L=1,ISECT)
    WRITE(6,1002) (WG1(L,K),L=1,ISECT)
    WRITE(6,1003) (PT0PS1(L,K),L=1,ISECT)
1000 FORMAT(2X,6H WGT0=F8.3,2X,6H WGT1=F8.3/2X,6H WG0=6F8.3)
1001 FORMAT(1X,7HPT0PS0=6F8.5)
1002 FORMAT(2X,6H WG1=6F8.3)
1003 FORMAT(1X,7HPT0PS1=6F8.5)
23 CALL CHECK (J)
    GO TO (25,26),J

```

Listing of Code (continued)

25 CALL DIAGT(1)	ST01 137
26 IF(SRFLAG) WRITE(6,20000)	*****
20000 FORMAT(45H AN EXIT HAS BEEN MADE FROM SUBROUTINE STA01)	*****
RETURN	*****
END	ST01 139

Listing of Code (continued)

```

      SUBROUTINE FLOW1(I)
CFLOW1
C      ESTABLISH VALUES FOR STATOR EXIT FLOW
C
      REAL MFSTOP
      LOGICAL PREVER,SRFLAG
      COMMON SRFLAG
      COMMON /SNTCP/G,AJ,PRFC,ICASE,PREVER,MFSTOP,JUMP,LOPIN,ISCASE,
1KN,GAMF,IP,SCRIT,PTRN,ISECT,KSTG,WTOI,RHOTOL,PRTOL,TRLOOP,LSTG,
2LBRC,IRRC,ICHOKE,ISORH,CHOKE,PTOPSI(6,8),PTRS2(6,8),TRDIAG,SC,RC,
3DELPR,PASS,IPC,LOPC,ISS
C
      COMMON /SINIT/H1(6,8),H2(6,8),DP0(6,8),DP1(6,8),DP1A(6,8),DP2(6,8),
1,DP2A(6,8),CSALF1(6,8),ALF1(6,8),CSHET2(6,8),BET2(6,8),RADSD(6,8),
2RADRD(6,8),ANN1(6,8),ANN2(6,8),ANN2A(6,8),ANN1A(6,8),U1A(6,8),
3U2(6,8),ANN0(6,8),PT0(6,8),TT0(6,8),ALPHA0(6,8),PTP(6,8)
C
      COMMON /SINPUT/ HSL,TSL,PSL,GAMSL,
1PTPS,PTIN,TTIN,WAIR,FAIR,DELC,DELL,DELA,AACS,VCTD,STG,SECT,EXPN,
2EXPP,EXPHE, RPM,PAF,SLI,STGCH,FNDJUH,NAME(10),TITLE(10),PCNH(6),
3RV(6,8),GAM(6,8),DR(6,8),DT(6,8),HWG(6,8),ALPHAS(6,8),ALPHA1(6,8),
4ETARS(6,8),ETAS(6,8),CFS(6,8),ANN0(6,8),BETA1(6,8),BETA2(6,8),ETARFLW1
5R(6,8),ETAR(6,8),CFR(6,8),TFR(6,8),ANDQH(6,8),OMEGAS(6,8),AS0(6,8),
6,ASMP0(6,8),ACMN0(6,8),A1(6,8),A2(6,8),A3(6,8),A4(6,8),A5(6,8),A6(
7,8),OMEGAR(6,8),BSIA(6,8),RSMPIA(6,8),HCMNIA(6,8),B1(6,8),B2(6,8),
8,B3(6,8),B4(6,8),B5(6,8),B6(6,8),SESTHI(8),RERTHI(8)
C
      REAL M0
      COMMON /SSTA01/CP0(8),
18),VU0(6,8),VZ0(6,8),RHOS0(6,8),PS1(6,8),WGT1(8),TA1(8),WG1(6,8),
2DPDH1(6,8),SI(6,8), CP1(8),PHI1(6,8),TS1(6,8),V1(6,8),
3,RHOS1(6,8),ALF1E(6,8),VU1(6,8),VZ1(6,8),M0(6,8),WGT0(8),WG0(6,8)
C
      DIMENSION PHI1C(8),PTPS1C(8),VIC(6,8),TS1C(6,8),RHOS1C(6,8),WG1C(6
1,8),CSAL1E(6,8),SFF(6,8)
C
C
      IF(SRFLAG) WRITE(6,10000)
10000 FORMAT(44H AN ENTRY HAS BEEN MADE IN SUBROUTINE FLOW1 )
      K=KN
      EX=(GAM(2,K)-1.)/GAM(2,K)
C      COMPUTE ISENTROPIC STATOR TEMPERATURE RATIO
7 PHI1(I,K)=PTOPSI(I,K)**EX
C      TEST FOR LOSS COEFFICIENT INPUT
      IF (OMEGAS(1,1))2,2,1
1 CALL LOSS1(I,K,EX)

```

Listing of Code (continued)

2	TS1(I,K)=TT0(I,K)*(1.-ETAS(I,K)*(1.-1./PHI1(I,K)))	FLW1 043
	IF(I-IP)6,3,6	FLW1 044
3	IF(GAMF)4,4,5	FLW1 045
4	TA1(K)=.5*(TT0(I,K)+TS1(I,K))	FLW1 046
	CALL GAMMA(PT0(IP,K),TA1(K),FAIR,WAIR,GAM(2,K))	FLW1 047
5	EX=(GAM(2,K)-1.0)/GAM(2,K)	FLW1 048
	FXI=1./EX	FLW1 049
C	CRITICAL PRESSURE RATIO	FLW1 050
	CALL PHIM(EXI,ETAS(I,K),PHI1C(K),PTPS1C(K))	FLW1 051
	CP1(K)=RV(2,K)*EXI/AJ	*****
C	EXIT VELOCITY	FLW1 053
6	V1(I,K)=SQRT(2.*G*AJ*CP1(K)*(TT0(I,K)-TS1(I,K)))	FLW1 054
C	EXIT PRESSURE	FLW1 055
	PS1(I,K)=PT0(I,K)/PT0PS1(I,K)	FLW1 056
C	EXIT DENSITY	FLW1 057
	RHOS1(I,K)=144.*PS1(I,K)/(RV(2,K)*TS1(I,K))	*****
C	TEST CRITICAL PRESSURE RATIO	FLW1 059
	IF(PT0PS1(I,K)-PTPS1C(K))15, 8,8	FLW1 060
C	GREATER THAN CRITICAL	FLW1 061
8	IF (IP-1) 21,9,21	FLW1 062
9	IF (PRPC)10,10,22	FLW1 063
C	PREVIOUS PITCH NONCRITICAL	FLW1 064
10	PRPC=1.	FLW1 065
	PT0PS1(I,K)=PTPS1C(K)*(1.+PRTOL)	FLW1 066
	GO TO 7	FLW1 067
21	IF (PT0PS1(I,K).LE.PT0PS1(IP,K)) GO TO 22	FLW1 068
	GO TO 12	FLW1 069
22	IF ((I.EQ.1).OR.(I.EQ.ISECT)) SCRIT=1.	FLW1 070
	GO TO 11	FLW1 071
C	PITCH OR OUTBOARD SECTION	FLW1 072
11	CONTINUE	FLW1 073
	V1C(I,K)=SQRT(2.*G*AJ*CP1(K)*TT0(I,K)*ETAS(I,K)*(PHI1C(K)	FLW1 074
	1-1.)/PHI1C(K))	FLW1 075
	TS1C(I,K)=TT0(I,K)*(1.-ETAS(I,K)*(1.-1./PHI1C(K)))	FLW1 076
	RHOS1C(I,K)=144.*PT0(I,K)/(PTPS1C(K)*TS1C(I,K)*RV(2,K))	*****
	WG1C(I,K)=RHOS1C(I,K)*V1C(I,K)*ANN1(I,K)*CSALF1(I,K)	FLW1 078
	WG1(I,K)=WG1C(I,K)	FLW1 079
13	CSAL1E(I,K)=WG1(I,K)/(RHOS1(I,K)*V1(I,K)*ANN1(I,K))	FLW1 080
C	EFFECTIVE STATION EXIT ANGLE	FLW1 081
14	ALF1E(I,K)=ATAN2(SQRT(1.-CSAL1E(I,K)*CSAL1E(I,K)),	FLW1 082
	1CSAL1E(I,K))	FLW1 083
	GO TO 16	FLW1 084
12	IF (PRPC-1.)15,15,24	FLW1 085
24	WG1(I,K)=SFF(I,K)*PT0(I,K)/SQRT(TT0(I,K))	FLW1 086
	GO TO 13	FLW1 087
C	PRESSURE RATIO LESS THAN CRITICAL OR SUPERSONIC FLOW DECREASE	FLW1 088
15	WG1(I,K)=RHOS1(I,K)*V1(I,K)*ANN1(I,K)*CSALF1(I,K)	FLW1 089

Listing of Code (continued)

CSALIE(I,K)=CSALF1(I,K)	FLW1 090
ALF1E(I,K)=ALF1(I,K)	FLW1 091
SFF(I,K)=WG1(I,K)*SQRT(TT0(I,K))/PT0(I,K)	FLW1 092
16 VU1(I,K)=V1(I,K)*SIN(ALF1E(I,K))	FLW1 093
DPDR1(I,K)=.01388889*RHOS1(I,K)*VU1(I,K)*VU1(I,K)/	*****
1(G*DP1(I,K))	FLW1 095
VZ1(I,K)=V1(I,K)*CSALIE(I,K)	FLW1 096
IF(I.LT.ISECT) GO TO 17	FLW1 097
IF(PRPC.EQ.1.) PRPC=2.	FLW1 098
17 CALL CHECK(J)	FLW1 099
GO TO (19,20),J	FLW1 100
19 CALL DIAGT(2)	FLW1 101
20 IF(SRFLAG) WRITE(6,20000)	*****
20000 FORMAT(45H AN EXIT HAS BEEN MADE FROM SUBROUTINE FLOW1)	*****
RETURN	*****
END	FLW1 103

Listing of Code (continued)

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SUBROUTINE LOSS1(I,K,EX)                                LOS1 001
CLOSS1                                                    LOS1 002
C                                                         LOS1 003
C   CALCULATE EFFICIENCY                                LOS1 004
C                                                         LOS1 005
REAL MFSTOP                                              LOS1 006
LOGICAL PREVER,SRFLAG                                   *****
COMMON SRFLAG                                           *****
COMMON /SNTCP/G,AJ,PRFC,ICASE,PREVER,MFSTOP,JUMP,LOPIN,ISCASE, LOS1 008
1KN,GAMF,IP,SCHIT,PTRN,ISECT,KSTG,WOTOL,RHOTOL,PTOL,TRLOOP,LSTG, LOS1 009
2LBRC,IBRC,ICHOKE,ISORH,CHOKE,PTOPSI(6,8),PTRS2(6,8),TRDIAG,SC,RC, LOS1 010
3DELPR,PASS,IPC,LOPC,ISS                                LOS1 011
C                                                         LOS1 012
COMMON /SINIT/H1(6,8),H2(6,8),DP0(6,8),DP1(6,8),DP1A(6,8),DP2(6,8) LOS1 013
1,DP2A(6,8),CSALF1(6,8),ALF1(6,8),CSHET2(6,8),BET2(6,8),RADSD(6,8), LOS1 014
2RADRD(6,8),ANN1(6,8),ANN2(6,8),ANN2A(6,8),ANN1A(6,8),U1A(6,8), LOS1 015
3U2(6,8),ANN0(6,8),PT0(6,8),TT0(6,8),ALPHA0(6,8),PTP(6,8)      LOS1 016
C                                                         LOS1 017
COMMON /SINPUT/ RSL,TSL,PSL,GAMSI,                     *****
1PTPS,PTIN,TTIN,WAIR,FAIR,DELC,DELL,DELA,AACS,VCTD,STG,SECT,EXPN, LOS1 019
2EXPP,EXPRE, RPM,PAF,SLI,STGCH,FNDJOB,NAME(10),TITLE(10),PCNH(6), *****
3RV(6,8),GAM(6,8),DR(6,8),DT(6,8),RWG(6,8),ALPHAS(6,8),ALPHA1(6,8), *****
4ETAHS(6,8),ETAS(6,8),CFS(6,8),ANN0(6,8),BETA1(6,8),BETA2(6,8),ETARLOS1 022
5R(6,8),ETAR(6,8),CFR(6,8),TFR(6,8),ANDOR(6,8),OMEGAS(6,8),ASO(6,8) LOS1 023
6,ASMP0(6,8),ACMNO(6,8),A1(6,8),A2(6,8),A3(6,8),A4(6,8),A5(6,8),A6(6,8) LOS1 024
7,OMEGAR(6,8),BSIA(6,8),RSMPIA(6,8),HCMNIA(6,8),R1(6,8),B2(6,8) LOS1 025
8,B3(6,8),H4(6,8),B5(6,8),B6(6,8),SESTHI(8),RETHI(8)          LOS1 026
C                                                         LOS1 027
REAL M0                                                  LOS1 028
COMMON /SSTA01/CP0(8),                                PS0(6,8),V0(6,8),TS0(6,8) LOS1 029
18),VU0(6,8),VZ0(6,8),RHOS0(6,8),PS1(6,8),WGT1(8),TA1(8),WG1(6,8), LOS1 030
2DPDM1(6,8),SI(6,8), CP1(8),PHI1(6,8),TS1(6,8),V1(6,8) LOS1 031
3,RHOS1(6,8),ALF1E(6,8),VU1(6,8),VZ1(6,8),M0(6,8),WGT0(8),WG0(6,8) *****
C                                                         LOS1 033
C                                                         LOS1 034
IF(SRFLAG) WRITE(6,10000)                               *****
10000 FORMAT(44H AN ENTRY HAS BEEN MADE IN SUBROUTINE LOSS1 ) *****
EXPX=0.0                                                  LOS1 035
EXPP=0.0                                                  LOS1 036
ETAHS(I,K)=1.0                                           LOS1 037
SI(I,K)=ALPHA0(I,K)-RADSD(I,K)                           LOS1 038
IF(SI(I,K))5,1,2                                         LOS1 039
1 W01=OMEGAS(I,K)                                        LOS1 040
GO TO 9                                                  LOS1 041
2 AS=A1(I,K)                                             LOS1 042
AC=A2(I,K)                                              LOS1 043

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Listing of Code (continued)

AQ=A3(I,K)	LOS1 044
IF(ASMP0(I,K)-SI(I,K))3.4.4	LOS1 045
3 WMWS=SI(I,K)/ASMP0(I,K)	LOS1 046
AR=ASMP0(I,K)/ASO(I,K)	LOS1 047
GO TO 8	LOS1 048
4 WMWS=1.0	LOS1 049
AR=SI(I,K)/ASO(I,K)	LOS1 050
GO TO 8	LOS1 051
5 AS=A4(I,K)	LOS1 052
AC=A5(I,K)	LOS1 053
AQ=A6(I,K)	LOS1 054
IF(SI(I,K)-ACMNO(I,K))6.4.4	LOS1 055
6 WMWS=SI(I,K)/ACMNO(I,K)	LOS1 056
AR=ACMNO(I,K)/ASO(I,K)	LOS1 057
8 W01=(1.+AR*AR*(AS+AR*(AC+AR*AQ)))*WMWS*OMEGAS(I,K)	LOS1 058
9 ETAS(I,K)=(1.-(1./(P70PS1(I,K)*(1.-W01)+W01))*EX)*PHI1(I,K)/	LOS1 059
1(PHI1(I,K)-1.)	LOS1 060
CALL CHECK(J)	LOS1 061
IF(SRFLAG) WRITE(6,20000)	*****
20000 FORMAT(45H AN EXIT HAS BEEN MADE FROM SUBROUTINE LOSS1)	*****
RETURN	LOS1 062
END	LOS1 063

Listing of Code (continued)

	SUBROUTINE R(P,T,F,W,RX)	R	001
CR		R	002
C	CALCULATE GAS CONSTANT	R	003
	WRITE (6,100)		*****
100	FORMAT (//120H SUBROUTINE R HAS BEEN CALLED UPON *****		
	1*****		
	2**,//)		*****
	RX=53.35045+(.658*F+32.433*W)/(1.+F+W)	R	004
	RETURN	R	005
	END	R	006

Listing of Code (continued)

SUBROUTINE GAMMA(P,T,F,w,GAMX)	GAMA 001
CGAMMA	GAMA 002
C	GAMA 003
C CALCULATE SPECIFIC HEAT RATIO FOR MIXTURE	GAMA 004
WRITE (6,100)	*****
100 FORMAT (//120H SUBROUTINE GAMMA HAS BEEN CALLED UPON *****	*****
1 *****	*****
2 **,//)	*****
CALL CPA(P,T,F,w,CPAX)	GAMA 005
IF(F)2,2,1	GAMA 006
1 CALL CPF(P,T,F,w,CPFX)	GAMA 007
2 IF(w)4,4,3	GAMA 008
3 CALL CPW(P,T,F,w,CPWX)	GAMA 009
4 CPGX=(CPAX+F*CPFX+W*CPWX)/(1.+F+w)	GAMA 010
CALL R(P,T,F,w,RX)	GAMA 011
GAMX=CPGX/(CPGX-RX/778.161)	GAMA 012
RETURN	GAMA 013
END	GAMA 014

Listing of Code (continued)

	SUBROUTINE CPA(P,T,F,W,CPAX)	CPA 001
CCPA		CPA 002
C	CALCULATE SPECIFIC HEAT RATIO FOR AIR	CPA 003
	DIMENSION	CPA 004
	1XT(7),A(7)	CPA 005
	WRITE (6,100)	*****
100	FORMAT (//120H SUBROUTINE CPA HAS BEEN CALLED UPON *****	*****
	1*****	*****
	2**,//)	CPA 006
	IF(T-100.)1,2,2	CPA 007
1	TX=100.	CPA 008
	GO TO 5	CPA 009
2	IF(6400.-T)3,4,4	CPA 010
3	TX=6400.	CPA 011
	GO TO 5	CPA 012
4	TX=T	CPA 013
5	XT(1)=TX/1000.	CPA 014
	DO 6 I=2,7	CPA 015
6	XT(I)=XT(I-1)*XT(1)	CPA 016
	CPAX=2.4264907E-01-2.6657395E-02*XT(1)+4.6617756E-02*XT(2)	CPA 017
	1-1.3546542E-02*XT(3)-8.4500931E-04*XT(4)+1.0303393E-03*	CPA 018
	2XT(5)-1.7159795E-04*XT(6)+9.1627911E-06*XT(7)	CPA 019
	RETURN	CPA 020
	END	

Listing of Code (continued)

SUBROUTINE CPF(P,T,F,*,CPFX)	CPF	001
CCPF	CPF	002
C CALCULATE SPECIFIC HEAT RATIO FOR FUEL	CPF	003
DIMENSION	CPF	004
1 XT(7),A(7)	CPF	005
WRITE (6,100)	*****	
100 FORMAT (//120H SUBROUTINE CPF HAS BEEN CALLED UPON *****	*****	
1 *****	*****	
2 **,//)	*****	
IF(T-400.)1,2,2	CPF	006
1 TX=400.	CPF	007
GO TO 5	CPF	008
2 IF(3000.-T)3,4,4	CPF	009
3 TX=3000.	CPF	010
GO TO 5	CPF	011
4 TX=T	CPF	012
5 XT(1)=TX/1000.	CPF	013
DO 6 I=2,7	CPF	014
6 XT(I)=XT(I-1)*XT(1)	CPF	015
CPFX=1.0625243E-01+9.5291284E-01*XT(1)-7.2605169E-01*XT(2)	CPF	016
1+2.4481406E-01*XT(3)+5.3332162E-02*XT(4)-6.4699814E-02*XT(5)	CPF	017
2+1.7495567E-02*XT(6)-1.6029820E-03*XT(7)	CPF	018
RETURN	CPF	019
END	CPF	020

Listing of Code (continued)

CCPW	SUBROUTINE CPW(P,T,F,*,CPWX)	CPW	001
C	CALCULATE SPECIFIC HEAT FOR WATER VAPOR	CPW	002
	DIMENSION	CPW	003
	1XT(7),A(7)	CPW	004
	WRITE (6,100)	CPW	005

100	FORMAT (//120H SUBROUTINE CPW HAS BEEN CALLED UPON *****		
	1*****		
	2**,//)		*****
	IF(T-400.)1,2,2	CPW	006
1	TX=400.	CPW	007
	GO TO 5	CPW	008
2	IF(3000.-T)3,4,4	CPW	009
3	TX=3000.	CPW	010
	GO TO 5	CPW	011
4	TX=T	CPW	012
5	XT(1)=TX/1000.	CPW	013
	DO 6 I=2,7	CPW	014
6	XT(I)=XT(I-1)*XT(1)	CPW	015
	CPWX=4.5728850E-01+9.7007556E-02*XT(1)+1.6536409E-01	CPW	016
	1*XT(2)-4.1138066E-02*XT(3)-2.6979575E-02*XT(4)+2.2619243E-02	CPW	017
	2*XT(5)-6.2706207E-03*XT(6)+6.2246710E-04*XT(7)	CPW	018
	RETURN	CPW	019
	END	CPW	020

Listing of Code (continued)

SUBROUTINE PRATIO(TFF,GAMX,RX,PTPS,PRTOL)	PRIO 001
CPRATIO	PRIO 002
C CALCULATE PRESSURE RATIO	PRIO 003
LOGICAL PREVER,SRFLAG	*****
COMMON SRFLAG	*****
IF(SRFLAG) WRITE(6,10000)	*****
10000 FORMAT(44H AN ENTRY HAS BEEN MADE IN SUBROUTINE PRATIO)	*****
A=GAMX/(GAMX-1.)	PRIO 004
R=2./GAMX	PRIO 005
C=(GAMX+1.)/GAMX	PRIO 006
D=TFF*SQRT(RX/(64.3481*A))	PRIO 007
PCHIT=((GAMX+1.)/2.)*A	PRIO 008
PUP=PCHIT	PRIO 009
PLOW=1.0	PRIO 010
PTRMO=0.0	PRIO 011
1 PTR=(PUP+PLOW)/2.	PRIO 012
DELFM=SQRT(1./(PTR**R)-1./(PTR**C))-D	PRIO 013
IF(DELFM)2,3,3	PRIO 014
2 PLOW=PTR	PRIO 015
GO TO 4	PRIO 016
3 PUP=PTR	PRIO 017
4 PRE=(PTR-PTRMO)/PTR	PRIO 018
IF (ABS(PRE)-PRTOL)6,6,5	PRIO 019
5 PTRMO=PTR	PRIO 020
GO TO 1	PRIO 021
6 IF(PCHIT-PTR)7,8,8	PRIO 022
7 PTPS=PCHIT	PRIO 023
GO TO 9	PRIO 024
8 PTPS=PTR	PRIO 025
9 CONTINUE	PRIO 026
IF(SRFLAG) WRITE(6,20000)	*****
20000 FORMAT(45H AN EXIT HAS BEEN MADE FROM SUBROUTINE PRATIO)	*****
RETURN	PRIO 027
END	PRIO 028

Listing of Code (continued)

SUBROUTINE CHECK(J)	CHCK 001
CCHECK	CHCK 002
C SUBROUTINE TO CHECK SENSE LIGHTS	CHCK 003
C	CHCK 004
REAL MFSTOP	CHCK 005
LOGICAL PREVEN,SRFLAG	*****
COMMON SRFLAG	*****
COMMON /SNTCP/G,AJ,PRPC,ICASE,PREVEN,MFSTOP,JUMP,LOPIN,ISCASE,	CHCK 007
1KN,GAMF,IP,SCHIT,PTNK,ISECT,KSTG,WIUL,HHOTOL,PRTOL,TRLOOP,LSTG,	CHCK 008
2LHRC,IHHC,ICHOK,ISORH,CHOKE,PTOPS1(6,8),PTRS2(6,8),TRDIAG,SC,RC,	CHCK 009
3DELPH,PASS,IPC,LOPC,ISS	CHCK 010
C	CHCK 011
IF(SRFLAG) WRITE(6,10000)	*****
10000 FORMAT(44H AN ENTRY HAS BEEN MADE IN SUBROUTINE CHECK)	*****
DO 1 I=1,4	CHCK 012
CALL SLITET(I,J)	CHCK 013
GO TO (2,1),J	CHCK 014
1 CONTINUE	CHCK 015
J=2	CHCK 016
IF(SRFLAG) WRITE(6,20000)	*****
RETURN	CHCK 017
2 J=1	CHCK 018
PREVEN=.TRUE.	CHCK 019
IF(SRFLAG) WRITE(6,20000)	*****
20000 FORMAT(45H AN EXIT HAS BEEN MADE FROM SUBROUTINE CHECK)	*****
RETURN	CHCK 020
END	CHCK 021

Listing of Code (continued)

```

      SUBROUTINE STA1A
CSTA1A
C
      REAL MFSTOP
      LOGICAL PREVER,SRFLAG
      COMMON SRFLAG
      COMMON /SNTCP/G,AJ,PRFC,ICASE,PREVER,MFSTOP,JUMP,LOPIN,ISCASE,
      1KN,GAMF,IP,SCRIT,PTRN,ISECT,KSTG,WOTOL,RHOTOL,PTOL,TRLOOP,LSTG,
      2LHRC,IHRC,ICHOKE,ISORH,CHOKE,PTOP1(6,8),PTRS2(6,8),TRDIAG,SC,RC,
      3DELPR,PASS,IPC,LOPC,ISS
C
      COMMON /SINIT/H1(6,8),H2(6,8),DP0(6,8),DP1(6,8),DP1A(6,8),DP2(6,8)
      1,DP2A(6,8),CSALF1(6,8),ALF1(6,8),CSHET2(6,8),BET2(6,8),RADSD(6,8),
      2RADRD(6,8),ANN1(6,8),ANN2(6,8),ANN2A(6,8),ANN1A(6,8),U1A(6,8),
      3U2(6,8),ANN0(6,8),PT0(6,8),TT0(6,8),ALPHA0(6,8),PTP(6,8)
C
      COMMON /SINPUT/ RSL,TSL,PSL,GAMSI,
      1PTPS,PTIN,TTIN,WAIR,FAIR,DELC,DELL,DELA,AACS,VCTD,STG,SECT,EXPN,
      2EXPP,EXPRE, RPM,PAF,SLI,STGCH,FNDJOH,NAME(10),TITLE(10),PCNH(6),
      3RV(6,8),GAM(6,8),UR(6,8),UT(6,8),RWG(6,8),ALPHAS(6,8),ALPHA1(6,8),
      4ETARS(6,8),ETAS(6,8),CFS(6,8),ANN0(6,8),BETA1(6,8),BETA2(6,8),ETARST1A
      5R(6,8),ETAR(6,8),CFR(6,8),TFR(6,8),ANDOR(6,8),OMEGAS(6,8),AS0(6,8)
      6,ASMP0(6,8),ACMNO(6,8),A1(6,8),A2(6,8),A3(6,8),A4(6,8),A5(6,8),A6(
      7,8),OMEGAR(6,8),BSIA(6,8),BSMP1A(6,8),HCMN1A(6,8),B1(6,8),B2(6,8)
      8,B3(6,8),B4(6,8),B5(6,8),B6(6,8),SESTHI(8),RERTHI(8)
C
      REAL M0
      COMMON /SSTA01/CP0(8),
      1R),VU0(6,8),VZ0(6,8),RHOS0(6,8),PS1(6,8),WGT1(8),TA1(8),WG1(6,8),
      2DPDR1(6,8),SI(6,8), CP1(8),PHI1(6,8),TS1(6,8),V1(6,8)
      3,RHOS1(6,8),ALF1E(6,8),VU1(6,8),VZ1(6,8),M0(6,8),WGT0(8),WG0(6,8)
      REAL MR1A
      COMMON /SSTA1A/VU1A(6,8),WG1A(6,8),WGT1A(8),VZ1A(6,8), CP1A(8),
      1PS1A(6,8),RU1A(6,8),RI1A(6,8),HET1A(6,8),RI(6,8),TTR1A(6,8),PTR1A(6
      2,8),MR1A(6,8),TS1A(6,8)
C
      DETERMINE FLOW CONDITIONS RELATIVE TO ROTOR, FIND INCIDENCE
C
      ANGLE RECOVERY ROTOR INLET STATIONS, OBTAIN GAS PROPERTIES,
C
      ABSOLUTE TANGENTIAL COMPONENT VELOCITY ADJUSTED FOR DIAMETER
C
      CHANGE TO CONSERVE ANGULAR MOMENTUM, AXIAL COMPONENT
C
      VELOCITY ADJUSTED FOR WEIGHT FLOW, AREA, AND DENSITY CHANGE
C
      FROM STA 1.
C
      IF(SRFLAG) WRITE(6,10000)
      10000 FORMAT(44H AN ENTRY HAS BEEN MADE IN SUBROUTINE STA1A )
      K=KN

```

Listing of Code (continued)

I=IP	ST1A 043
ID=-1	ST1A 044
TS1A(I,K)=TS1(I,K)	*****
C RATIO OF FLOW CHANGE	ST1A 046
WR=RWG(3,K)/RWG(2,K)	ST1A 047
C TOTAL STATION FLOW	ST1A 048
WGT1A(K)=WR*WGT1(K)	ST1A 049
C ADJUST TANGENTIAL VELOCITY	ST1A 050
13 VU1A(I,K)=VU1(I,K)*DPl(I,K)/DPlA(I,K)	ST1A 051
C ADJUST FLOW	ST1A 052
WG1A(I,K)=WR*WG1(I,K)	ST1A 053
RHOSTR=RHOS1(I,K)	ST1A 054
C ADJUST AXIAL VELOCITY	ST1A 055
1 VZ1A(I,K)=WR*VZ1(I,K)*ANN1(I,K)*RHOS1(I,K)/(ANN1A(I,K)	ST1A 056
1*RHOSTR)	ST1A 057
V1A =SQRT(VU1A(I,K)*VU1A(I,K)+VZ1A(I,K)*VZ1A(I,K))	ST1A 058
IF(I-IP)2,3,2	ST1A 059
2 EX=(GAM(3,K)-1.)/GAM(3,K)	ST1A 060
EXI=1./EX	ST1A 061
GO TO 4	ST1A 062
3 IF(GAMF)12,12,2	ST1A 063
12 TAlA =.5*(Tf0(I,K)+TS1A(I,K))	*****
CALL GAMMA(PT0(I,K),TAlA ,FAIR,wAIR,GAM(3,K))	ST1A 065
EX=(GAM(3,K)-1.)/GAM(3,K)	ST1A 066
EXI=1./EX	ST1A 067
4 CPlA(K)=RV(3,K)*EXI/AJ	*****
DELTS=(V1(I,K)*V1(I,K)-V1A *V1A)/(2.*G*AJ*CPlA(K))	ST1A 069
TS1A(I,K)=TS1(I,K)+DELTS	*****
PS1A(I,K)=PS1(I,K)*(1.+DELTS/TS1(I,K))*EXI	ST1A 071
RHOS1A =144.*PS1A(I,K)/(RV(3,K)*TS1A(I,K))	*****
C DENSITY ERROR	ST1A 073
RHOE=(RHOS1A -RHOSTR)/RHOS1A	ST1A 074
IF (ABS(RHOE)-RHOTOL)6,6,5	ST1A 075
5 RHOSTR=RHOS1A	ST1A 076
GO TO 1	ST1A 077
6 RU1A(I,K)=VU1A(I,K)-U1A(I,K)	ST1A 078
R1A(I,K)=SQRT(RU1A(I,K)*RU1A(I,K)+VZ1A(I,K)*VZ1A(I,K))	ST1A 079
SBET1A =RU1A(I,K)/R1A(I,K)	ST1A 080
BET1A(I,K)=ATAN2(SBET1A ,SQRT(1.-SBET1A *SBET1A))	ST1A 081
IF(OMEGAR(I,K))8,8,7	ST1A 082
7 ETARR(I,K)=1.	ST1A 083
EXPRE=0.0	ST1A 084
8 MR1A(I,K)=R1A(I,K)/SQRT(GAM(3,K)*G*RV(3,K)*TS1A(I,K))	*****
TRTS1A =1.+(GAM(3,K)-1.)*MR1A(I,K)*MR1A(I,K)/2.	ST1A 086
IF(TRTS1A.GT.1.) GO TO 32	ST1A 087
PREVER = .TRUE.	ST1A 088
GO TO 17	ST1A 089

Listing of Code (continued)

32	TTR1A(I,K)=TS1A(I,K)*IRTS1A	*****
	RI(I,K)=BET1A(I,K)-RACRD(I,K)	ST1A 091
	IF(RI(I,K).GT.1.570796) RI(I,K)=1.570796	*****
	IF(RI(I,K).LT.-1.570796) RI(I,K)=-1.570796	*****
	IF(RI(I,K))9,9,10	ST1A 094
9	EXPH=EXPN	ST1A 095
	GO TO 11	ST1A 096
10	EXPH=EXPP	ST1A 097
11	PRPS1A=(1.+(TRTS1A-1.)*ETARR(I,K)*(COS(RI(I,K)))**	ST1A 098
	1EXPH)**EXI	ST1A 099
	PTR1A(I,K)=PS1A(I,K)*PRPS1A	ST1A 100
	IF (ISECT-I)14,16,14	ST1A 101
14	I=I-ID	ST1A 102
	IF (I)15,15,13	ST1A 103
15	ID=1	ST1A 104
	I=IP-ID	ST1A 105
	GO TO 13	ST1A 106
16	CONTINUE	ST1A 107
	CALL CHECK(J)	ST1A 108
	GO TO (17,18),J	ST1A 109
17	CALL DIAGT(3)	ST1A 110
18	IF(SRFLAG) WRITE(6,20000)	*****
20000	FORMAT(45H AN EXIT HAS BEEN MADE FROM SUBROUTINE STA1A)	*****
	RETURN	*****
	END	ST1A 112

Listing of Code (continued)

```

SUBROUTINE STA2                                ST2  001
CSTA2                                          ST2  002
C    SATISFY CONTINUITY OF FLOW AT ROTOR EXIT  ST2  003
C                                              ST2  004
      REAL MFSTOP                               ST2  005
      LOGICAL PREVER,SRFLAG                     *****
      COMMON SRFLAG                           *****
      COMMON /SNTCP/G,AJ,PRFC,ICASE,PREVER,MFSTOP,JUMP,LOPIN,ISCASE, ST2  007
      1KN,GAMF,IP,SCHIT,PTRN,ISECT,KSTG,WIOL,HHOTOL,PRITOL,TRLOOP,LSTG, *****
      2LHRC,IHRC,ICHOKE,ISORH,CHUKE,PTOPSL(6,8),PTRSZ(6,8),TRDIAG,SC,RC, ST2  009
      3DELPR,PASS,IPC,LOPC,ISS                 ST2  010
C                                              ST2  011
      COMMON /SIN[T/H1(6,8),H2(6,8),DP0(6,8),DP1(6,8),DP1A(6,8),DP2(6,8) ST2  012
      1,DP2A(6,8),CSALF1(6,8),ALF1(6,8),CSHET2(6,8),HET2(6,8),RADSD(6,8), ST2  013
      2RADRU(6,8),ANN1(6,8),ANN2(6,8),ANN2A(6,8),ANN1A(6,8),U1A(6,8), ST2  014
      3U2(6,8),ANN0(6,8),PT0(6,8),TT0(6,8),ALPHA0(6,8),PTP(6,8)         ST2  015
C                                              ST2  016
      COMMON /SINPUT/ RSL,TSL,PSL,GAMSI, *****
      1PTPS,PTIN,TTIN,WAIR,FAIR,DELC,DEIL,DELA,AACS,VCTD,STG,SECT,EXPN, ST2  018
      2FXPP,EXPRE, RPM,PAF,SLI,STGCH,FNDJUH,NAME(10),TITLE(10),PCNH(6), *****
      3RV(6,8),GAM(6,8),OR(6,8),DT(6,8),RWG(6,8),ALPHAS(6,8),ALPHA1(6,8), *****
      4ETARS(6,8),ETAS(6,8),CFS(6,8),AND0(6,8),BETA1(6,8),BETA2(6,8),ETARST2  021
      5R(6,8),ETAR(6,8),CFR(6,8),TFR(6,8),ANDCH(6,8),OMEGAS(6,8),AS0(6,8) ST2  022
      6,ASMP0(6,8),ACMN0(6,8),A1(6,8),A2(6,8),A3(6,8),A4(6,8),A5(6,8),A6( ST2  023
      76,8),OMFGAR(6,8),BSIA(6,8),RSMPIA(6,8),HCMNIA(6,8),B1(6,8),B2(6,8) ST2  024
      8,B3(6,8),H4(6,8),H5(6,8),H6(6,8),SESTHI(8),HERTHI(8)             ST2  025
C                                              ST2  026
      REAL MR1A                                ST2  027
      COMMON /SSTA1A/VU1A(6,8),WG1A(6,8),WGT1A(8),VZ1A(6,8), CP1A(8), ST2  028
      1PS1A(6,8),RU1A(6,8),RIA(6,8),BET1A(6,8),RI(6,8),TTR1A(6,8),PTR1A(6 ST2  029
      2,8),MR1A(6,8),TS1A(6,8) *****
C                                              ST2  031
      COMMON /SSTA2/V2(6,8),TTR2(6,8),PTH2(6,8),WG2(6,8),WGT2(8),TA2(8), ST2  032
      1 PS2(6,8),PHI2(6,8)          ST2  033
C                                              ST2  034
      REAL MR2,M2, MF2                        ST2  035
      COMMON /SFLOW2/TS2(6,8),CP2(8),R2(6,8),RHOS2(6,8),BET2E(6,8),RU2(6 ST2  036
      1,8),VU2(6,8),UPDR2(6,8),VZ2(6,8),MR2(6,8),MF2(6,8),M2(6,8)       ST2  037
C                                              ST2  038
      DIMENSION WGT2C(8),FFA2(6,8),IS2(8)   ST2  039
C                                              ST2  040
C                                              ST2  041
C                                              *****
10000 IF(SRFLAG) WRITE(6,10000)              *****
      FORMAT(44H AN ENTRY HAS BEEN MADE IN SUBROUTINE STA2 ) *****
      K=KN                                     *****
      J=1                                     ST2  042

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Listing of Code (continued)

SCRIT=0.0	ST2	043
PTRMO=1.	ST2	044
IS2(K)=0	ST2	045
EXI=GAM(3,K)/(GAM(3,K)-1.)	ST2	046
WR=RWG(4,K)/RWG(3,K)	ST2	047
DO 1 I=1,ISECT	ST2	048
TTR2(I,K)=TTR1A(I,K)+(U2(I,K)**2 - U1A(I,K)**2)/(2.*G*AJ*CP1A(K))	ST2	049
PTR2(I,K)=PTR1A(I,K)*(TTR2(I,K)/TTR1A(I,K))*EXI	ST2	050
1 WG2(I,K)=WR*WG1A(I,K)	ST2	051
WGT2(K)=WR*WGT1A(K)	ST2	052
I=IP	ST2	053
ID=-1	ST2	054
WGT2C(K)=0.	ST2	055
IF(ICH0KE)26,26,3	ST2	056
26 IF(LOPIN)27,27,3	ST2	057
27 IF(GAMF)2,2,16	ST2	058
2 TA2(K)=.95*TTR2(IP,K)	ST2	059
CALL GAMMA(PTR2(I,K),TA2(K),FAIR,WAIR,GAM(4,K))	ST2	060
16 FFA2(I,K)=WG2(I,K)*SQRT(TTR2(I,K))/(144.*PTR2(I,K)*CSBET2(I,K)*	ST2	061
1ANN2(I,K))	ST2	062
CALL PRATIO(FFA2(I,K),GAM(4,K),RV(4,K),PTRS2(I,K),PRTOL)	*****	
3 CALL FLOW2(I)	ST2	064
IF (PHEVER) GO TO 22	ST2	065
WGT2C(K)=WGT2C(K)+WG2(I,K)	ST2	066
L=1	ST2	067
IF (PTRS2(I,K).LE.PTRS2(IP,K)) L=I	ST2	068
IF(ISECT-I)7,7,4	ST2	069
4 I=I+ID	ST2	070
IF(I)5,5,6	ST2	071
5 ID=1	ST2	072
I=IP+ID	ST2	073
6 L=I-ID	ST2	074
PS2(I,K)=PS2(L,K)+FLOAT(ID)*DPDR2(L,K)*(H2(I,K)+H2(L,K)	ST2	075
1)/2.	ST2	076
PTRS2(I,K)=PTR2(I,K)/PS2(I,K)	ST2	077
IF (PTRS2(I,K)-1.)19,19,3	ST2	078
19 PTRS2(I,K) = 1.0 + PHTOL	ST2	079
GO TO 3	ST2	080
7 IF(IS2(K))8,8,9	ST2	081
8 EXI=GAM(4,K)/(GAM(4,K)-1.)	ST2	082
CALL PHIM(EXI,ETAR(L,K),PHIX,PRC0IT)	ST2	083
PRUP=PTR2(IP,K)*PRC1I*PS2(L,K)/(PTR2(L,K)*PS2(IP,K))	ST2	084
1*(1.+PRTOL)	ST2	085
PRL0W=1.	ST2	086
GO TO 10	ST2	087
9 IS2(K)=IS2(K)+1	ST2	088
10 L = IHRC + 1	ST2	089

Listing of Code (continued)

IF (ICHOKE.EQ.L) PTRS2(IP,K) = PRUP	ST2 090
IF (WGT2(K)-WGT2C(K))12,15,11	ST2 091
11 PRLow= PTRS2(IP,K)	ST2 092
GO TO 13	ST2 093
12 PRUP= PTRS2(IP,K)	ST2 094
IS2(K)=1	ST2 095
13 WE=1.-WGT2(K)/WGT2C(K)	ST2 096
J=J+1	ST2 097
IF (J-32)29,18,18	ST2 098
29 IF (ICHOKE-L) 30,31,30	ST2 099
31 SCRIT= -WE	ST2 100
GO TO 15	ST2 101
30 IF (LOPIN)14,14,15	ST2 102
14 PRE=(PTRS2(IP,K)-PTRMC)/PTRS2(IP,K)	ST2 103
IF (ABS(PRE)-PR1OL)17,17,24	ST2 104
17 CONTINUE	ST2 105
IF (ABS(WE)-WTOL)15,15,23	ST2 106
24 PTRMO=PTRS2(IP,K)	ST2 107
WGT2C(K)=0.0	ST2 108
I=IP	ST2 109
ID=-1	ST2 110
IF (SCRIT)28,28,15	ST2 111
28 PTRS2(IP,K)=.5*(PRLow+PRUP)	ST2 112
IF (PTRS2(IP,K).LE.PRCRIT) PRCR=0.0	ST2 113
GO TO 3	ST2 114
23 SCRIT= 1.	ST2 115
15 IF (IRLOOP.EQ.0.) GO TO 25	ST2 116
18 WRITE(6,1000)K,PRUP,PRLow,WE,PRCRIT,J,WGT2(K),WGT2C(K),(WG2(L,K),	ST2 117
1 L=1,ISECT)	ST2 118
WRITE(6,1001)(PTRS2(L,K),L=1,ISECT)	ST2 119
1000 FORMAT(2X,2H K=I4, 2X,6H PRUP=F8.5,2X,6H PRLow=F8.5,2X,6H WE=	ST2 120
1F8.5,1X,7H PRCRIT=F8.5,2X,2H J=I4/	ST2 121
22X,6H WGT2=F8.3,2X,6H WGT2C=F8.3/	ST2 122
32X,6H WG2=6F8.3)	ST2 123
1001 FORMAT(2X,6H PTRS2=6F8.5)	ST2 124
25 CALL CHECK(J)	ST2 125
GO TO (20,21),J	ST2 126
20 CALL DIAGT(4)	ST2 127
GO TO 22	ST2 128
21 CALL LOOP	ST2 129
22 IF (SRFLAG) WRITE(6,20000)	*****
20000 FORMAT(45H AN EXIT HAS BEEN MADE FROM SUBROUTINE STA2)	*****
RETURN	*****
END	ST2 131

Listing of Code (continued)

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SUBROUTINE FLOW2(I)
CFLOW2
C   CALCULATE ROTOR EXIT SECTOR FLOW
C
  REAL MFSTOP
  LOGICAL PREVER,SRFLAG
  COMMON SRFLAG
  COMMON /SNTCP/G,AJ,PRPC,ICASE,PREVER,MFSTOP,JUMP,LOPIN,ISCASE,
1KN,GAMF,IP,SCRIT,PTRN,ISECT,KSTG,WTOL,HOTOL,PTOL,TRLOOP,LSTG,
2LRRC,IBRC,ICHOKE,ISORH,CHOKE,PTOPSL(6,8),PTRS2(6,8),TRDIAG,SC,RC,
3DELPR,PASS,IPC,LOPC,ISS
C
  COMMON /SINIT/H1(6,8),H2(6,8),DP1(6,8),DP1A(6,8),DP2(6,8),
1,DP2A(6,8),CSALF1(6,8),ALF1(6,8),CSBET2(6,8),BET2(6,8),RADSD(6,8),
2RADRD(6,8),ANN1(6,8),ANN2(6,8),ANN2A(6,8),ANN1A(6,8),U1A(6,8),
3U2(6,8),ANNO(6,8),PT0(6,8),TT0(6,8),ALPHA0(6,8),PTP(6,8)
C
  COMMON /SINPUT/ RSL,TSL,PSL,GAMSL,
1PTPS,PTIN,TTIN,WAIR,FAIR,DELC,DELL,DELA,AACS,VCTD,STG,SECT,EXPN,
2EXPP,EXPRE, RPM,PAF,SLI,STGCH,FNDJON,NAME(10),TITLE(10),PCNH(6,8),
3RV(6,8),GAM(6,8),DR(6,8),DT(6,8),RWG(6,8),ALPHAS(6,8),ALPHA1(6,8),
4ETAS(6,8),ETAS(6,8),CFS(6,8),ANNO(6,8),BETA1(6,8),BETA2(6,8),ETARFLW2
5R(6,8),ETAR(6,8),CFR(6,8),TFR(6,8),ANDOR(6,8),OMEGAS(6,8),AS0(6,8),
6,ASMP0(6,8),ACMN0(6,8),A1(6,8),A2(6,8),A3(6,8),A4(6,8),A5(6,8),A6(
76,8),OMEGAR(6,8),BSIA(6,8),RSMPIA(6,8),HCMNIA(6,8),B1(6,8),B2(6,8),
8,B3(6,8),B4(6,8),B5(6,8),B6(6,8),SESTH1(8),RERTH1(8)
C
  COMMON /SSTA2/V2(6,8),TTR2(6,8),PTR2(6,8),WG2(6,8),WGT2(8),TA2(8),
1
  PS2(6,8),PHI2(6,8)
C
  REAL MR2,M2 ,MF2
  COMMON /SFLOW2/TS2(6,8),CP2(8),W2(6,8),RHOS2(6,8),BET2E(6,8),RU2(6
1,8),VU2(6,8),DPUH2(6,8),VZ2(6,8),MR2(6,8),MF2(6,8),M2(6,8)
C
  DIMENSION P1AS2C(8),PHI2C(8),R2C(6,8),TS2C(6,8),RHOS2C(6,8),WG2C(6
1,8),CBET2E(6,8),AS2(6,8),RFF(6,8)
C
  IF(SRFLAG) WRITE(6,10000)
10000 FORMAT(44H AN ENTRY HAS BEEN MADE IN SUBROUTINE FLOW2 )
  K=KN
  EX=(GAM(4,K)-1.)/GAM(4,K)
C   ISENTROPIC ROTOR RELATIVE TEMPERATURE RATIO
10 PHI2(I,K)= PTRS2(I,K)**EX
  IF(OMEGAR(I,K))2,2,1
  1 CALL LOSS2(I,K)

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Listing of Code (continued)

C	EXIT TEMPERATURES	FLW2 043
2	TS2(I,K)=TTR2(I,K)*(1.-ETAR(I,K)*(1.-1./PHI2(I,K)))	FLW2 044
	IF(I-IP)6,3,6	FLW2 045
3	IF(GAMF)4,4,5	FLW2 046
4	TA2(K)=.5*(TTH2(I,K)+TS2(I,K))	FLW2 047
	CALL GAMMA(PTR2(I,K),TA2(K),FAIR,WAIR,GAM(4,K))	FLW2 048
5	EXI=GAM(4,K)/(GAM(4,K)-1.)	FLW2 049
	EX=1./EXI	FLW2 050
C	CRITICAL PRESSURE RATIO	FLW2 051
	CALL PHIM(EXI,ETAR(I,K),PHI2C(K),PLAS2C(K))	FLW2 052
C	SPECIFIC HEAT AT CONSTANT PRESSURE	FLW2 053
6	CP2(K)=HV(4,K)*EXI/AL	*****
C	RELATIVE EXIT VELOCITY	FLW2 055
	R2(I,K)=SQRT(2.*G*AJ*CP2(K)*(TTR2(I,K)-TS2(I,K)))	FLW2 056
C	EXIT PRESSURE	FLW2 057
	PS2(I,K)=PTR2(I,K)/PTRS2(I,K)	FLW2 058
C	EXIT DENSITY	FLW2 059
	RHOS2(I,K)=144.*PS2(I,K)/(RV(4,K)*TS2(I,K))	*****
C	TEST CRITICAL PRESSURE RATIO	FLW2 061
	IF(PTRS2(I,K)-PLAS2C(K))15,7,7	FLW2 062
7	IF(IP-1)27,8,22	FLW2 063
8	IF(PHPC)9,9,18	FLW2 064
9	PHPC=1.	FLW2 065
	PTH2(I,K)=PLAS2C(K)*(1.+PRTOL)	FLW2 066
	GO TO 10	FLW2 067
22	IF(PTRS2(I,K).LE.PTH2(IP,K)) GO TO 18	FLW2 068
	GO TO 13	FLW2 069
18	IF((1.EQ.1).OR.(1.EQ.ISECT)) SCRT=1.	FLW2 070
	GO TO 11	FLW2 071
11	CONTINUE	FLW2 072
	R2C(I,K)=SQRT(2.*G*AJ*CP2(K)*TTR2(I,K)*ETAR(I,K)*(FLW2 073
	1PHI2C(K)-1.)/PHI2C(K))	FLW2 074
	TS2C(I,K)=TTH2(I,K)*(1.-ETAR(I,K)*(1.-1./PHI2C(K)))	FLW2 075
	RHOS2C(I,K)=144.*PTH2(I,K)/(RV(4,K)*PLAS2C(K)*TS2C(I,K))	*****
	WG2C(I,K)=RHOS2C(I,K)*R2C(I,K)*ANN2(I,K)*CSHET2(I,K)	FLW2 077
12	WG2(I,K)=WG2C(I,K)	FLW2 078
	GO TO 14	FLW2 079
13	IF(PHPC-1.)15,15,24	FLW2 080
24	WG2(I,K)=RFF(I,K)*PTR2(I,K)/SQRT(TTH2(I,K))	FLW2 081
	GO TO 14	FLW2 082
C	OVEREXPANSION AFTER SUPERSONIC FLOW DECREASE	FLW2 083
14	CHET2E(I,K)=WG2(I,K)/(RHOS2(I,K)*R2(I,K)*ANN2(I,K))	FLW2 084
	BET2E(I,K)=ATAN2(SQRT(1.-CHET2E(I,K)*CHET2E(I,K)),CHET2E(I,K))	FLW2 085
	GO TO 16	FLW2 086
15	WG2(I,K)=RHOS2(I,K)*R2(I,K)*ANN2(I,K)*CSHET2(I,K)	FLW2 087
	CHET2E(I,K)=CSHET2(I,K)	FLW2 088
	BET2E(I,K)=BET2(I,K)	FLW2 089

Listing of Code (continued)

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      RFF(I,K)=WG2(I,K)*SQRT(TTR2(I,K))/PTR2(I,K)          FLW2 090
16  RU2(I,K)=R2(I,K)*SIN(UBET2E(I,K))                     FLW2 091
      VU2(I,K)=RU2(I,K)-U2(I,K)                             FLW2 092
      DPDR2(I,K)= (RHOS2(I,K)*VU2(I,K)*VU2(I,K)/(G*DP2(I,K)))*.01388889*****
      VZ2(I,K)=R2(I,K)*CBET2E(I,K)                         FLW2 094
      AS2(I,K)=SQRT(GAM(4,K)*G*RV(4,K)*TS2(I,K))           *****
      V2(I,K)=SQRT(VZ2(I,K)*VZ2(I,K)+VU2(I,K)*VU2(I,K))    FLW2 096
      M2(I,K)=V2(I,K)/AS2(I,K)                             FLW2 097
      MR2(I,K)=R2(I,K)/AS2(I,K)                             FLW2 098
      MF2(I,K)=MR2(I,K)*CHEI2E(I,K)                       FLW2 099
      IF(I.LT.1SECT) GO TO 17                              FLW2 100
      IF(PRPC.EQ.1.) PRPC=2.                                FLW2 101
17  CALL CHECK(J)                                           FLW2 102
      GO TO (19,21),J                                       FLW2 103
19  CALL DIAGT(4)                                           FLW2 104
21  IF(SRFLAG) WRITE(6,20000)                               *****
20000 FORMAT(45H AN EXIT HAS BEEN MADE FROM SUBROUTINE FLOW2 ) *****
      RETURN                                               *****
      END                                                  FLW2 106

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Listing of Code (continued)

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      SUBROUTINE LOSS2(I,K)                                LOS2 001
CLOSS2                                                    LOS2 002
C    CALCULATE ETA R FROM QUADRATIC POLYNOMIAL          LOS2 003
C                                                        LOS2 004
      REAL MFSTOP                                          LOS2 005
      LOGICAL PREVER,SRFLAG                                *****
      COMMON SRFLAG                                       *****
      COMMON /SNTCP/G,AJ,PRFC,ICASE,PREVER,MFSTOP,JUMP,LOPIN,ISCASE, LOS2 007
      1KN,GAMF,IP,SCHIT,PTHN,ISECT,KSTG,WOTOL,WHOTOL,PTOL,TRLOOP,LSTG, LOS2 008
      2LBRC,IBRC,ICHOKE,ISORH,CHOKE,PTOPS1(6,8),PTRS2(6,8),TRDIAG,SC,RC, LOS2 009
      3DELPR,PASS,IPC,LOPC,ISS                            LOS2 010
C                                                        LOS2 011
      COMMON /SINPUT/ HSL,TSL,PSL,GAMSL,                 *****
      1PTPS,PTIN,TTIN,WAIR,FAIR,UCLC,DEIL,DELA,AACS,VCTD,STG,SECT,EXPN, LOS2 013
      2EXPP,EXPRE, RPM,PAF,SLI,STGCH,FNUJOR,NAME(10),TITLE(10),PCNH(6), *****
      3RV(6,8),GAM(6,8),DR(6,8),UT(6,8),RWG(6,8),ALPHAS(6,8),ALPHA1(6,8), *****
      4ETARS(6,8),ETAS(6,8),CFS(6,8),AND0(6,8),BETA1(6,8),BETA2(6,8),ETARLOS2 016
      5R(6,8),ETAR(6,8),CFH(6,8),TFR(6,8),ANDUR(6,8),OMEGAS(6,8),AS0(6,8)LOS2 017
      6,ASMP0(6,8),ACMN0(6,8),A1(6,8),A2(6,8),A3(6,8),A4(6,8),A5(6,8),A6(LOS2 018
      76,8),OMEGAR(6,8),BSIA(6,8),BSMPIA(6,8),BSMNIA(6,8),B1(6,8),B2(6,8)LOS2 019
      8,B3(6,8),B4(6,8),B5(6,8),B6(6,8),SESTHI(8),RERTHI(8)      LOS2 020
C                                                        LOS2 021
      REAL MR1A                                           LOS2 022
      COMMON /SSTA1A/VU1A(6,8),WG1A(6,8),WGT1A(8),VZ1A(6,8), CP1A(8), LOS2 023
      1PS1A(6,8),RU1A(6,8),R1A(6,8),HET1A(6,8),RI(6,8),TTR1A(6,8),PTR1A(6,LOS2 024
      2,8),MR1A(6,8),TS1A(6,8)                               *****
C                                                        LOS2 026
      COMMON /SSTA2/V2(6,8),TTR2(6,8),PTR2(6,8),WG2(6,8),WGT2(8),TA2(8),LOS2 027
      1 PS2(6,8),PHI2(6,8)                                  LOS2 028
C                                                        LOS2 029
C                                                        LOS2 030
      IF(SRFLAG) WRITE(6,10000)                          *****
10000 FORMAT(44H AN ENTRY HAS BEEN MADE IN SUBROUTINE LOSS2 ) *****
      ETARR(I,K)=1.0                                       LOS2 031
      IF(RI(I,K))4,1,2                                     LOS2 032
      1 W1A2=OMEGAR(I,K)                                   LOS2 033
      GO TO 8                                              LOS2 034
      2 AS=R1(I,K)                                          LOS2 035
      AC=B2(I,K)                                           LOS2 036
      AQ=B3(I,K)                                           LOS2 037
      IF(BSMPIA(I,K)-RI(I,K))3,6,6                       LOS2 038
      3 WMWR=RI(I,K)/BSMPIA(I,K)                          LOS2 039
      AR=BSMPIA(I,K)/BSIA(I,K)                            LOS2 040
      GO TO 7                                              LOS2 041
      4 AS=B4(I,K)                                          LOS2 042
      AC=B5(I,K)                                          LOS2 043

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Listing of Code (continued)

AQ=H6(I,K)	LOS2 044
IF(RI(I,K)-HCMNIA(I,K))5,6,6	LOS2 045
5 WMWR=RI(I,K)/HCMNIA(I,K)	LOS2 046
AR=HCMNIA(I,K)/HSIA(I,K)	LOS2 047
GO TO 7	LOS2 048
6 WMWR=1.0	LOS2 049
AR=RI(I,K)/HSIA(I,K)	LOS2 050
7 W1A2=OMEGAR(I,K)*(1.+AR*AR*(AS+AR*(AC+AR*AQ)))*WMWR	LOS2 051
8 EX=(GAM(3,K)-1.)/GAM(3,K)	LOS2 052
ETAR(I,K)=(1.-(1./(PTH2(I,K)*(1.-W1A2)+W1A2))**EX)*PHI2(I,K)/	LOS2 053
1(PHI2(I,K)-1.)	LOS2 054
CALL CHECK(J)	LOS2 055
IF(SRFLAG) WRITE(6,20000)	*****
20000 FORMAT(45H AN EXIT HAS BEEN MADE FROM SUBROUTINE LOSS2)	*****
RETURN	LOS2 056
END	LOS2 057

Listing of Code (continued)

```

SUBROUTINE LOOP                                LOOP n01
CLOOP                                           LOOP n02
C        HANDLES ALL LOGIC FOR ITERATING TO OBTAIN EXACT CHOKE POINT- LOOP n03
C        UNDERFLOW, NO CHOKE INITIAL CHOKE, CHOKE ITERATION    LOOP n04
C        SUBCRITICAL, CHOKE ITERATION SUPERCRITICAL, MULTIPLE  LOOP n05
C        CHOKE, CHOKE ITERATION COMPLETE                      LOOP n06
C                                                         LOOP n07
C        REAL MFSTOP                                           LOOP n08
C        LOGICAL PREVER, SRFLAG                                *****
C        COMMON SRFLAG                                         *****
C        COMMON /SNTCP/G, AJ, PRFC, ICASE, PREVER, MFSTOP, JUMP, LOPIN, ISCASE, LOOP n10
C        1KN, GAMF, IP, SCHIT, PTRN, ISECT, KSTG, WTOL, RHOTOL, PRITOL, TRL00P, LSTG, LOOP n11
C        2LHRC, IBRC, ICHOKE, ISORH, CHOKE, PTOPSL(6,8), PTRS2(6,8), TRDIAG, SC, RC, LOOP n12
C        3DELPR, PASS, IPC, LOPC, ISS                          LOOP n13
C                                                         LOOP n14
C        COMMON /SINPUT/ RSL, TSL, PSL, GAMS1,                *****
C        1PTPS, PTIN, TTIN, WAIR, FAIR, UELC, UELL, DELA, AACS, BLLO, STG, SECT, EXPN, LOOP n16
C        2FXPP, EXPRF, RPM, PAF, SLI, STGCH, FNDJUH, NAME(10), TITLE(10), PCNH(6), *****
C        3RV(6,8), GAM(6,8), DR(6,8), DT(6,8), RWG(6,8), ALPHAS(6,8), ALPHA1(6,8), *****
C        4ETARS(6,8), ETAS(6,8), CFS(6,8), ANNO(6,8), HETA1(6,8), HETA2(6,8), ETARLOOP n19
C        5R(6,8), ETAR(6,8), CFH(6,8), TFR(6,8), ANDCH(6,8), OMEGAS(6,8), AS0(6,8) LOOP n20
C        6, ASMP0(6,8), ACMN0(6,8), A1(6,8), A2(6,8), A3(6,8), A4(6,8), A5(6,8), A6( LOOP n21
C        7, A7(6,8), OMEGAR(6,8), BSIA(6,8), RSMPIA(6,8), HCMNIA(6,8), B1(6,8), R2(6,8) LOOP n22
C        8, B3(6,8), B4(6,8), B5(6,8), B6(6,8), SESTHI(8), RERTHI(8)    LOOP n23
C                                                         LOOP n24
C        IF(SRFLAG) WRITE(6,10000)                                *****
10000 FORMAT(44H AN ENTRY HAS BEEN MADE IN SUBROUTINE LOOP ) *****
C        IJ=R+KSTG                                              LOOP n25
C        INCREASE BLADE ROW COUNTER                            LOOP n26
C        IBRC=IBRC+1                                           LOOP n27
C        TEST NEGATIVE SECTOR PRESSURE RATIO                  LOOP n28
C        IF (PTRN)18,1,1                                       LOOP n29
C        TEST CHOKE ITERATION ON BLADE ROW                    LOOP n30
C        1 IF (ICHOKE-IBRC)3,2,3                                LOOP n31
C        TEST INCREMENT TOLERANCE                             LOOP n32
C        2 IF (PRITOL-DELPR)3,3,4                              LOOP n33
C        TEST STATION FLOW CRITICAL                           LOOP n34
C        3 IF (SCHIT)5,5,6                                     LOOP n35
C        CHOKE ITERATION COMPLETE                             LOOP n36
C        4 ICHOKE=0                                             LOOP n37
C        IPC=IBRC                                              LOOP n38
C        ISS=IBRC                                              LOOP n39
C        ISORH=2+(IBRC/2)*2-IBRC                               LOOP n40
C        JL=(ISORH-1)*8+KN                                     LOOP n41
C        IF (JL-IJ)22,23,23                                    LOOP n42
22 DELPR=DELL                                                LOOP n43

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Listing of Code (continued)

24	LOPC=0	LOOP	n44
	CHOKE=1.	LOOP	n45
	LSTG=KN	LOOP	n46
	LBRC=IBRC-1	LOOP	n47
	GO TO 18	LOOP	n48
23	DELPR=DELA	LOOP	n49
	GO TO 24	LOOP	n50
	5 IF (ICHOKE-IBRC)18,7,18	LOOP	n51
C	TEST CHOKE ITERATION LOOP	LOOP	n52
	6 IF (ISS-IBRC)8,18,18	LOOP	n53
C	CHOKE ITERATION	LOOP	n54
C	ISORR = 1 FOR STATOR	LOOP	n55
C	= 2 FOR ROTOR	LOOP	n56
	7 DELPR=DELPR/2.	LOOP	n57
	JL=(ISORR-1)*8+LSTG	LOOP	n58
	PTOPSI(IP,JL)=PTOPSI(IP,JL)+DELPR	LOOP	n59
	GO TO 16	LOOP	n60
C	CHOKE HAS OCCURRED	LOOP	n61
	8 IF (ICHOKE)80,80,13	LOOP	n62
	80 J=(IBRC-2*(KN-1)-1)*8+KN	LOOP	n63
	WRITE(6,801)IBRC,PTOPSI(IP,J)	LOOP	n64
801	FORMAT(16X10HBLADE ROW 13,8H CHOKED,4X5HPTPS=F10.5)	LOOP	n65
C	TEST SINGLE CALCULATION POINT	LOOP	n66
	9 IF (DELC)18,18,10	LOOP	n67
C	TEST PREVIOUS CHOKE	LOOP	n68
	10 IF (IPC)11,11,12	LOOP	n69
C	SAVE COMBINATIONS PRIOR FIRST CHOKE	LOOP	n70
	11 LBRC=LBRC	LOOP	n71
	ISORRS=ISORR	LOOP	n72
	JL=(ISORR-1)*8+LSTG	LOOP	n73
	SPTPS=PTOPSI(IP,JL)-DELPR	LOOP	n74
	LSTGS=LSTG	LOOP	n75
	SDELPR=DELPR	LOOP	n76
	GO TO 13	LOOP	n77
	12 JL=LSTGS+(ISORRS-1)*8	LOOP	n78
	DELNU = (PTOPSI(IP,JL)-SPTPS)/4.	LOOP	n79
	IF (DELNU.LE.0.0001) DELNU = SDELPR/4.	LOOP	n80
	DELPR = DELNU	LOOP	n81
	SDELPR = DELNU	LOOP	n82
	WRITE(6,1201)IPC,IBRC,DELPR	LOOP	n83
1201	FORMAT(6X11HBLADE ROWS 15,5H AND 15,25H, CHOKED - INCREMENT NOW	LOOP	n84
	1F10.5)	LOOP	n85
	LBRC=LBRC	LOOP	n86
	LSTG=LSTGS	LOOP	n87
	ISORR=ISORRS	LOOP	n88
	PTOPSI(IP,JL) = SPTPS + SDELPR	LOOP	n89
	LOPC=10	LOOP	n90

Listing of Code (continued)

ICHOKE=0	LOOP 091
IPC=0	LOOP 092
ISS=0	LOOP 093
CHOKE=0.0	LOOP 094
GO TO 17	LOOP 095
C TEST PREVIOUS COMPLETE CALCULATION	LOOP 096
13 IF (PASS)15,15,14	LOOP 097
14 ICHCKE=IHRG	LOOP 098
DELPR=.5*DELPR	LOOP 099
15 JL=(ISORR-1)*8+LSTG	LOOP 100
PTOPS1(IP,JL)=PTOPS1(IP,JL)-DELPR	LOOP 101
C SET INDEX REGISTERS	LOOP 102
16 CONTINUE	LOOP 103
LOPC=LOPC+1	LOOP 104
C SET JUMP FOR CHOKE ITERATION	LOOP 105
17 JUMP=1	LOOP 106
GO TO 19	LOOP 107
C JUMP SET FOR NO CHOKE OR CHOKE COMPLETE	LOOP 108
18 JUMP=0	LOOP 109
C TEST LOOP-TRACE	LOOP 110
19 IF (THLOOP)21,21,20	LOOP 111
20 WRITE(6,2001)IHRG,LHRC,ISORR,KN,LSTG,IPC,ISS,ICHOKE,JUMP,LHRCS,	LOOP 112
1ISORRS,LSTGS,SPTPS,PTOPS1(IP,JL),DELPR,DELL,SCRIT,LOPC	LOOP 113
2001 FORMAT(3X12I5/3X4F10.5,F10.0,I10)	LOOP 114
21 IF(SRFLAG) WRITE(6,20000)	*****
20000 FORMAT(45H AN EXIT HAS BEEN MADE FROM SUBROUTINE LOOP)	*****
RETURN	*****
END	LOOP 116

Listing of Code (continued)

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SUBROUTINE STA2A                                ST2A 001
CSTA2A                                           ST2A 002
C DETERMINE INLET FLOW CONDITIONS TO ALL STATORS ST2A 003
C AFTER THE FIRST STATUR                        ST2A 004
C                                              ST2A 005
REAL MFSTOP                                     ST2A 006
LOGICAL PREVER,SRFLAG                          *****
COMMON SRFLAG                                *****
COMMON /SNTCP/G,AJ,PRFC,ICASE,PHEVER,MFSTOP,JUMP,LOPIN,ISCASE, ST2A 008
1KN,GAMF,IP,SCRIT,PTRN,ISECT,KSTG,WOTOL,RHOTOL,PRTOL,TRLOOP,LSTG, *****
2LBRC,IHRC,ICHOKE,ISORN,CHOKE,PTOPS1(6,8),PTRS2(6,8),TRDIAG,SC,RC, ST2A 010
3DELPR,PASS,IPC,LOPC,ISS                     ST2A 011
C                                              ST2A 012
COMMON /SINIT/H1(6,8),H2(6,8),DP0(6,8),DP1(6,8),DP1A(6,8),DP2(6,8) ST2A 013
1,DP2A(6,8),CSALF1(6,8),ALF1(6,8),CSHET2(6,8),BET2(6,8),RADSD(6,8), ST2A 014
2RADRD(6,8),ANN1(6,8),ANN2(6,8),ANN2A(6,8),ANN1A(6,8),U1A(6,8), ST2A 015
3U2(6,8),ANN0(6,8),PT0(6,8),TT0(6,8),ALPHA0(6,8),PTP(6,8) ST2A 016
C                                              ST2A 017
C                                              ST2A 018
COMMON /SINPUT/ RSL,TSL,PSL,GAMSI, *****
1PTPS,PTIN,TTIN,WAIR,FAIR,DELC,DELL,DELA,AACS,VCTD,STG,SECT,EXPN. ST2A 020
2EXPP,EXPRE, RPM,PAF,SLI,STGCH,FNDJOH,NAME(10),TITLE(10),PCNH(6), *****
3RV(6,8),GAM(6,8),DR(6,8),UT(6,8),RWG(6,8),ALPHAS(6,8),ALPHA1(6,8), *****
4ETARS(6,8),FTAS(6,8),CFS(6,8),AND0(6,8),BETA1(6,8),BETA2(6,8),ETAHST2A 023
5R(6,8),ETAR(6,8),CFR(6,8),TFR(6,8),ANDOH(6,8),OMEGAS(6,8),AS0(6,8) ST2A 024
6,ASMP0(6,8),ACMN0(6,8),A1(6,8),A2(6,8),A3(6,8),A4(6,8),A5(6,8),A6(6,8) ST2A 025
76,8),OMEGAR(6,8),HSIA(6,8),RSMPIA(6,8),HCMNIA(6,8),R1(6,8),R2(6,8) ST2A 026
8,R3(6,8),R4(6,8),R5(6,8),R6(6,8),SESTHI(8),RERTHI(8) ST2A 027
C                                              ST2A 028
C                                              ST2A 029
REAL M0                                         ST2A 030
COMMON /SSTA01/CP0(8), PS0(6,8),V0(6,8),TS0(6,8) ST2A 031
18),VU0(6,8),VZ0(6,8),MHUS0(6,8),PS1(6,8),WGT1(8),TA1(8),WG1(6,8), ST2A 032
2 DPDH1(6,8),SI(6,8), CP1(8),PHI1(6,8),TS1(6,8),V1(6,8) ST2A 033
3,RHOS1(6,8),ALF1E(6,8),VU1(6,8),VZ1(6,8),M0(6,8),WGT0(8),WG0(6,8) *****
C                                              ST2A 034
C                                              ST2A 035
REAL MR1A                                       ST2A 036
COMMON /SSTA1A/VU1A(6,8),WG1A(6,8),WGT1A(8),VZ1A(6,8), CP1A(8), ST2A 037
1PS1A(6,8),RU1A(6,8),R1A(6,8),HET1A(6,8),RI(6,8),TTR1A(6,8),PTR1A(6,8) ST2A 038
2,8),MR1A(6,8),TS1A(6,8) *****
C                                              ST2A 039
C                                              ST2A 040
COMMON /SSTA2/V2(6,8),TTR2(6,8),PTR2(6,8),WG2(6,8),WGT2(8),TA2(8), ST2A 041
1 PS2(6,8),PHI2(6,8) ST2A 042
C                                              ST2A 043
REAL MR2,M2, MF2                               ST2A 044
COMMON /SFLOW2/TS2(6,8),CP2(8),R2(6,8),RHUS2(6,8),HET2E(6,8),RU2(6,8) ST2A 045
1,8),VU2(6,8),DPDR2(6,8),VZ2(6,8),MR2(6,8),MF2(6,8),M2(6,8)

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Listing of Code (continued)

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REAL M2A,MF2A
COMMON /SSTA2A/WG2A(6,8),WGT2A(8),VU2A(6,8),VZ2A(6,8),PS2A(6,8),
1ALF2A(6,8),TT2A(6,8),FT2A(6,8),TTBAR(8),PTBAR(8),STT0(8),SPT0(8),
2M2A(6,8),MF2A(6,8),CP2A(8),V2A(6,8),TS2A(6,8),TAS(8),PAS(8),GAMS(8
3),CPS(8),DELMVD(6,8),HVBAR(8)
C
C      DIMENSION          TTTS2A(6,8)
C
C      IF(SRFLAG) WRITE(6,10000)
10000 FORMAT(44H AN ENTRY HAS BEEN MADE IN SUBROUTINE STA2A )
      K=KN
      ID=-1
      I=IP
      TS2A(I,K)=TS2(I,K)
      WR=RWG(5,K)/RWG(4,K)
      SUMT=0.0
      SUMLT=0.0
      SUMLP=0.0
      WGT2A(K)=WR*WGT2(K)
12  VU2A(I,K)=VU12(I,K)*DP2(I,K)/DP2A(I,K)
      WG2A(I,K)=WR*WG2(I,K)
      RHOS2A=RHOS2(I,K)
1  VZ2A(I,K)=WR*VZ2(I,K)*ANN2(I,K)*RHOS2(I,K)/(ANN2A(I,K)*RHOSTR)
      V2A(I,K)=SQRT(VU2A(I,K)*VU2A(I,K)+VZ2A(I,K)*VZ2A(I,K))
      IF(I-IP)4,2,4
2  IF( GAMF)3,3,4
3  TA2A =.5*(TTR2(I,K)+TS2A(I,K))
      CALL GAMMA(PTH2(IP,K),TA2A ,FATR,WAIR,GAM(5,K))
4  EX=(GAM(5,K)-1.)/GAM(5,K)
      EXI=1./EX
      CP2A(K)=RV(5,K)*EXI/AJ
      DELTS=(V2(I,K)*V2(I,K)-V2A(I,K)*V2A(I,K))/(2.*G*AJ*CP2A(K))
      TS2A(I,K)=TS2(I,K)+DELTS
      IF(TS2A(I,K).GT.0.) GO TO 32
      PREVER = .TRUE.
      MFSTOP = 2.
      GO TO 30
32  PS2A(I,K)=PS2(I,K)*(1.+DELTS/TS2(I,K))*EXI
      RHOS2A =144.*PS2A(I,K)/(RV(5,K)*TS2A(I,K))
      IF(ABS(RHOSTR-RHOS2A )-1.E-07)6,6,5
5  RHOSTR=RHOS2A
      GO TO 1
6  SALF2A =VU2A(I,K)/V2A(I,K)
      ALF2A(I,K)=ATAN2(SALF2A ,SQRT(1.-SALF2A *SALF2A ))
11 IF (I-IP)28,24,28

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ST2A 046
ST2A 047
ST2A 048
ST2A 049
ST2A 050
*****
ST2A 052
ST2A 053
ST2A 054
ST2A 055
*****
*****
ST2A 056
ST2A 057
ST2A 058
ST2A 059
ST2A 060
ST2A 061
ST2A 062
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ST2A 068
ST2A 069
ST2A 070
ST2A 071
ST2A 072
ST2A 073
ST2A 074
*****
ST2A 076
ST2A 077
ST2A 078
ST2A 079
ST2A 080
ST2A 081
ST2A 082
*****
ST2A 084
ST2A 085
ST2A 086
ST2A 087
ST2A 088
ST2A 089

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Listing of Code (continued)

24 IF (GAMF)25,25,26	ST2A 090
25 TAS(K)=.5*(TA1(K)+TA2(K))	ST2A 091
PAS(K)=.5*(PT0(IP,K)+PT2A(IP,K))	ST2A 092
CALL GAMMA(PAS(K),TAS(K),FAIR,WATR,GAMS(K))	ST2A 093
GO TO 27	ST2A 094
26 GAMS(K)=.5*(GAM(2,K)+GAM(4,K))	ST2A 095
27 E4=GAMS(K)/(GAMS(K)-1.)	ST2A 096
RVBAR(K)=.5*(RV(2,K)+RV(4,K))	*****
CPS(K)=RVBAR(K)*E4/AJ	*****
28 DELHVD(I,K)=(U1A(I,K)*VU1A(I,K)+U2(I,K)*VU2(I,K))/AJ/G	ST2A 098
M2A(I,K)=V2A(I,K)/SQRT(GAM(5,K)*G*RV(5,K)*TS2A(I,K))	*****
DELT=TFR(I,K)*DELHVD(I,K)/CPS(K)	ST2A 100
TT2A(I,K)=TT0(I,K)-DELT	ST2A 101
TTTS2A(I,K)=1.+(M2A(I,K)*M2A(I,K)*(GAM(5,K)-1.)/2.)	ST2A 102
PTPS2A=(TTTS2A(I,K))*EXI	ST2A 103
PT2A(I,K)=PS2A(I,K)*PIPS2A	ST2A 104
MF2A(I,K)=M2A(I,K)*COS(ALF2A(I,K))	ST2A 105
IF (ISECT-1)13,15,13	ST2A 106
13 I=I+ID	ST2A 107
IF (I)14,14,12	ST2A 108
14 ID=1	ST2A 109
I=IP+ID	ST2A 110
GO TO 12	ST2A 111
15 CONTINUE	ST2A 112
DO 16 I=1,ISECT	ST2A 113
RW=WG2A(I,K)/WG2A(K)	ST2A 114
TR=TT2A(I,K)/TT2A(IP,K)	ST2A 115
PR=PT2A(I,K)/PT2A(IP,K)	ST2A 116
SUMT=SUMT+RW*TR	ST2A 117
SUMLT=SUMLT+RW*ALOG(TH)	ST2A 118
16 SUMLP=SUMLP+RW*ALOG(PH)	ST2A 119
E3=GAM(5,K)/(GAM(5,K)-1.)	ST2A 120
TTBAR(K)=TT2A(IP,K)*SUMT	ST2A 121
PTBAR(K)=PT2A(IP,K)*EXP(SUMLP+E3*(ALOG(SUMT)-SUMLT))	ST2A 122
IF (K-KSTG)17,18,18	ST2A 123
17 STT0(K+1)=TTBAR(K)	ST2A 124
SPT0(K+1)=PTBAR(K)	ST2A 125
DO 23 I=1,ISECT	ST2A 126
29 SI(I,K+1)=ALF2A(I,K)-RADSD(I,K+1)	ST2A 127
IF(SI(I,K+1).GT. 1.570796) SI(I,K+1)= 1.570796	*****
IF(SI(I,K+1).LT.-1.570796) SI(I,K+1)=-1.570796	*****
IF(OMEGAS(I,K))8,8,7	ST2A 130
7 ETARS(I,K+1)=1.0	ST2A 131
EXPSI=0.	ST2A 132
GO TO 117	ST2A 133
8 IF(SI(I,K+1))9,9,10	ST2A 134
9 EXPSI=EXPN	ST2A 135

Listing of Code (continued)

	GO TO 117	ST2A 136
	10 EXPST=EXPP	ST2A 137
	117 IF (PAF-1.) 19,20,21	ST2A 138
C	UNIFORM PROFILES	ST2A 139
	19 PTP(I,K+1)=PTBAR(K)	ST2A 140
	PTO(I,K+1)= PTP(I,K+1)	ST2A 141
	1*(1.+(TTTS2A(I,K)-1.)*ETARS(I,K+1)*(COS(SI(I,K+1))**EXPSI))**EXI	ST2A 142
	2/(TTTS2A(I,K))**EXI	ST2A 143
	TTO(I,K+1)=TTBAR(K)	ST2A 144
	GO TO 23	ST2A 145
C	SAVE PROFILES	ST2A 146
	20 PTP(I,K+1)=PT2A(I,K)	ST2A 147
	PTO(I,K+1)= PTP(I,K+1)	ST2A 148
	1*(1.+(TTTS2A(I,K)-1.)*ETARS(I,K+1)*(COS(SI(I,K+1))**EXPSI))**EXI	ST2A 149
	2/(TTTS2A(I,K))**EXI	ST2A 150
	GO TO 22	ST2A 151
C	SMOOTH PRESSURE PROFILES	ST2A 152
	21 PTP(I,K+1)=PTBAR(K)*(TT2A(I,K)/TTBAR(K))**E3	*****
	PTO(I,K+1)= PTP(I,K+1)	ST2A 154
	1*(1.+(TTTS2A(I,K)-1.)*ETARS(I,K+1)*(COS(SI(I,K+1))**EXPSI))**EXI	ST2A 155
	2/(TTTS2A(I,K))**EXI	ST2A 156
	22 TTO(I,K+1)=TT2A(I,K)	ST2A 157
	23 CONTINUE	ST2A 158
	18 MFSTOP=MF2A(IP,K)/AACS	ST2A 159
	CALL CHECK(J)	ST2A 160
	GO TO (30,31),J	ST2A 161
	30 CALL DIAGT(5)	ST2A 162
	31 IF(SRFLAG) WRITE(6,20000)	*****
20000	FORMAT(45H AN EXIT HAS BEEN MADE FROM SUBROUTINE STA2A)	*****
	RETURN	*****
	END	ST2A 164

Listing of Code (continued)

```

SUBROUTINE STA1                                ST1  001
CSTA1                                          ST1  002
C    SATISFY CONTINUITY OF FLOW AT EXIT OF ALL STATORS    ST1  003
C    AFTER THE FIRST STATOR                      ST1  004
C                                                ST1  005
C    REAL MFSTOP                                  ST1  006
C    LOGICAL PREVER,SRFLAG                      *****
C    COMMON SRFLAG                             *****
C    COMMON /SNTCP/G,AJ,PRPC,ICASE,PREVER,MFSTOP,JUMP,LOPIN,ISCASE, ST1  008
1KN,GAMF,IP,SCHIT,PTNN,ISECT,KSTG,WIOL,HHOTOL,PTOL,TRLOOP,LSTG, *****
2LHRC,IHRC,IHCKE,ISORH,CHOKE,PTOPSL(6,8),PTRS2(6,8),TRDIAG,SC,RC, ST1  010
3DELPR,PASS,IPC,LOPC,ISS                      ST1  011
C                                                ST1  012
C    COMMON /SINIT/H1(6,8),H2(6,8),DP0(6,8),DP1(6,8),DP1A(6,8),DP2(6,8) ST1  013
1,DP2A(6,8),CSALF1(6,8),ALF1(6,8),CSHET2(6,8),HET2(6,8),RADSD(6,8), ST1  014
2RADHD(6,8),ANN1(6,8),ANN2(6,8),ANN2A(6,8),ANN1A(6,8),U1A(6,8), ST1  015
3U2(6,8),ANN0(6,8),PT0(6,8),TT0(6,8),ALPHA0(6,8),PTP(6,8) ST1  016
C                                                ST1  017
C    COMMON /SINPUT/ RSL,TSL,PSL,GAMSI, *****
1PTPS,PTIN,TTIN,WAIR,FAIR,DELC,DEIL,DELA,AACS,VCTD,STG,SECT,EXPN, ST1  019
2EXPP,EXPE, RPM,PAF,SLI,STGCH,FNDJOB,NAME(10),TITLE(10),PCNH(6), *****
3RV(6,8),GAM(6,8),DR(6,8),DT(6,8),RWG(6,8),ALPHAS(6,8),ALPHA1(6,8), *****
4ETARS(6,8),ETAS(6,8),LFS(6,8),AND0(6,8),BETA1(6,8),BETA2(6,8),ETARST1  022
5R(6,8),ETAR(6,8),CFR(6,8),TFR(6,8),ANDUR(6,8),OMEGAS(6,8),AS0(6,8) ST1  023
6,ASMP0(6,8),ACMN0(6,8),A1(6,8),A2(6,8),A3(6,8),A4(6,8),A5(6,8),A6( ST1  024
76,8),OMEGAR(6,8),BSIA(6,8),BSMPIA(6,8),HCMNIA(6,8),B1(6,8),B2(6,8) ST1  025
8,B3(6,8),B4(6,8),B5(6,8),B6(6,8),SESTHI(8),RERTHI(8) ST1  026
C                                                ST1  027
C    REAL M0                                      ST1  028
C    COMMON /SSTA01/CP0(8), PS0(6,8),V0(6,8),TS0(6,8) ST1  029
18),VU0(6,8),VZ0(6,8),RHOS0(6,8),PS1(6,8),WGT1(8),TA1(8),WG1(6,8), ST1  030
2 DPDR1(6,8),SI(6,8), CP1(8),PHI1(6,8),TS1(6,8),V1(6,8) ST1  031
3,RHCS1(6,8),ALF1E(6,8),VU1(6,8),VZ1(6,8),M0(6,8),WGT0(8),WG0(6,8) *****
C                                                ST1  033
C    REAL M2A,MF2A                                ST1  034
C    COMMON /SSTA2A/WG2A(6,8),WGT2A(8),VU2A(6,8),VZ2A(6,8),PS2A(6,8), ST1  035
1ALF2A(6,8),TT2A(6,8),PT2A(6,8),TTBAR(8),PTBAR(8),STT0(8),SPT0(8), ST1  036
2M2A(6,8),MF2A(6,8),CP2A(8),V2A(6,8),TS2A(6,8),TAS(8),PAS(8),GAMS(8) ST1  037
3,CPS(8),DELHVD(6,8),HVBAR(8) *****
C                                                ST1  039
C    DIMENSION WGT1C(8),LC1(8),FFA1(6,8)          ST1  040
C                                                ST1  041
C                                                ST1  042
C    IF(SRFLAG) WRITE(6,10000) *****
10000 FORMAT(44H AN ENTRY HAS BEEN MADE IN SUBROUTINE STA1 ) *****
K=KN *****

```

Listing of Code (continued)

J=1	ST1 043
SCRIT=0.0	ST1 044
PTRM=1.	ST1 045
WR1=RWG(1,K)/RWG(5,K-1)	*****
WR=RWG(2,K)/RWG(5,K-1)	ST1 046
DO 1 I=1,ISECT	ST1 047
WG0(I,K)=WR1*WG2A(I,K-1)	*****
WG1(I,K)=WR*WG2A(I,K-1)	ST1 048
ALPHA0(I,K)=ALF2A(I,K-1)	ST1 049
PS0(I,K)=PS2A(I,K-1)	ST1 050
V0(I,K)=V2A(I,K-1)	ST1 051
TS0(I,K)=TS2A(I,K-1)	ST1 052
VU0(I,K)=VU2A(I,K-1)	ST1 053
VZ0(I,K)=VZ2A(I,K-1)	ST1 054
M0(I,K)=M2A(I,K-1)	ST1 055
1 CONTINUE	*****
CP0(K)=CP2A(K-1)	*****
WGT0(K)=WR1*WGT2A(K-1)	*****
WGT1(K)=WR*WGT2A(K-1)	*****
I=IP	ST1 057
ID=-1	ST1 058
WGT1C(K)=0.0	ST1 059
LC1(K)=0	ST1 060
IF(ICHOKE)17,17,16	ST1 061
17 IF(LOPIN)18,18,16	ST1 062
18 IF(GAMF)2,2,3	ST1 063
2 TA1(K)=.95*TT0(IP,K)	ST1 064
CALL GAMMA(PT0(IP,K),IA1(K),FAIR,WAIR,GAM(2,K))	ST1 065
3 FFA1(I,K)=WG1(I,K)*SQRT(TT0(I,K))/(144.*PT0(I,K)*ANN1(I,K)	ST1 066
1*CSALF1(I,K))	ST1 067
CALL PRATIO(FFA1(I,K),GAM(2,K),RV(2,K),PT0PS1(I,K),PRTOL)	*****
16 CALL FLOW1(I)	ST1 069
IF(PHFEVER) GO TO 25	ST1 070
WGT1C(K)=WGT1C(K)+WG1(I,K)	ST1 071
L=1	ST1 072
IF(PT0PS1(I,K).LE.PT0PS1(IP,K)) L=I	ST1 073
IF(ISECT-I)7,7,4	ST1 074
4 I=I+ID	ST1 075
IF(I)5,5,6	ST1 076
5 ID=1	ST1 077
I=IP+ID	ST1 078
6 L=I-ID	ST1 079
PS1(I,K)=PS1(L,K)+FLOAT(ID)*DPDR1(L,K)*(H1(I,K)+H1(L,K))/2.	ST1 080
PT0PS1(I,K)=PT0(I,K)/PS1(I,K)	ST1 081
GO TO 16	ST1 082
7 IF(LC1(K))8,8,9	ST1 083
8 LC1(K)=1	ST1 084

Listing of Code (continued)

EX=GAM(2,K)/(GAM(2,K)-1.)	ST1	085
CALL PHIM(EX,ETAS(L,K),PHIX,PRCRIT)	ST1	086
PRUP= PT0PS1(IP,K)*PRCRIT/PT0PS1(L,K)	ST1	087
1*(1.+PRTOL)	ST1	088
PRLOW=1.0	ST1	089
GO TO 10	ST1	090
9 LC1(K)=LC1(K)+1	ST1	091
10 L = ICHC + 1	ST1	092
IF (ICHOKE.EQ.L) PT0PS1(IP,K) = PRUP	ST1	093
IF (WGT1(K)-WGT1C(K))1<.15.11	ST1	094
11 PRLOW=PT0PS1(IP,K)	ST1	095
GO TO 13	ST1	096
12 PRUP=PT0PS1(IP,K)	ST1	097
13 WE=1.-WGT1(K)/WGT1C(K)	ST1	098
J=J+1	ST1	099
IF (J-32)29,22,22	ST1	100
29 IF (ICHOKE=L) 30,31,30	ST1	101
31 SCRIT= -WE	ST1	102
GO TO 15	ST1	103
30 IF (LOPIN)14,14,15	ST1	104
14 PRE=(PT0PS1(IP,K)-PTRM0)/PT0PS1(IP,K)	ST1	105
IF (ABS(PRE)-PRTOL)21,21,27	ST1	106
21 CONTINUE	ST1	107
IF (ABS(WE)-WTOL)15,15,20	ST1	108
27 PTRM0=PT0PS1(IP,K)	ST1	109
WGT1C(K)=0.0	ST1	110
I=IP	ST1	111
ID=-1	ST1	112
IF (SCRIT)19,19,15	ST1	113
19 PT0PS1(IP,K)=.5*(PRLOW+PRUP)	ST1	114
IF (PT0PS1(IP,K).LE.PRCRIT) PRPC=0.	ST1	115
GO TO 16	ST1	116
20 SCRIT= 1.	ST1	117
15 IF (TRLOOP.EQ.0.) GO TO 28	ST1	118
22 WRITE(6,1000)K,PRUP,PRLOW,WE,PRCRIT,J,WGT1(K),WGT1C(K),(WG1(L,K),	ST1	119
1 L=1,ISECT)	ST1	120
WRITE(6,1001)(PT0PS1(L,K),L=1,ISECT)	ST1	121
1000 FORMAT(2X,2HK=14, 2X,6H PRUP=F8.5,2X,6HPRLOW=F8.5,2X,6H WE=	ST1	122
1F8.5,1X,7HPRCRIT=F8.5,2X,2HJ=14/	ST1	123
22X,6H WGT1=F8.3,2X,6HWGT1C=F8.3/	ST1	124
32X,6H WG1=6F8.3)	ST1	125
1001 FORMAT(1X,7HPT0PS1=6F8.5)	ST1	126
28 CALL CHECK(J)	ST1	127
GO TO (23,24),J	ST1	128
23 CALL DIAGT(2)	ST1	129
GO TO 25	ST1	130
24 CALL LOOP	ST1	131

Listing of Code (continued)

```
      25 IF(SHFLAG) WRITE(6,20000)
20000 FORMAT(45H AN EXIT HAS BEEN MADE FROM SUBROUTINE STA1 )
      RETURN
      END
```

```
*****
*****
*****
ST1  133
```

Listing of Code (continued)

```

SUBROUTINE OVHALL
OVLL 001
COVRALL
OVLL 002
C PURPOSE IS TO CALCULATE STAGE PERFORMANCE VALUES
OVLL 003
C AFTER FLOW ITERATION IS COMPLETED THROUGH THE LAST STAGE
OVLL 004
C
OVLL 005
REAL MFSTOP
OVLL 006
LOGICAL PREVER,SRFLAG
*****
COMMON SRFLAG
*****
COMMON /SNTCP/G,AJ,PRFC,ICASE,PREVER,MFSTOP,JUMP,LOPIN,ISCASE,
OVLL 008
1KN,GAMF,IP,SCRIT,PTRN,ISECT,KSTG,WTOL,RHOTOL,PRTOL,TRLOOP,LSTG,
OVLL 009
2LBRC,IBHC,ICHUKE,ISORR,CHUKE,PTOPSI(6,8),PTRS2(6,8),TRDIAG,SC,RC,
OVLL 010
3DELPR,PASS,IPC,LOPC,ISS
OVLL 011
C
OVLL 012
COMMON /SINIT/H1(6,8),H2(6,8),DP0(6,8),DP1(6,8),DP1A(6,8),DP2(6,8)
OVLL 013
1,DP2A(6,8),CSALF1(6,8),ALF1(6,8),CSHET2(6,8),HET2(6,8),RADSD(6,8),
OVLL 014
2RADRD(6,8),ANN1(6,8),ANN2(6,8),ANN0(6,8),ANN1A(6,8),U1A(6,8),
OVLL 015
3U2(6,8),ANN0(6,8),PT0(6,8),TT0(6,8),ALPHA0(6,8),PTP(6,8)
OVLL 016
C
OVLL 017
COMMON /SINPUT/ RSL,TSL,PSL,GAMSL,
*****
1PTPS,PTIN,TTIN,WAIR,FAIR,DELC,DELL,DELA,AACS,VCTD,STG,SECT,EXPN,
OVLL 019
2EXPP,EXPRE, RPM,PAF,SLI,STGCH,FNDJOR,NAME(10),TITLE(10),PCNH(6),
*****
3RV(6,8),GAM(6,8),DR(6,8),DT(6,8),RWG(6,8),ALPHAS(6,8),ALPHA1(6,8),
*****
4ETARS(6,8),ETAS(6,8),CFS(6,8),AND0(6,8),BETA1(6,8),BETA2(6,8),ETAROVLL 022
5R(6,8),ETAR(6,8),CFR(6,8),TFR(6,8),ANDCR(6,8),OMEGAS(6,8),AS0(6,8)
OVLL 023
6,ASMP0(6,8),ACMN0(6,8),A1(6,8),A2(6,8),A3(6,8),A4(6,8),A5(6,8),A6(
OVLL 024
7,8),OMEGAR(6,8),BSIA(6,8),RSMPIA(6,8),HCMNIA(6,8),B1(6,8),B2(6,8)
OVLL 025
8,B3(6,8),B4(6,8),B5(6,8),B6(6,8),SESTHI(8),RERTHI(8)
OVLL 026
C
OVLL 027
REAL M0
OVLL 028
COMMON /SSTA01/CP0(8),
OVLL 029
18),VU0(6,8),VZ0(6,8),RHOS0(6,8),PS1(6,8),WGT1(8),TA1(8),WG1(6,8),
OVLL 030
2
DPDR1(6,8),SI(6,8),
CP1(8),PHI1(6,8),TS1(6,8),V1(6,8)
OVLL 031
3,RHOS1(6,8),ALF1E(6,8),VU1(6,8),VZ1(6,8),M0(6,8),WGT0(8),WG0(6,8)
*****
OVLL 033
C
OVLL 034
REAL MR1A
OVLL 035
COMMON /SSTA1A/VU1A(6,8),WG1A(6,8),WGT1A(8),VZ1A(6,8),
CP1A(8),
OVLL 036
1PS1A(6,8),RU1A(6,8),R1A(6,8),RET1A(6,8),RI(6,8),TTR1A(6,8),PTR1A(6,8)
OVLL 036
2,8),MR1A(6,8),TS1A(6,8)
*****
COMMON /SSTA2/V2(6,8),TTR2(6,8),PTR2(6,8),WG2(6,8),WGT2(8),TA2(8),
OVLL 038
1
PS2(6,8),PHI2(6,8)
OVLL 039
C
OVLL 040
REAL MR2,M2
OVLL 041
COMMON /SFLOW2/TS2(6,8),CP2(8),R2(6,8),RHOS2(6,8),BET2E(6,8),RU2(6,8)
OVLL 042
1,8),VU2(6,8),DPDR2(6,8),VZ2(6,8),MR2(6,8),MF2(6,8),M2(6,8)
OVLL 043
C
OVLL 044
REAL M2A,MF2A
OVLL 045

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Listing of Code (continued)

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COMMON /SSTA2A/WG2A(6,8),WGT2A(8),VU2A(6,8),VZ2A(6,8),PS2A(6,8), OVLL n46
1ALF2A(6,8),TT2A(6,8),FT2A(6,8),TTBAR(8),PTBAR(8),STT0(8),SPT0(8), OVLL n47
2M2A(6,8),MF2A(6,8),CP2A(8),V2A(6,8),TS2A(6,8),TAS(8),PAS(8),GAMS(8)OVLL n48
3),CPS(8),DELHVD(6,8),HVBAR(H) *****
C OVLL n50
COMMON /SOVRAL/DELHT(6,8),DELHTI(6,8),DELHSI(6,8),DEHATI(6,8), OVLL n51
1ETATT(6,8),ETATS(6,8),ETATAT(6,8) OVLL n52
C OVLL n53
REAL MIS(H),MIRS(8),MM1AR(8),MH2T(8) OVLL n54
DIMENSION SA0(8),SIS(8),SB1A(8),SIR(8),SA2(8),THCR(8),EPSI(8),DELTOVLL n55
1(8),SETATT(8),SETATS(8),SETAAT(8),SWRTP(8),SNRT(8),SDHT(8),SETHC(8)OVLL n56
2),SNRTHC(8),SWRTED(8),SPTPT2(8),SPTPS2(8),ST2TT0(8),STRIT0(8),UPS(OVLL n57
3),UPUPS(8),URS(8),URLRS(8),VIS(8),UPVIS(8),URVIS(8),PSIPS(8),PSIROVLL n58
4S(8),RXP(8),RXR(8),DBETAR(8),DELHTS(8),DEHTIS(8),DEHSIS(8),DHATIS(OVLL n59
5),PAT2A(6,8) *****
C OVLL n61
C ***** CARD DELETED ***** *****
C OVLL n63
C OVLL n64
C *****
10000 IF(SHFLAG) WRITE(6,10000) *****
FORMAT(44H AN ENTRY HAS BEEN MADE IN SUBROUTINE OVRALL) *****
STT0(1)=TTIN OVLL n65
SPT0(1)=PTIN OVLL n66
RGO=0.0 *****
TAO=0.0 OVLL n67
PAO=0.0 OVLL n68
GAMC=0.0 OVLL n69
NUPUP=0.0 OVLL n70
NURUR=0.0 OVLL n71
NDELHT=0.0 OVLL n72
5 E1=GAMSL/(GAMSL-1.) OVLL n73
DO 17 K=1,KSTG OVLL n74
RGO=RGO+HVBAR(K) *****
IF (GAMF)1,1,2 OVLL n75
1 TAO=TAO+TAS(K) OVLL n76
PAO=PAO+PAS(K) OVLL n77
GO TO 3 OVLL n78
2 GAMC=GAMC+GAMS(K) OVLL n79
3 E2=GAM(1,K)/(GAM(1,K)-1.) OVLL n80
E3=GAM(5,K)/(GAM(5,K)-1.) OVLL n81
E4=GAMS(K)/(GAMS(K)-1.) OVLL n82
E5=1./E4 OVLL n83
DELHTS(K)=0.0 OVLL n84
DEHTIS(K)=0.0 OVLL n85
DEHSIS(K)=0.0 OVLL n86
DHATIS(K)=0.0 OVLL n87
DO 6 I=1,ISECT OVLL n88

```

Listing of Code (continued)

```

RW=AG2A(I,K)/#GT2A(K)
DELHT(I,K)=DELHVN(I,K)*TFR(I,K)
DELHTI(I,K)=CPS(K)*TT0(I,K)*(1.-(PT2A(I,K)/PTP(I,K))**E5)
ETATT(I,K)=DELHT(I,K)/DELHTI(I,K)
DELHSI(I,K)=CPS(K)*TT0(I,K)*(1.-(PS2A(I,K)/PTP(I,K))**E5)
ETATS(I,K)=DELHT(I,K)/DELHSI(I,K)
PAT2A(I,K)=PS2A(I,K)*(1.+(GAM(S,K)-1.)*MF2A(I,K)*MF2A(I,K)
1/2.)*E3
DEHATI(I,K)=CPS(K)*TT0(I,K)*(1.-(PAT2A(I,K)/PTP(I,K))**E5)
ETATAT(I,K)=DELHT(I,K)/DEHATI(I,K)
DELHTS(K)=DELHTS(K)+R**DELHT(I,K)
DEHTIS(K)=DEHTIS(K)+R**DELHTI(I,K)
DEHSIS(K)=DEHSIS(K)+R**DELHSI(I,K)
DHATIS(K)=DHATIS(K)+R**DEHATI(I,K)
6 CONTINUE
13 SA0(K)=ALPHA0(IP,K)*57.2958
SIS(K)=SI(IP,K)*57.2958
SB1A(K)=BET1A(IP,K)*57.2958
SIR(K)=HI(IP,K)*57.2958
SA2(K)=ALF2A(IP,K)*57.2958
THCR(K)=GAM(1,K)*(GAMSL+1.)*RV(j,K)*STT0(K)/
1(GAMSL*(GAM(1,K)+1.)*HSL*TSL)
EPSI(K)=GAMSL*((GAM(1,K)+1.)/2.)*E2/(GAM(1,K)*((GAMSL
1+1.)/2.)*E1)
DELT(K)=SPT0(K)/PSL
SETATT(K)=DELHTS(K)/DEHTIS(K)
SETATS(K)=DELHTS(K)/DEHSIS(K)
SETAAT(K)=DELHTS(K)/DHATIS(K)
C ***** CARD DELETED*****
SWRTP(K)=WGTO(K)*SQRT(STT0(K))/SPT0(K)
SNRT(K)=RPM/SQRT(STT0(K))
SOHT(K)=DELHTS(K)/STT0(K)
SETHC(K)=DELHTS(K)/THCR(K)
RTHCR=SQRT(THCR(K))
SNRTHC(K)=RPM/RTHCR
SWRTED(K)=WGTO(K)*RTHCR*EPSI(K)/DELT(K)
SPTPT2(K)=SPT0(K)/PTBAR(K)
SPTPS2(K)=SPT0(K)/PS2(IP,K)
ST2TT0(K)=TTBAR(K)/STT0(K)
STRIT0(K)=TTR1A(IP,K)/STT0(K)
UPS(K)=.5*(U1A(IP,K)+U2(IP,K))
UPUPS(K)=UPS(K)*UPS(K)
OUPUP=OUPUP+UPUPS(K)
URS(K)=.5*(U1A(1,K)*CH(3,K)/DPLA(1,K)+U2(1,K)*DR(4,K)/DP2(1,K))
URURS(K)=URS(K)*URS(K)
OURUR=OURUR+URURS(K)
ODELHT=ODELHT+DELHTS(K)
OVLL 089
OVLL 090
OVLL 091
OVLL 092
OVLL 093
OVLL 094
OVLL 095
OVLL 096
OVLL 097
OVLL 098
OVLL 099
OVLL 100
OVLL 101
OVLL 102
OVLL 103
OVLL 104
OVLL 105
OVLL 106
OVLL 107
OVLL 108
*****
OVLLA109
OVLL 110
OVLL 111
OVLL 112
OVLL 113
OVLL 114
OVLL 115
*****
OVLL 117
OVLL 118
OVLL 119
OVLL 120
OVLL 121
OVLL 122
OVLL 123
OVLL 124
OVLL 125
OVLL 126
OVLL 127
OVLL 128
OVLL 129
OVLL 130
OVLL 131
OVLL 132
OVLL 133
OVLL 134

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Listing of Code (continued)

IF (DELHSI(IP,K))14,14,15	OVLL 135
14 VIS(K)=1.	OVLL 136
GO TO 16	OVLL 137
15 VIS(K)=SQRT(2.*G*AJ*DELHSI(IP,K))	OVLL 138
16 UPVIS(K)=UPS(K)/VIS(K)	OVLL 139
URVIS(K)=URS(K)/VIS(K)	OVLL 140
PSIPS(K)=G*AJ*DELHTS(K)/(2.*UPUPS(K))	OVLL 141
PSIRS(K)=G*AJ*DELHTS(K)/(2.*URURS(K))	OVLL 142
RXP(K)=1.-(1.-(PS1(IP,K)/PTP(IP,K))*E5)/(1.-(PS2(IP,K)/	OVLL 143
PTP(IP,K))*E5)	OVLL 144
VUIR=VUI(1,K)*OP1(1,K)/DR(2,K)	OVLL 145
V1R=SQRT(VUIR**2+VZ1(1,K)**2)	OVLL 146
PH1R=1./(1.-V1R**2/(2.*G*AJ*CP1(K)*TT0(1,K)*ETAS(1,K)))	OVLL 147
PTPS1R=PH1R*(GAM(2,K)/(GAM(2,K)-1.))*PTP(1,K)/PT0(1,K)	OVLL 148
RXR(K)=1.-(1.-(1./PTPS1R))*E5)/(1.-(PS2(1,K)/PTP(1,K))*E5)	OVLL 149
DBETAR(K)=(HET1A(1,K)*BET2E(1,K))*57.2958	OVLL 150
MIS(K)=V1(IP,K)/SQRT(GAM(2,K)*G*PV(2,K)*TS1(IP,K))	*****
TS1R=TT0(1,K)-V1R**2/(2.*G*AJ*CP1(K))	OVLL 152
MIRS(K)=V1R/SQRT(GAM(2,K)*G*RV(2,K)*TS1R)	*****
VUIAR=VUIA(1,K)*OP1A(1,K)/DR(3,K)	OVLL 154
V1AR=SQRT(VUIAR**2+VZ1A(1,K)**2)	OVLL 155
TS1AR=TT0(1,K)-V1AR**2/(2.*G*AJ*CP1A(K))	OVLL 156
RU1AR=VUIAR-U1A(1,K)*LR(3,K)/DP1A(1,K)	OVLL 157
R1AR=SQRT(RU1AR**2+VZ1A(1,K)**2)	OVLL 158
MH1AR(K)=R1AR/SQRT(GAM(3,K)*G*RV(3,K)*TS1AR)	*****
VU2T=VU2(ISECT,K)*DP2(ISECT,K)/DT(4,K)	OVLL 160
V2T=SQRT(VU2T**2+VZ2(ISECT,K)**2)	OVLL 161
TS2T=TS2(ISECT,K)+(V2(ISECT,K)**2-V2T**2)/(2.*G*AJ*CP2(K))	OVLL 162
RU2T=VU2T+U2(ISECT,K)*DT(4,K)/DP2(ISECT,K)	OVLL 163
R2T=SQRT(RU2T**2+VZ2(ISECT,K)**2)	OVLL 164
MR2T(K)=R2T/SQRT(GAM(4,K)*G*RV(4,K)*TS2T)	*****
17 CONTINUE	OVLL 166
IF (GAMF)4,4,7	OVLL 167
4 TAO=TAO/STG	OVLL 168
PAO=PAO/STG	OVLL 169
CALL GAMMA(PAU,TAO,FAIR,WAIR,GAM0)	OVLL 170
GO TO 8	OVLL 171
7 GAM0=GAM0/STG	OVLL 172
8 EO=(GAM0-1.)/GAM0	OVLL 173
RG0=RG0/STG	*****
CPO=RG0/EO/AJ	*****
K=KSTG	OVLL 175
ODEHTI = 0.	OVLL 176
ODEHSI = 0.	OVLL 177
ODHATI = 0.	OVLL 178
DO 9 I=1,ISECT	OVLL 179
RW=WG2A(I,K)/WGT2A(K)	OVLL 180

Listing of Code (continued)

ODEHTI = CPO*TT0(I,1)*(1.-(PT2A(I,K)/PTP(I,1))**EO)*RW+ODEHTI	OVLL 181
ODEHSI = CPO*TT0(I,1)*(1.-(PS2A(I,K)/PTP(I,1))**EO)*RW+ODEHSI	OVLL 182
9 ODHATI = CPO*TT0(I,1)*(1.-(PAT2A(I,K)/PTP(I,1))**EO)*RW+ODHATI	OVLL 183
OPSIP=G*AJ*ODELHT/(2.*OUPUP)	OVLL 184
OPSIR=G*AJ*ODELHT/(2.*OURUR)	OVLL 185
OWRTP=SWRTP(1)	OVLL 186
OWNED=SWRTED(1)*SNRTHC(1)/60.	OVLL 187
ONRTHC=SNRTHC(1)	OVLL 188
ONRT=SNRT(1)	OVLL 189
ODHT=ODELHT/TTIN	OVLL 190
OPT0T2=PTIN/PTBAR(KSTG)	OVLL 191
OPT0S2=PTIN/PS2(IP,KSIG)	OVLL 192
OPTAT2=PTIN/PAT2A(IP,KSTG)	OVLL 193
OETATI=ODELHT/ODEHTI	OVLL 194
OETATS=ODELHT/ODEHSI	OVLL 195
OETAAT=ODELHT/ODHATI	OVLL 196
OETHC=ODELHT/THCH(1)	OVLL 197
C	OVLL 198
C PRINT OUT FOR STAGE PERFORMANCE	OVLL 199
I=1	OVLL 200
WRITE(6,1000)NAME,TITLE,ICASE,ISCASE	OVLL 201
1000 FORMAT(1H1,21X,29HNASA TURBINE COMPUTER PROGRAM /6X,10A6/	OVLL 202
1 6X,10A6/ 30X,6HCASE 13,1H.,13/28X,1/HSTAGE PERFORMANCE /19X	*****
27HSTAGE 1,6X,7HSTAGE 2,6X,7HSTAGE 3,6X,7HSTAGE 4/)	OVLL 204
IF (KSTG-4)19,19,18	OVLL 205
18 KS=4	OVLL 206
GO TO 20	OVLL 207
19 KS=KSTG	OVLL 208
20 WRITE(6,1001)(STT0(K),K=1,KS)	OVLL 209
1001 FORMAT(2X,12H TTBAR 02X,F10.1,3X,F10.1,3X,F10.1,3X,F10.1)	*****
WRITE(6,1002)(SPT0(K),K=1,KS)	OVLL 211
1002 FORMAT(2X,12H PTBAR 02X,F10.3,3X,F10.3,3X,F10.3,3X,F10.3)	*****
WRITE(6,1003)(WGTO(K),K=1,KS)	*****
1003 FORMAT(2X,12H WG 02X,F10.3,3X,F10.3,3X,F10.3,3X,F10.3)	*****
WRITE(6,1004)(DELHTS(K),K=1,KS)	OVLL 215
1004 FORMAT(2X,12H DEL H2X,F10.3,3X,F10.3,3X,F10.3,3X,F10.3)	*****
WRITE(6,1005)(SWRTP(K),K=1,KS)	OVLL 217
1005 FORMAT(2X,12H WRT/P2X,F10.3,3X,F10.3,3X,F10.3,3X,F10.3)	*****
WRITE(6,1006)(SDHT(K),K=1,KS)	OVLL 219
1006 FORMAT(2X,12H DH/TTBAR02X,F10.5,3X,F10.5,3X,F10.5,3X,F10.5)	*****
WRITE(6,1007)(SNRT(K),K=1,KS)	OVLL 221
1007 FORMAT(2X,12H N/RT2X,F10.3,3X,F10.3,3X,F10.3,3X,F10.3)	*****
WRITE(6,1008)(SETATT(K),K=1,KS)	OVLL 223
1008 FORMAT(2X,12H ETA TT2X,F10.5,3X,F10.5,3X,F10.5,3X,F10.5)	*****
WRITE(6,1009)(SETATS(K),K=1,KS)	OVLL 225
1009 FORMAT(2X,12H ETA TS2X,F10.5,3X,F10.5,3X,F10.5,3X,F10.5)	*****
WRITE(6,1010)(SETAAT(K),K=1,KS)	OVLL 227

Listing of Code (continued)

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1010 FORMAT(2X,12H      F1A AT2X,F10.5,3X,F10.5,3X,F10.5,3X,F10.5) *****
      WRITE(6,1011) (PT0PS1(IP,K),K=I,KS) OVLL 229
1011 FORMAT(2X,12H      PT0/PS12X,F10.3,3X,F10.3,3X,F10.3,3X,F10.3) *****
      WRITE(6,1012) (SPTPT2(K),K=I,KS) OVLL 231
1012 FORMAT(1X,13HPTBAR0/PTBAR2,2X,F10.3,3X,F10.3,3X,F10.3,3X,F10.3) *****
      WRITE(6,1013) (SPTPS2(K),K=I,KS) OVLL 233
1013 FORMAT(2X,12H      PTBAR0/PS22X,F10.3,3X,F10.3,3X,F10.3,3X,F10.3) *****
      WRITE(6,1014) (PTRS2(IF,K),K=I,KS) OVLL 235
1014 FORMAT(2X,12H      PTH2/PS22X,F10.3,3X,F10.3,3X,F10.3,3X,F10.3) *****
      WRITE(6,1015) (ST2TT0(K),K=I,KS) OVLL 237
1015 FORMAT(1X,13HTTBAR2/TIBAR02X,F10.5,3X,F10.5,3X,F10.5,3X,F10.5) *****
      WRITE(6,1016) (STRTT0(K),K=I,KS) OVLL 239
1016 FORMAT(2X,12HTTH14/TTBAR02X,F10.5,3X,F10.5,3X,F10.5,3X,F10.5) *****
      WRITE(6,2003) (WGT1(K),K=I,KS) *****
2003 FORMAT(2X,12H      WG 12X,F10.3,3X,F10.3,3X,F10.3,3X,F10.3) *****
      WRITE(6,1017) (PS1A(IP,K),K=I,KS) *****
1017 FORMAT(2X,12H      PS 1A2X,F10.3,3X,F10.3,3X,F10.3,3X,F10.3) *****
      WRITE(6,1018) (TTR1A(IF,K),K=I,KS) OVLL 243
1018 FORMAT(2X,12H      TTH 1A2X,F10.1,3X,F10.1,3X,F10.1,3X,F10.1) *****
      WRITE(6,1019) (PTR1A(IF,K),K=I,KS) OVLL 245
1019 FORMAT(2X,12H      PTH 1A2X,F10.3,3X,F10.3,3X,F10.3,3X,F10.3) *****
      WRITE(6,3003) (WGT1A(K),K=I,KS) *****
3003 FORMAT(2X,12H      WG 1A2X,F10.3,3X,F10.3,3X,F10.3,3X,F10.3) *****
      WRITE(6,1020) (PS2(IP,K),K=I,KS) OVLL 247
1020 FORMAT(2X,12H      PS 2X,F10.3,3X,F10.3,3X,F10.3,3X,F10.3) *****
      WRITE(6,1021) (TTBAR(K),K=I,KS) OVLL 249
1021 FORMAT(2X,12H      TTBAR 22X,F10.1,3X,F10.1,3X,F10.1,3X,F10.1) *****
      WRITE(6,1022) (PTBAR(K),K=I,KS) OVLL 251
1022 FORMAT(2X,12H      PTHAR 22X,F10.3,3X,F10.3,3X,F10.3,3X,F10.3) *****
      WRITE(6,4003) (WGT2(K),K=I,KS) *****
4003 FORMAT(2X,12H      WG 22X,F10.3,3X,F10.3,3X,F10.3,3X,F10.3) *****
      WRITE(6,5003) (WGT2A(K),K=I,KS) *****
5003 FORMAT(2X,12H      WG 2A2X,F10.3,3X,F10.3,3X,F10.3,3X,F10.3) *****
      WRITE(6,1023) (UPVIS(K),K=I,KS) OVLL 253
1023 FORMAT(2X,12H      UP/V12X,F10.5,3X,F10.5,3X,F10.5,3X,F10.5) *****
      WRITE(6,1024) (URVIS(K),K=I,KS) OVLL 255
1024 FORMAT(2X,12H      UP/V12X,F10.5,3X,F10.5,3X,F10.5,3X,F10.5) *****
      WRITE(6,1025) (PSIPS(K),K=I,KS) OVLL 257
1025 FORMAT(2X,12H      PSI P2X,F10.5,3X,F10.5,3X,F10.5,3X,F10.5) *****
      WRITE(6,1026) (PSIRS(K),K=I,KS) OVLL 259
1026 FORMAT(2X,12H      PSI R2X,F10.5,3X,F10.5,3X,F10.5,3X,F10.5) *****
      WRITE(6,1027) (RXP(K),K=I,KS) OVLL 261
1027 FORMAT(2X,12H      RX P2X,F10.5,3X,F10.5,3X,F10.5,3X,F10.5) *****
      WRITE(6,1028) (RXR(K),K=I,KS) OVLL 263
1028 FORMAT(2X,12H      RX R2X,F10.5,3X,F10.5,3X,F10.5,3X,F10.5) *****
      WRITE(6,1029) (SA0(K),K=I,KS) OVLL 265
1029 FORMAT(2X,12H      ALPHA 02X,F10.3,3X,F10.3,3X,F10.3,3X,F10.3) *****

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Listing of Code (continued)

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WRITE(6,1030) (SIS(K),K=I,KS)
1030 FORMAT(2X,12H      I STATOR2X,F10.3,3X,F10.3,3X,F10.3,3X,F10.3)
WRITE(6,1031) (SB1A(K),K=I,KS)
1031 FORMAT(2X,12H      BETA 1A2X,F10.3,3X,F10.3,3X,F10.3,3X,F10.3)
WRITE(6,1032) (SIR(K),K=I,KS)
1032 FORMAT(2X,12H      I RCTOR2X,F10.3,3X,F10.3,3X,F10.3,3X,F10.3)
WRITE(6,1033) (SA2(K),K=I,KS)
1033 FORMAT(2X,12H      ALPHA 2A2X,F10.3,3X,F10.3,3X,F10.3,3X,F10.3)
WRITE(6,1034) (DBETAR(K),K=I,KS)
1034 FORMAT(2X,12H      DBETA R2X,F10.3,3X,F10.3,3X,F10.3,3X,F10.3)
WRITE(6,1035) (MIS(K),K=I,KS)
1035 FORMAT(2X,12H      M 12X,F10.5,3X,F10.5,3X,F10.5,3X,F10.5)
WRITE(6,1036) (MIRS(K),K=I,KS)
1036 FORMAT(2X,12H      M1 RT2X,F10.5,3X,F10.5,3X,F10.5,3X,F10.5)
WRITE(6,1037) (MR1A(IP,K),K=I,KS)
1037 FORMAT(2X,12H      M1 1A2X,F10.5,3X,F10.5,3X,F10.5,3X,F10.5)
WRITE(6,1038) (MR1AR(K),K=I,KS)
1038 FORMAT(2X,12H      MR1A RT2X,F10.5,3X,F10.5,3X,F10.5,3X,F10.5)
WRITE(6,1039) (MR2(IP,K),K=I,KS)
1039 FORMAT(2X,12H      MR 22X,F10.5,3X,F10.5,3X,F10.5,3X,F10.5)
WRITE(6,1040) (MR2T(K),K=I,KS)
1040 FORMAT(2X,12H      MR2 TIP2X,F10.5,3X,F10.5,3X,F10.5,3X,F10.5)
WRITE(6,1041) (SETHC(K),K=I,KS)
1041 FORMAT(2X,12H      E/TH CR2X,F10.3,3X,F10.3,3X,F10.3,3X,F10.3)
WRITE(6,1042) (SNRTHC(K),K=I,KS)
1042 FORMAT(2X,12H      N/RTH CR2X,F10.3,3X,F10.3,3X,F10.3,3X,F10.3)
WRITE(6,1043) (SWRTED(K),K=I,KS)
1043 FORMAT(2X,12H      WRTHCE/D2X,F10.3,3X,F10.3,3X,F10.3,3X,F10.3)
IF (KSTG-KS)22,22,21
21 WRITE(6,1045)NAME,TITLE,ICASE,ISCASE
1045 FORMAT(1H1,21X,29HNASA TURBINE COMPUTER PROGRAM /6X,10A6/
1 6X,10A6/ 30X,6HCASE 13,1H.,13/28X,17HSTAGE PERFORMANCE /19X
27HSTAGE 5,6X,7HSTAGE 6,6X,7HSTAGE 7,6X,7HSTAGE 8/ )
I=5
KS=KSTG
GO TO 20
22 WRITE(6,1044)OPSIP,OPSIH,ODELHT,OWRTP,CNRT,ODHT,OPT0T2,
1OPT0S2,OPTAT2,OETATT,CETATS,OETAAT,OWNED,ONRTHC,OETHC
1044 FORMAT(/31X,19HOVERALL PERFORMANCE/7X,9HPSI P
1F10.5, 5X,10HPSI R F10.5, 5X9HDEL T F10.5/7X,9HWRT/P
2F10.5, 5X,10HN/RT F10.5, 5X9HDELH/TTINF10.5/7X,10HPT0/PTBAR2
3F9.5, 5X,10HPT0/PS2 F10.5, 5X9HPT0/PAT2AF10.5/7X,9HETA TT
4F10.5, 5X,10HETA TS F10.5, 5X9HETA TAT F10.5/7X,9HWNE/60D
5F10.3, 5X,10HN/RTH CR F10.3, 5X,9HE/TH CR F10.5/)
IF(SRFLAG) WRITE(6,20000)
20000 FORMAT(1H1,45H AN EXIT HAS BEEN MADE FROM SUBROUTINE OVRALL)
RETURN

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Listing of Code (continued)

END

OVLL 312

Listing of Code (continued)

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      SUBROUTINE DIAGT(M)
CDIAGT
C
      REAL MFSTOP
      LOGICAL PREVER,SRFLAG
      COMMON SRFLAG
      COMMON /SNTCP/G,AJ,PRFC,ICASE,PREVER,MFSTOP,JUMP,LOPIN,ISCASE,
1KN,GAMF,IP,SCHIT,PTRN,ISECT,KSTG,WITOL,RHOTOL,PTOL,TRLOOP,LSTG,
2LBHC,IRHC,ICHOKE,ISORH,CHUKE,PTOP1(6,B),PTRS2(6,B),TRDIAG,SC,RC,
3DELPH,PASS,IPC,LUPC,ISS
C
      COMMON /SINIT/H1(6,B),H2(6,B),DP0(6,B),UP1(6,B),DP1A(6,B),DP2(6,B)
1,DP2A(6,B),CSALF1(6,B),ALF1(6,B),CSHET2(6,B),BET2(6,B),RADSD(6,B),
2RADHD(6,B),ANN1(6,B),ANN2(6,B),ANN2A(6,B),ANN1A(6,B),U1A(6,B),
3U2(6,B),ANN0(6,B),PT0(6,B),TT0(6,B),ALPHA0(6,B),PTP(6,B)
C
      COMMON /SINPUT/ HSL,TSL,PSL,GAMSI,
1PTPS,PTIN,TTIN,WAIR,FAIR,DELC,DEL L,DELA,AACS,VCTD,STG,SECT,EXPN,
2EXPP,EXPRE,   RPM,PAF,SLT,STGCH,FNDJUH,NAME(10),TITLE(10),PCNH(4),
3RV(6,B),GAM(6,B),DR(6,B),UT(6,B),RWG(6,B),ALPHAS(6,B),ALPHA1(6,B),
4ETARS(6,B),ETAS(6,B),CFS(6,B),ANNO(6,B),BETA1(6,B),BETA2(6,B),ETARDIGT
5R(6,B),ETAR(6,B),CFR(6,B),TFR(6,B),ANDUH(6,B),OMEGAS(6,B),AS0(6,B)
6,ASVP0(6,B),ACMNO(6,B),A1(6,B),A2(6,B),A3(6,B),A4(6,B),A5(6,B),A6(
76,B),OMEGAR(6,B),BSIA(6,B),BSMPIA(6,B),HCMNIA(6,B),R1(6,B),R2(6,B)
8,R3(6,B),R4(6,B),R5(6,B),R6(6,B),SESTHI(8),HERTHI(8)
C
      REAL M0
      COMMON /SSTA01/CP0(8),
1R),VU0(6,B),VZ0(6,B),RHOS0(6,B),PS1(6,B),WGT1(8),TA1(8),WG1(6,B),
2DPDH1(6,B),SI(6,B),   CP1(8),PHI1(6,B),TS1(6,B),V1(6,B)
3,RHOS1(6,B),ALF1E(6,B),VU1(6,B),VZ1(6,B),M0(6,B),WGT0(8),WG0(6,B)
C
      REAL MR1A
      COMMON /SSTA1A/VU1A(6,B),WG1A(6,B),WGT1A(8),VZ1A(6,B),   CP1A(8),
1PS1A(6,B),RU1A(6,B),R1A(6,B),RET1A(6,B),RI(6,B),TTR1A(6,B),PTR1A(6
2,B),MR1A(6,B),TS1A(6,B)
C
      COMMON /SSTA2/V2(6,B),TTR2(6,B),PTR2(6,B),WG2(6,B),WGT2(8),TA2(8)
1PS2(6,B),PHI2(6,B)
C
      REAL MR2,M2   ,MF2
      COMMON /SFLW2/TS2(6,B),CP2(8),R2(6,B),RHOS2(6,B),BET2E(6,B),RU2(6
1,B),VU2(6,B),DPDH2(6,B),VZ2(6,B),MR2(6,B),MF2(6,B),M2(6,B)
C
      REAL M2A,MF2A
      COMMON /SSTA2A/WG2A(6,B),WGT2A(8),VU2A(6,B),VZ2A(6,B),PS2A(6,B),

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Listing of Code (continued)

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1ALF2A(6,8),TT2A(6,8),FT2A(6,8),TTBAR(8),PTBAR(8),STT0(8),SPT0(8), DIGT n46
2M2A(6,8),MF2A(6,8),CP2A(8),V2A(6,8),TS2A(6,8),TAS(8),PAS(8),GAMS(8)DIGT n47
3),CPS(8),DELHVD(6,8),HVBAR(8) *****
C DIGT n49
IF(SRFLAG) WRITE(6,10000) *****
10000 FORMAT(44H AN ENTRY HAS BEEN MADE IN SUBROUTINE DIAGT ) *****
WRITE(6,1000)NAME,TITLE DIGT n50
1000 FORMAT(1H1,5X,10A6/6X,10A6/20X,29HNASA TURBINE COMPUTER PROGRAM/ DIGT n51
131X,10HDIAGNOSTIC) DIGT n52
IF (M.EQ.0) GO TO 10 DIGT n53
GO TO (10,19,11,12,13),M DIGT n54
10 DO 14 K=1,KN DIGT n55
WRITE(6,1001)K,CP0(K),GAM(1,K) DIGT n56
1001 FORMAT(9X,1HK,15,9X,3F10.3,9X,5HGAMMA,F10.5) DIGT n57
WRITE(6,1002) (PTP(I,K),I=1,ISECT) DIGT n58
1002 FORMAT(3X,6H PTP,6F10.3) DIGT n59
WRITE(6,1003) (PT0(I,K),I=1,ISECT) DIGT n60
1003 FORMAT(3X,6H PT0,6F10.3) DIGT n61
WRITE(6,1004) (PS0(I,K),I=1,ISECT) DIGT n62
1004 FORMAT(3X,6H PS0,6F10.3) DIGT n63
WRITE(6,1005) (TT0(I,K),I=1,ISECT) DIGT n64
1005 FORMAT(3X,6H TT0,6F10.1) DIGT n65
WRITE(6,1006) (TS0(I,K),I=1,ISECT) DIGT n66
1006 FORMAT(3X,6H TS0,6F10.1) DIGT n67
WRITE(6,1007) (V0(I,K),I=1,ISECT) DIGT n68
1007 FORMAT(3X,6H V0,6F10.3) DIGT n69
WRITE(6,1008) (ALPHA0(I,K),I=1,ISECT) DIGT n70
1008 FORMAT(3X,6HALPHA0,6F10.3) DIGT n71
14 WRITE(6,1009) (SI(I,K),I=1,ISECT) DIGT n72
IF (M.EQ.0) GO TO 19 DIGT n73
GO TO 18 DIGT n74
19 DO 20 K=1,KN DIGT n75
1009 FORMAT(3X,6H SI,6F10.3) DIGT n76
WRITE(6,1010) K,CP1(K),GAM(2,K) DIGT n77
1010 FORMAT(9X,1HK,15,9X,3F10.3,9X,5HGAMMA,F10.5) DIGT n78
WRITE(6,1011) (PS1(I,K),I=1,ISECT) DIGT n79
1011 FORMAT(3X,6H PS1,6F10.3) DIGT n80
WRITE(6,1012) (OPDR1(I,K),I=1,ISECT) DIGT n81
1012 FORMAT(3X,6H OPDR1,6F10.5) DIGT n82
WRITE(6,1013) (TS1(I,K),I=1,ISECT) DIGT n83
1013 FORMAT(3X,6H TS1,6F10.1) DIGT n84
WRITE(6,1014) (WG1(I,K),I=1,ISECT) DIGT n85
1014 FORMAT(3X,6H WG1,6F10.3) DIGT n86
WRITE(6,1015) (V1(I,K),I=1,ISECT) DIGT n87
1015 FORMAT(3X,6H V1,6F10.3) DIGT n88
WRITE(6,1016) (ALF1E(I,K),I=1,ISECT) DIGT n89
1016 FORMAT(3X,6H ALF1E,6F10.3) DIGT n90

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Listing of Code (continued)

20 WRITE(6,1017) (ALF1(I,K),I=1,ISECT)	DIGT 091
1017 FORMAT(3X,6H ALF1,6F10.3)	DIGT 092
IF (M.EQ.0) GO TO 11	DIGT 093
GO TO 18	DIGT 094
11 DO 15 K=1,KN	DIGT 095
WRITE(6,1018) K,CP1A(K),GAM(3,K)	DIGT 096
1018 FORMAT(9X,1H K,15,9X,4H CP1A,F10.3,8X,5H GAMMA,F10.5)	DIGT 097
WRITE(6,1019) (PTR1A(I,K),I=1,ISECT)	DIGT 098
1019 FORMAT(3X,6H PTR1A,6F10.3)	DIGT 099
WRITE(6,1020) (PS1A(I,K),I=1,ISECT)	DIGT 100
1020 FORMAT(3X,6H PS1A,6F10.3)	DIGT 101
WRITE(6,1021) (ITR1A(I,K),I=1,ISECT)	DIGT 102
1021 FORMAT(3X,6H ITR1A,6F10.1)	DIGT 103
WRITE(6,1022) (WG1A(I,K),I=1,ISECT)	DIGT 104
1022 FORMAT(3X,6H WG1A,6F10.3)	DIGT 105
WRITE(6,1023) (R1A(I,K),I=1,ISECT)	DIGT 106
1023 FORMAT(3X,6H R1A,6F10.3)	DIGT 107
WRITE(6,1024) (HET1A(I,K),I=1,ISECT)	DIGT 108
1024 FORMAT(3X,6H HET1A,6F10.3)	DIGT 109
15 WRITE(6,1025) (R1(I,K),I=1,ISECT)	DIGT 110
1025 FORMAT(3X,6H R1,6F10.3)	DIGT 111
IF (M.EQ.0) GO TO 12	DIGT 112
GO TO 18	DIGT 113
12 DO 16 K=1,KN	DIGT 114
WRITE(6,1026) K,CP2(K),GAM(3,K)	DIGT 115
1026 FORMAT(9X,1H K,15,9X,3H CP2,F10.3,9X,5H GAMMA,F10.5)	DIGT 116
WRITE(6,1027) (PTR2(I,K),I=1,ISECT)	DIGT 117
1027 FORMAT(3X,6H PTR2,6F10.3)	DIGT 118
WRITE(6,1028) (PS2(I,K),I=1,ISECT)	DIGT 119
1028 FORMAT(3X,6H PS2,6F10.3)	DIGT 120
WRITE(6,1029) (DPR2(I,K),I=1,ISECT)	DIGT 121
1029 FORMAT(3X,6H DPR2,6F10.5)	DIGT 122
WRITE(6,1030) (ITR2(I,K),I=1,ISECT)	DIGT 123
1030 FORMAT(3X,6H ITR2,6F10.1)	DIGT 124
WRITE(6,1031) (TS2(I,K),I=1,ISECT)	DIGT 125
1031 FORMAT(3X,6H TS2,6F10.1)	DIGT 126
WRITE(6,1032) (WG2(I,K),I=1,ISECT)	DIGT 127
1032 FORMAT(3X,6H WG2,6F10.3)	DIGT 128
WRITE(6,1033) (R2(I,K),I=1,ISECT)	DIGT 129
1033 FORMAT(3X,6H R2,6F10.3)	DIGT 130
WRITE(6,1034) (HET2E(I,K),I=1,ISECT)	DIGT 131
1034 FORMAT(3X,6H HET2E,6F10.3)	DIGT 132
16 WRITE(6,1035) (HET2(I,K),I=1,ISECT)	DIGT 133
1035 FORMAT(3X,6H HET2,6F10.3)	DIGT 134
IF (M.EQ.0) GO TO 13	DIGT 135
GO TO 18	DIGT 136
13 DO 17 K=1,KN	DIGT 137

Listing of Code (continued)

L=K +1	DIGT 138
WRITE(6,1036)K,CP2A(K),GAM(5,K)	DIGT 139
1036 FORMAT(9X,1HK,15,9X,4F,CP2A,F10.3,9X,5HGAMMA,F10.5)	DIGT 140
WRITE(6,1037) (PT2A(I,K),I=1,ISECT)	DIGT 141
1037 FORMAT (3X,6H PT2A,6F10.3)	DIGT 142
WRITE(6,1038) (PS2A(I,K),I=1,ISECT)	DIGT 143
1038 FORMAT (3X,6H PS2A,6F10.3)	DIGT 144
WRITE(6,1039) (TT2A(I,K),I=1,ISECT)	DIGT 145
1039 FORMAT (3X,6H TT2A,6F10.1)	DIGT 146
WRITE(6,1040) (TS2A(I,K),I=1,ISECT)	DIGT 147
1040 FORMAT (3X,6H TS2A,6F10.1)	DIGT 148
WRITE(6,1041) (WG2A(I,K),I=1,ISECT)	DIGT 149
1041 FORMAT (3X,6H WG2A,6F10.3)	DIGT 150
WRITE(6,1042) (V2A(I,K),I=1,ISECT)	DIGT 151
1042 FORMAT (3X,6H V2A,6F10.3)	DIGT 152
WRITE(6,1043) (ALF2A(I,K),I=1,ISECT)	DIGT 153
1043 FORMAT (3X,6H ALF2A,6F10.3)	DIGT 154
WRITE(6,1044) (SI(I,K),I=1,ISECT)	DIGT 155
1044 FORMAT (3X,6H SI,6F10.3)	DIGT 156
WRITE(6,1045) L,CPS(K),GAMS(K)	DIGT 157
1045 FORMAT(9X,1HL,15,9X,3F,CPS,F10.3,9X,5HGAMMA,F10.5)	DIGT 158
WRITE(6,1046) (PTP(I,L),I=1,ISECT)	DIGT 159
1046 FORMAT (3X,6H PTP,6F10.3)	DIGT 160
WRITE(6,1047) (PT0(I,L),I=1,ISECT)	DIGT 161
1047 FORMAT (3X,6H PT0,6F10.3)	DIGT 162
17 WRITE(6,1048) (TT0(I,L),I=1,ISECT)	DIGT 163
1048 FORMAT (3X,6H TT0,6F10.1)	DIGT 164
18 CONTINUE	DIGT 165
IF(SHFLAG) WRITE(6,20000)	*****
20000 FORMAT(1H1,45H AN EXIT HAS BEEN MADE FROM SUBROUTINE DIAGT)	*****
RETURN	DIGT 166
END	DIGT 167

Listing of Code (continued)

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SUBROUTINE INSTG
CINSTG
C INTERSTAGE OUTPUT
C NUMBER OF SECTORS IS THREE OR LESS, HUB AND CASING VALUES ARE
C CALCULATED AND PRINTED
C NUMBER OF SECTORS IS MORE THAN THREE, ONLY SECTOR PITCHLINE
C VALUES ARE PRINTED
C
REAL MFSTOP
LOGICAL PREVER, SRFLAG
COMMON SRFLAG
COMMON /SNTCP/G, AJ, PRPC, ICASE, PREVER, MFSTOP, JUMP, LOPIN, ISCASE,
1KN, GAMF, IP, SCRIT, PTRN, ISECT, KSTG, WTOL, RHOTOL, PRTOL, TRLOOP, LSTG,
2LBRC, IBRC, ICHOKE, ISORH, CHOK, PTOP1(6,8), PTRS2(6,8), TRDIAG, SC, RC,
3DELPR, PASS, IPC, LOPC, ISS
C
COMMON /SINIT/H1(6,8), H2(6,8), DP0(6,8), DP1(6,8), DP1A(6,8), DP2(6,8),
1, DP2A(6,8), CSALF1(6,8), ALF1(6,8), CSDET2(6,8), BET2(6,8), RADSD(6,8),
2RADRD(6,8), ANN1(6,8), ANN2(6,8), ANN2A(6,8), ANN1A(6,8), U1A(6,8),
3U2(6,8), ANNO(6,8), PT0(6,8), TT0(6,8), ALPHA0(6,8), PTP(6,8)
C
COMMON /SINPUT/ RSL, TSL, PSL, GAMS1,
1PTPS, PTIN, TTIN, WAIR, FAIR, DELC, DEL L, DELA, AAC, VCTD, STG, SECT, EXPN,
2EXPP, EXPRE, RPM, PAF, SLI, STGCH, FNDOJ, NAME(10), TITLE(10), PCNH(6),
3RV(6,8), GAM(6,8), DR(6,8), DT(6,8), RWG(6,8), ALPHAS(6,8), ALPHA1(6,8),
4ETARS(6,8), ETAS(6,8), CFS(6,8), ANNO(6,8), BETA1(6,8), BETA2(6,8), ETARINST
5R(6,8), ETAR(6,8), CFR(6,8), TFR(6,8), ANDOR(6,8), OMEGAS(6,8), ASO(6,8),
6, ASMP0(6,8), ACMN0(6,8), A1(6,8), A2(6,8), A3(6,8), A4(6,8), A5(6,8), A6(
7, 6,8), OMEGAR(6,8), HSIA(6,8), ASMPIA(6,8), HCMNIA(6,8), B1(6,8), B2(6,8),
8, B3(6,8), B4(6,8), B5(6,8), B6(6,8), SESTHI(8), RERTHI(8)
C
REAL M0
COMMON /SSTA01/CP0(8),
18), VU0(6,8), VZ0(6,8), RHOS0(6,8), PS1(6,8), WGT1(8), TA1(8), WG1(6,8),
2 DPOR1(6,8), SI(6,8), CP1(8), PHI1(6,8), TS1(6,8), V1(6,8),
3, RHOS1(6,8), ALF1E(6,8), VU1(6,8), VZ1(6,8), M0(6,8), WGT0(8), WG0(6,8)
C
REAL MR1A
COMMON /SSTA1A/VU1A(6,8), WG1A(6,8), WGT1A(8), VZ1A(6,8),
1PS1A(6,8), RU1A(6,8), R1A(6,8), HET1A(6,8), RI(6,8), TTR1A(6,8), PTR1A(6
2, 8), MR1A(6,8), TS1A(6,8)
C
COMMON /SSTA2/V2(6,8), TTR2(6,8), PTR2(6,8), WG2(6,8), WGT2(8), TA2(8),
1 PS2(6,8), PHI2(6,8)
C
REAL MR2, M2, MF2

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Listing of Code (continued)

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COMMON /SFLOW2/TS2(6,8),CP2(8),R2(6,8),RHOS2(6,8),BET2E(6,8),RU2(6,8) INST 046
1,8),VU2(6,8),DPDR2(6,8),VZ2(6,8),MR2(6,8),MF2(6,8),M2(6,8) INST 047
C INST 048
REAL M2A,MF2A INST 049
COMMON /SSTA2A/WG2A(6,8),WGT2A(8),VU2A(6,8),VZ2A(6,8),PS2A(6,8), INST 050
1ALF2A(6,8),TT2A(6,8),PT2A(6,8),TTBAR(8),PTBAR(8),STT0(8),SPT0(8), INST 051
2M2A(6,8),MF2A(6,8),CP2A(8),V2A(6,8),TS2A(6,8),TAS(8),PAS(8),GAMS(8) INST 052
3),CPO(8),DELHVD(6,8),HVBAR(8) *****
C INST 054
COMMON /SOVRAL/DELHT(6,8),DELHTI(6,8),DELHSI(6,8),DEHATI(6,8), INST 055
1ETATT(6,8),ETATS(6,8),ETATAT(6,8) INST 056
C INST 057
COMMON STDP0(7),STFT0(7),STALF(7),STSI(7),STV0(7),STVU0(7), INST 058
1STVZ0(7),STTS0(7),STPS0(7),STDEN0(7),STM0(7),STDP1(7),STALFE(7), INST 059
2STDELA(7),STV1(7),STVU1(7),STVZ1(7),STTS1(7),STPS1(7),STDEN1(7), INST 060
3STM1(7),ZWINC(7), CPS(7),STDP1A(7), INST 061
4STPTR1(7),STBET1(7),STRI(7),STR1A(7),STHU1A(7),STMR1A(7),STU1A(7), INST 062
5STDP2(7),STHET2(7),SDBETA(7),SR2(7),SRU2(7),SMR2(7),SU2(7),RX(7), INST 063
6STDELH(7),STPS1(7),SETATT(7),SETATS(7),SETAAT(7),RZWINC(7), INST 064
7 CFR(7),STPT2A(7),STTT2A(7),STV2A(7),STVU2A(7), *****
8STALF2(7),STMF2A(7),STTTR1(7),STVZ2A(7),STTS2A(7),STPS2A(7),STDEN2 *****
9(7),STM2A(7),STT0(7),LJ,JJ,K,STWG0(7),STWG1(7),STWG1A(7),STWG2(7) *****
9,STWG2A(7),SFLO0,SFLO1,SFLO1A,SFLO2,SFLO2A,STPS1A(7),STTS1A(7), *****
9STPTR2(7),STTTR2(7),STPS2(7),STTS2(7) *****
C INST 068
C INST 069
IF(SRFLAG) WRITE(6,10000) *****
10000 FORMAT(44H AN ENTRY HAS BEEN MADE IN SUBROUTINE INSTG ) *****
1 DO 4 K=1,KSTG *****
SFL00 =0.0 *****
SFL01 =0.0 *****
SFL01A=0.0 *****
SFL02 =0.0 *****
SFL02A=0.0 *****
E1=(GAMS(K)-1.)/GAMS(K) INST 071
E2=GAM(1,K)/(GAM(1,K)-1.) INST 072
E3=GAM(2,K)/(GAM(2,K)-1.) INST 073
E4=GAM(3,K)/(GAM(3,K)-1.) INST 074
E5=GAM(4,K)/(GAM(4,K)-1.) INST 075
E6=GAM(5,K)/(GAM(5,K)-1.) INST 076
C RELOCATE PITCHLINE VALUES INST 077
J=ISECT+1 INST 078
4 DO 5 I=1,ISECT INST 079
KS=J-I+1 INST 080
STWG0(KS)=WG0(KS-1,K) *****
SFL00 =SFL00 +STWG0(KS) *****
STT0(KS)=TT0(KS-1,K) INST 081

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Listing of Code (continued)

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STOP0(KS)=NP0(KS-1,K)
STPT0(KS)=PTP(KS-1,K)
STALF(KS)=ALPHA0(KS-1,K)*57.2958
STSI(KS)=SI(KS-1,K)*57.2958
STV0(KS)=V0(KS-1,K)
STVL0(KS)=VU0(KS-1,K)
STVZ0(KS)=VZ0(KS-1,K)
STTS0(KS)=TS0(KS-1,K)
STPS0(KS)=PS0(KS-1,K)
STDEN0(KS)=144.*STPS0(KS)/(STTS0(KS)*RV(1,K))
STM0(KS)=M0(KS-1,K)
STWG1(KS)=WG1(KS-1,K)
SFLO1 =SFLO1 +STWG1(KS)
STDP1(KS)=DP1(KS-1,K)
STALFE(KS)=ALF1E(KS-1,K)*57.2958
STDFLA(KS)=(ALPHA0(KS-1,K)+ALF1E(KS-1,K))*57.2958
STV1(KS)=V1(KS-1,K)
STVU1(KS)=VU1(KS-1,K)
STVZ1(KS)=VZ1(KS-1,K)
STTS1(KS)=TS1(KS-1,K)
STPS1(KS)=PS1(KS-1,K)
STDEN1(KS)=HHOS1(KS-1,K)
STM1(KS)=V1(KS-1,K)/(SQRT(GAM(2,K)*G*RV(2,K)*TS1(KS-1,K)))
ZS =-2.*ALF1E(KS-1,K) -1.570796
ZINC(KS)=COS(ZS)*(SIN(ALPHA0(KS-1,K))*COS(ALF1E(KS-1,K))
1-1,K)/((COS(ALPHA0(KS-1,K))*SIN(ALF1E(KS-1,K)))+1.)
CPS(KS)=1.-(STVU(KS)/STV1(KS))*2
STWG1A(KS)=WG1A(KS-1,K)
SFLO1A=SFL01A+STWG1A(KS)
STDP1A(KS)=DP1A(KS-1,K)
STPTR1(KS)=PTR1A(KS-1,K)
STTTR1(KS)=TTR1A(KS-1,K)
STBET1(KS)=HET1A(KS-1,K)*57.2958
STR1(KS)=RI(KS-1,K)*57.2958
STR1A(KS)=R1A(KS-1,K)
STRU1A(KS)=PU1A(KS-1,K)
STMW1A(KS)=MW1A(KS-1,K)
STU1A(KS)=U1A(KS-1,K)
STPS1A(KS)=PS1A(KS-1,K)
STTS1A(KS)=TS1A(KS-1,K)
STWG2(KS)=WG2(KS-1,K)
SFLO2 =SFLO2 +STWG2(KS)
STDP2(KS)=DP2(KS-1,K)
STHET2(KS)=HET2E(KS-1,K)*57.2958
SDHFTA(KS)=(HET1A(KS-1,K)+HET2E(KS-1,K))*57.2958
SR2(KS)=R2(KS-1,K)
SHU2(KS)=RU2(KS-1,K)

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INST 042
INST 043
INST 084
INST 095
INST 086
INST 087
INST 088
INST 099
INST 090
*****
INST 092
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INST 093
INST 094
INST 095
INST 096
INST 097
INST 098
INST 099
INST 100
INST 101
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INST 103
INST 104
INST 105
INST 106
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INST 107
INST 108
INST 109
INST 110
INST 111
INST 112
INST 113
INST 114
INST 115
*****
*****
*****
INST 116
INST 117
INST 118
INST 119
INST 120

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Listing of Code (continued)

SU2(KS)=U2(KS-1,K)	INST 121
STPTR2(KS)=PTR2(KS-1,K)	INST 122
STTH2(KS)=TH2(KS-1,K)	*****
RX(KS)=1.-(1.-(PS1(KS-1,K)/PTP(KS-1,K))*E1)/(1.-(PS2(KS-1,K)/	INST 123
1PTP(KS-1,K))*E1)	INST 124
STDELH(KS)=DELHT(KS-1,K)	INST 125
STPSI(KS)=2.*G*AJ*DELFT(KS-1,K)/(U1A(KS-1,K)*U1A(KS-1,K)	INST 126
1+U2(KS-1,K)*U2(KS-1,K))	INST 127
SETATT(KS)=ETATT(KS-1,K)	INST 128
SETATS(KS)=ETATS(KS-1,K)	INST 129
SETAAT(KS)=ETATAT(KS-1,K)	INST 130
ZR = -2.*HET2E(KS-1,K) -1.570796	INST 131
RZWINC(KS)=COS(ZR)*(SIN(HET1A(KS-1,K))*COS(HET2E(KS-	INST 132
11,K))/COS(HET1A(KS-1,K))*SIN(HET2E(KS-1,K)))+1.)	INST 133
CPR(KS)=1.-(STR1A(KS)/SH2(KS))*2	INST 134
STPS2(KS)=PS2(KS-1,K)	*****
STTS2(KS)=TS2(KS-1,K)	*****
STWG2A(KS)=WG2A(KS-1,K)	*****
SFL02A=SFL02A+STWG2A(KS)	*****
STPT2A(KS)=PT2A(KS-1,K)	*****
STTT2A(KS)=TT2A(KS-1,K)	*****
STV2A(KS)=V2A(KS-1,K)	*****
STVU2A(KS)=VU2A(KS-1,K)	*****
STALF2(KS)=ALF2A(KS-1,K)*57.2958	INST 139
STMF2A(KS)=MF2A(KS-1,K)	*****
STVZ2A(KS)=VZ2A(KS-1,K)	*****
STPS2A(KS)=PS2A(KS-1,K)	*****
STTS2A(KS)=TS2A(KS-1,K)	*****
STM2A(KS)=M2A(KS-1,K)	*****
STDEN2(KS)=144.*STPS2A(KS)/(STTS2A(KS)*RV(5,K))	*****
5 CONTINUE	INST 146
IF (ISECT-3)3,3,6	INST 147
C CALCULATE HUB VALUES	INST 148
3 LJ=1	INST 149
JJ=ISECT+2	INST 150
I=1	INST 151
L=1	INST 152
STDP0(L)=DR(1,K)	*****
R1=DP0(I,K)/DR(1,K)	*****
STDP1(L)=DR(2,K)	*****
R2=DP1(I,K)/DR(2,K)	INST 156
STDP1A(L)=DR(3,K)	*****
R3=DP1A(I,K)/DR(3,K)	INST 158
STDP2(L)=DR(4,K)	INST 159
R4=DP2(I,K)/DR(4,K)	*****
TALF=SIN(ALF1(I,K))*R3/COS(ALF1(I,K))	INST 161

Listing of Code (continued)

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      R5=CP2A(I,K)/DR(5,K)
C      STATION 0          STATOR INLET
10 STT0(L)=TT0(I,K)
   STP0(L)=PTP(I,K)
   STVZ0(L)=VZ0(I,K)
   STVU0(L)=VU0(I,K)*R1
   STV0(L)=SQRT(VZ0(I,K)*VZ0(I,K)+STVU0(L)*STVU0(L))
   STTS0(L)=TT0(I,K)-STV0(L)*STV0(L)/(2.*G*AJ*CP0(K))
   STPS0(L)=PS0(I,K)*(STTS0(L)/TS0(I,K))*E2
   STDEN0(L)=144.*STPS0(L)/(RV(1,K)*STTS0(L))
   STALF(L)=ATAN2(STVU0(L),STVZ0(L))*57.2958
   STSI(L)=STALF(L)-ATAN2(SIN(RADSD(I,K))*R1,COS(RADSD(I,K)))
1 *57.2958
   AS0H=SQRT(GAM(1,K)*G*HV(1,K)*STTS0(L))
   STM0(L)=STV0(L)/AS0H
C      STATION 1          STATOR EXIT
   STVZ1(L)=VZ1(I,K)
   STVU1(L)=VU1(I,K)*R2
   STV1(L)=SQRT(VZ1(I,K)*VZ1(I,K)+STVU1(L)*STVU1(L))
   STTS1(L)=TT0(I,K)-STV1(L)*STV1(L)/(2.*G*AJ*CP1(K))
   STPS1(L)=PS1(I,K)*(STTS1(L)/TS1(I,K))*E3
   STDEN1(L)=144.*STPS1(L)/STTS1(L)/RV(2,K)
   STALFE(L)=ATAN2(STVU1(L),STVZ1(L))*57.2958
   STDELA(L)=STALF(L)+STALFE(L)
   AS1H=SQRT(GAM(2,K)*G*HV(2,K)*STTS1(L))
   STM1(L)=STV1(L)/AS1H
   ZS=-2.*STALFE(L)/57.2958-1.570796
   ZWIINC(L)=COS(ZS)*(STVU0(L)*STVZ1(L)/(STVZ0(L)*STVU1(L))+1.)
   CPS(L)=1.-(STV0(L)/STV1(L))*2
C      STATION 1A         ROTOR INLET
   VU1AH=VU1A(I,K)*R3
   STRU1A(L)=VU1AH-U1A(I,K)/R3
   STBET1(L)=ATAN2(STRU1A(L),VZ1A(I,K))*57.2958
   T=TALF-(TALF/R3-SIN(RADWD(I,K))/COS(RADWD(I,K)))/R3
   STRI(L)=STBET1(L)-ATAN2(T,1.)*57.2958
   STR1A(L)=SQRT(STRU1A(L)*STRU1A(L)+VZ1A(I,K)*VZ1A(I,K))
   V1A1AH=VZ1A(I,K)*VZ1A(I,K)+VU1AH*VU1AH
   DELTSH=(V1(I,K)*V1(I,K)-V1A1AH)/(2.*G*AJ*CP1A(K))
   TS1AH=TS1(I,K)+DELTSH
   STTS1A(L)=TS1AH
   STM1A(L)=STR1A(L)/SQRT(GAM(3,K)*G*RV(3,K)*TS1AH)
   TTRSH=1.+STM1A(L)*STM1A(L)*(GAM(3,K)-1.)/2.
   STTTR1(L)=TS1AH*TTRSH
   IF(RI(I,K))2,2,7
2 EXPRI=EXPN
  GO TO 11
7 EXPRI=EXPP

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INST 162
 INST 163
 INST 164
 INST 165
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 INST 202
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 INST 206
 INST 207

Listing of Code (continued)

```

11 PTRSH=(1.+(TTRSH-1.)*ETARR(I,K)*COS(RI(I,K))*EXPRI)**E4      INST 208
PS1AH=PS1(I,K)*(1.+DELTS/TS1(I,K))*E4                             INST 209
STPS1A(L)=PS1AH                                                    *****
STPTR1(L)=PS1AH*PTRSH                                             INST 210
STU1A(L)=U1A(I,K)/R3                                             INST 211
C      STATION      2      ROTOR      EXIT      INST 212
VU2H=VU2(I,K)*R4                                                 INST 213
SHU2(L)=VU2H+U2(I,K)/R4                                          INST 214
STHET2(L)=ATAN2(SHU2(L),VZ2(I,K))*57.2958                       INST 215
SUBETA(L)=STHET1(L)+SIBET2(L)                                    INST 216
SR2(L)=SQRT(SHU2(L)*SHU2(L)+VZ2(I,K)*VZ2(I,K))                 INST 217
V2V2H=VZ2(I,K)*VZ2(I,K)+VU2H*VU2H                               INST 218
DELTSH=(V2(I,K)*V2(I,K)-V2V2H)/(2.*G*AJ*CP2(K))                 INST 219
TS2H=TS2(I,K)+DELTSH                                             INST 220
SITS2(L)=TS2H                                                    *****
SMR2(L)=SR2(L)/SQRT(GAM(4,K)*G*RV(4,K)*TS2H)                   *****
SU2(L)=U2(I,K)/R4                                               INST 222
PS2H=PS2(I,K)*(TS2H/TS2(I,K))*E5                                 INST 223
STPS2(L)=PS2H                                                    *****
RX(L)=1.-(1.-(STPS1(L)/PTP(I,K))*E1)/(1.-(PS2H/PTP(I,K))*E1)    INST 224
STDELH(L)=(STU1A(L)*VL1AH+SU2(L)*VU2H)*TFR(I,K)/(G*AJ)         INST 225
STPSI(L)=2.*G*AJ*STDELH(L)/(STU1A(L)**2+SU2(L)**2)             INST 226
SETATT(L)=STDELH(L)/DELHT1(I,K)                                  INST 227
SETATS(L)=STDELH(L)/DELHS1(I,K)                                  INST 228
SETAAT(L)=STDELH(L)/DEHAT1(I,K)                                  INST 229
ZR=-2.*STHET2(L)/57.2958-1.570796                                INST 230
RZWINC(L)=COS(ZR)*(STU1A(L)*VZ2(I,K)/(VZ1A(I,K)*SRU2(L))+1.)  INST 231
CPR(L)=1.-(STH1A(L)/SH2(L))*2                                     INST 232
STPT2A(L)=PT2A(I,K)                                              *****
STTT2A(L)=TT2A(I,K)                                              *****
STVZ2A(L)=VZ2A(I,K)                                              *****
STVL2A(L)=VL2A(I,K)*R5                                           *****
V2A2AH=STVU2A(L)**2+VZ2A(I,K)**2                                *****
STV2A(L)=SQRT(V2A2AH)                                             *****
STALF2(L)=ATAN2(STVU2A(L),VZ2A(I,K))*57.2958                   *****
DELTS2=(V2A(I,K)**2-V2A2AH)/(2.*G*AJ*CP2A(K))                   INST 240
STTS2A(L)=TS2A(I,K)+DELTS2                                       *****
STPS2A(L)=PS2A(I,K)*(1.+DELTS2/TS2A(I,K))*E6                   *****
STDEN2(L)=144.*STPS2A(L)/(RV(5,K)*STTS2A(L))                   *****
STM2A(L)=STV2A(L)/SQRT(GAM(5,K)*G*RV(5,K)*STTS2A(L))           *****
STMF2A(L)=STM2A(L)*COS(STALF2(L)/57.2958)                       *****
IF (L.GT.1) GO TO 8                                              INST 246
C      CALCULATE TIP VALUES                                     INST 247
I=ISECT                                                            INST 248
L=ISECT+2                                                         INST 249
STDP0(L)=DT(1,K)                                                  INST 250
R1=CP0(I,K)/CT(1,K)                                              INST 251

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Listing of Code (continued)

STDP1(L)=DT(2,K)	INST 252
R2=CP1(I,K)/DT(2,K)	INST 253
STDP1A(L)=DT(3,K)	INST 254
R3=CP1A(I,K)/DT(3,K)	INST 255
STDP2(L)=DT(4,K)	INST 256
R4=CP2(I,K)/DT(4,K)	INST 257
TALF=SIN(ALF1(I,K))*R3/COS(ALF1(I,K))	INST 258
R5=CP2A(I,K)/DT(5,K)	INST 259
GO TO 10	INST 260
6 LJ=2	INST 261
JJ=ISECT+1	INST 262
8 CALL WOUT	INST 263
9 CONTINUE	INST 264
IF(SRFLAG) WRITE(6,20000)	*****
20000 FORMAT(45H AN EXIT HAS BEEN MADE FROM SUBROUTINE INSTG)	*****
RETURN	INST 265
END	INST 266

Listing of Code (continued)

```

1ALF2A(6,8),TT2A(6,8),PT2A(6,8),TTHAR(8),PTHAR(8),STT0(8),SPT0(8), *****
2M2A(6,8),MF2A(6,8),CP2A(8),V2A(6,8),TS2A(6,8),TAS(8),PAS(8),GAMS(8)*****
3)CPO(8),DELMVD(6,8),HVBAR(8) *****
C *****
COMMON /SOVRAL/DELHT(6,8),DELHTI(6,8),DELHSI(6,8),DEHATI(6,8), *****
1ETATT(6,8),ETATS(6,8),ETATAT(6,8) *****
C *****
COMMON STCP0(7),STPT0(7),STALF(7),STSI(7),STV0(7),STVU0(7), WOUT n25
1STVZ0(7),STTS0(7),STPS0(7),STDEN0(7),STM0(7),STDP1(7),STALFE(7), WOUT n26
2STDELA(7),STV1(7),STVL1(7),STVZ1(7),STTS1(7),STPS1(7),STDEN1(7), WOUT n27
3STM1(7),ZWIINC(7), CPS(7),STDP1A(7), WOUT n28
4STPTR1(7),STBET1(7),SIRI(7),STR1A(7),STRU1A(7),STMR1A(7),STU1A(7), WOUT n29
5STDP2(7),STHET2(7),SDHETA(7),SR2(7),SRU2(7),SMR2(7),SU2(7),RX(7), WOUT n30
6STDELH(7),STPSI(7),SEIATT(7),SETATS(7),SETAAT(7),RZWINC(7), WOUT n31
7 CHR(7),STPT2A(7),STTT2A(7),STV2A(7),STVU2A(7),*****
8STALF2(7),STMF2A(7),STTTR1(7),STVZ2A(7),STTS2A(7),STPS2A(7),STDEN2*****
9(7),STM2A(7),STTT0(7),LJ,JJ,K,STWG0(7),STWG1(7),STWG1A(7),STWG2(7)*****
9,STWG2A(7),SFL00,SFL01,SFL01A,SFL02,SFL02A,STPS1A(7),STTS1A(7), *****
9STPTR2(7),STTTR2(7),SIPS2(7),STTS2(7) *****
C WOUT n36
C PRINT OUT FOR INTERSTAGE DATA WOUT n37
IF(SRFLAG) WRITE(6,10000) *****
10000 FORMAT(44H AN ENTRY HAS BEEN MADE IN SUBROUTINE WOUT ) *****
8 WRITE(6,1000)NAME,TITLE,ICASE,ISCASE WOUT n38
1000 FORMAT(1H1,20X29HNASA TURBINE COMPUTER PROGRAM/6X10A6/6X10A6/30X WOUT n39
15HCASE 13,1H,13/24X23HINTER-STAGE PERFORMANCE//) WOUT n40
WRITE(6,1001)K,(STDP0(I),I=LJ,JJ) WOUT n41
1001 FORMAT(5X5HSTA 02X12HSTATOR INLET10X5HSTAGE13,1H,/4X6HDIAM 02X, WOUT n42
16F10.3) WOUT n43
WRITE(6,1002)(STTT0(I),I=LJ,JJ) WOUT n44
1002 FORMAT (10H TT 0,2X,6F10.1) WOUT n45
WRITE(6,1003)( STPT0(I),I=LJ,JJ) WOUT n46
1003 FORMAT (10H PT 0,2X,6F10.3) WOUT n47
WRITE(6,1004)( STALF(I),I=LJ,JJ) WOUT n48
1004 FORMAT (10H ALPHA 0,2X,6F10.3) WOUT n49
WRITE(6,1005)( STSI(I),I=LJ,JJ) WOUT n50
1005 FORMAT (10H I STATOR,2X,6F10.3) WOUT n51
WRITE(6,1006)( STV0(I),I=LJ,JJ) WOUT n52
1006 FORMAT (10H V 0,2X,6F10.3) WOUT n53
WRITE(6,1007)( STVU0(I),I=LJ,JJ) WOUT n54
1007 FORMAT (10H VU 0,2X,6F10.3) WOUT n55
WRITE(6,1008)( STVZ0(I),I=LJ,JJ) WOUT n56
1008 FORMAT (10H VZ 0,2X,6F10.3) WOUT n57
WRITE(6,1009)( STTS0(I),I=LJ,JJ) WOUT n58
1009 FORMAT (10H TS 0,2X,6F10.1) WOUT n59
WRITE(6,1010)( STPS0(I),I=LJ,JJ) WOUT n60
1010 FORMAT (10H PS 0,2X,6F10.3) WOUT n61

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Listing of Code (continued)

1011	WRITE(6,1011)(STDEN0(1),I=LJ,JJ)	WOUT 062
	FORMAT(10H DENS 0,2X,6F10.5)	WOUT 063
	WRITE(6,1012)(STM0(1),I=LJ,JJ)	WOUT 064
1012	FORMAT(10H M 0,2X,6F10.5)	*****
	WRITE(6,1999)(CP0(K),I=LJ,JJ)	*****
1999	FORMAT(10H CP 0,2X,6F10.5)	*****
	WRITE(6,2000)(RV(1,K),I=LJ,JJ)	*****
2000	FORMAT(10H RG 0,2X,6F10.3)	*****
	WRITE(6,2001)(GAM(1,K),I=LJ,JJ)	*****
2001	FORMAT(10H GAMG 0,2X,6F10.5)	*****
	WRITE(6,2002)(HwG(1,K),I=LJ,JJ)	*****
2002	FORMAT(10H HwG 0,2X,6F10.5)	*****
	IF(ISECT.LE.3)GO TO 11013	*****
	WRITE(6,2003)(STWGO(1),I=LJ,JJ),SFLU0	*****
2003	FORMAT(10H WG 0,2X,6F10.5,2X,11H TOTAL FLOW ,F10.5)	*****
11013	WRITE(6,1013)(STDP1(I),I=LJ,JJ)	*****
1013	FORMAT(/5X5HSTA 12X11HSTATOR EXIT/4X6HDIAM 12X,6F10.3)	WOUT 067
	WRITE(6,1014)(STALFE(1),I=LJ,JJ)	WOUT 068
1014	FORMAT(10H ALPHA 1,2X,6F10.3)	WOUT 069
	WRITE(6,1015)(STDELA(1),I=LJ,JJ)	WOUT 070
1015	FORMAT(10H DEL A,2X,6F10.3)	WOUT 071
	WRITE(6,1016)(STV1(1),I=LJ,JJ)	WOUT 072
1016	FORMAT(10H V 1,2X,6F10.3)	WOUT 073
	WRITE(6,1017)(STVU1(1),I=LJ,JJ)	WOUT 074
1017	FORMAT(10H VU 1,2X,6F10.3)	WOUT 075
	WRITE(6,1018)(STVZ1(1),I=LJ,JJ)	WOUT 076
1018	FORMAT(10H VZ 1,2X,6F10.3)	WOUT 077
	WRITE(6,1019)(STTS1(1),I=LJ,JJ)	WOUT 078
1019	FORMAT(10H TS 1,2X,6F10.1)	WOUT 079
	WRITE(6,1064)(STPS1(1),I=LJ,JJ)	WOUT 080
1064	FORMAT(10H PS 1,2X,6F10.3)	WOUT 081
	WRITE(6,1020)(STDEN1(1),I=LJ,JJ)	WOUT 082
1020	FORMAT(10H DENS 1,2X,6F10.5)	WOUT 083
	WRITE(6,1021)(STM1(1),I=LJ,JJ)	WOUT 084
1021	FORMAT(10H M 1,2X,6F10.5)	WOUT 085
	WRITE(6,1022)(ZWINC(1),I=LJ,JJ)	WOUT 086
1022	FORMAT(10H ZWI INC,2X,6F10.5)	*****
	WRITE(6,1026)(CPS(1),I=LJ,JJ)	WOUT 088
1026	FORMAT(10H CP 5,2X,6F10.5)	*****
	WRITE(6,2999)(CP1(K),I=LJ,JJ)	*****
2999	FORMAT(10H CP 1,2X,6F10.5)	*****
	WRITE(6,3000)(RV(2,K),I=LJ,JJ)	*****
3000	FORMAT(10H RG 1,2X,6F10.3)	*****
	WRITE(6,3001)(GAM(2,K),I=LJ,JJ)	*****
3001	FORMAT(10H GAMG 1,2X,6F10.5)	*****
	WRITE(6,3002)(HwG(2,K),I=LJ,JJ)	*****
3002	FORMAT(10H HwG 1,2X,6F10.5)	*****

Listing of Code (continued)

```

      IF (ISECT.LE.3) GO TO 11000
      WRITE(6,3003) ( STWG1(I), I=LJ, JJ), SFL01
3003  FORMAT (10H      WG 1,2X,6F10.5,2X,11HTOTAL FLOW ,F10.5)
11000 WRITE(6,1000) NAME, TITLE, ICASE, ISCASE
      WRITE(6,1028) K, (STDP1A(I), I=LJ, JJ)
1028  FORMAT(4X6HSTA 1A2X11HROTOR INLET,10X5HSTAGE13,1H./3X7HDIAM 1A2X,
16F10.3)
      WRITE(6,1027) (STPTR1(I), I=LJ, JJ)
1027  FORMAT (10H      PTR 1A,2X,6F10.3)
      WRITE(6,1029) (STTTR1(I), I=LJ, JJ)
1029  FORMAT (10H      TTR 1A,2X,6F10.1)
      WRITE(6,1030) (STHET1(I), I=LJ, JJ)
1030  FORMAT (10H      BETA 1A,2X,6F10.3)
      WRITE(6,1031) ( STR1(I), I=LJ, JJ)
1031  FORMAT (10H      I ROTOR,2X,6F10.3)
      WRITE(6,1032) ( STR1A(I), I=LJ, JJ)
1032  FORMAT (10H      R 1A,2X,6F10.3)
      WRITE(6,1033) (STRU1A(I), I=LJ, JJ)
1033  FORMAT (10H      RU 1A,2X,6F10.3)
      WRITE(6,1034) (STMRI1A(I), I=LJ, JJ)
1034  FORMAT (10H      MR 1A,2X,6F10.5)
      WRITE(6,1035) ( STU1A(I), I=LJ, JJ)
1035  FORMAT (10H      U 1A,2X,6F10.3)
      WRITE (6,2035) (STPS1A(I), I=LJ, JJ)
2035  FORMAT (10H      PS 1A,2X,6F10.3)
      WRITE (6,2036) (STTS1A(I), I=LJ, JJ)
2036  FORMAT (10H      TS 1A,2X,6F10.1)
      WRITE(6,3999) (CP1A(K), I=LJ, JJ)
3999  FORMAT(10H      CP 1A,2X,6F10.5)
      WRITE(6,4000) (RV(3,K), I=LJ, JJ)
4000  FORMAT (10H      RG 1A,2X,6F10.3)
      WRITE(6,4001) (GAM(3,K), I=LJ, JJ)
4001  FORMAT (10H      GAMG 1A,2X,6F10.5)
      WRITE (6,4002) (RWG(3,K), I=LJ, JJ)
4002  FORMAT (10H      RWG 1A,2X,6F10.5)
      IF (ISECT.LE.3) GO TO 11037
      WRITE(6,4003) (STWG1A(I), I=LJ, JJ), SFL01A
4003  FORMAT (10H      WG 1A,2X,6F10.5,2X,11HTOTAL FLOW ,F10.5)
11037 WRITE(6,1037) (STDP2(I), I=LJ, JJ)
1037  FORMAT (/5X5HSTA 22X10HROTOR EXIT,4X6HDIAM 22X,6F10.3)
      IF (ISECT.LE.3) GO TO 11036
      WRITE (6,2037) (STPTR2(I), I=LJ, JJ)
2037  FORMAT (10H      PTR 2,2X,6F10.3)
      WRITE (6,2038) (STTTR2(I), I=LJ, JJ)
2038  FORMAT (10H      TTR 2,2X,6F10.1)
11036 WRITE(6,1036) (STHET2(I), I=LJ, JJ)
1036  FORMAT (10H      BETA 2,2X,6F10.3)

```


Listing of Code (continued)

WRITE(6,1056) (STVU2A(I),I=LJ,JJ)	*****
1056 FORMAT (10H VU 2A,2X,6F10.3)	WOUT 147
WRITE(6,1057) (STALF2(I),I=LJ,JJ)	WOUT 148
1057 FORMAT (10H ALPHA 2A,2X,6F10.3)	WOUT 149
WRITE(6,1058) (STMF2A(I),I=LJ,JJ)	*****
1058 FORMAT (10H MF 2A,2X,6F10.5)	WOUT 151
WRITE(6,1059) (STVZ2A(I),I=LJ,JJ)	*****
1059 FORMAT (10H VZ 2A,2X,6F10.3)	WOUT 153
WRITE(6,1060) (STTS2A(I),I=LJ,JJ)	*****
1060 FORMAT (10H TS 2A,2X,6F10.1)	WOUT 155
WRITE(6,1061) (STPS2A(I),I=LJ,JJ)	*****
1061 FORMAT (10H PS 2A,2X,6F10.3)	WOUT 157
WRITE(6,1062) (STDEN2(I),I=LJ,JJ)	WOUT 158
1062 FORMAT (10H DENS 2A,2X,6F10.5)	WOUT 159
WRITE(6,1063) (STM2A(I),I=LJ,JJ)	*****
1063 FORMAT (10H M 2A,2X,6F10.5)	WOUT 161
WRITE(6,5999) (CP2A(K),I=LJ,JJ)	*****
5999 FORMAT(10H CP 2A,2X,6F10.5)	*****
WRITE(6,6000) (RV(5,K),I=LJ,JJ)	*****
6000 FORMAT (10H RG 2A,2X,6F10.3)	*****
WRITE(6,6001) (GAM(5,K),I=LJ,JJ)	*****
6001 FORMAT (10H GAMG 2A,2X,6F10.5)	*****
WRITE (6,6002) (RwG(5,K),I=LJ,JJ)	*****
6002 FORMAT (10H RwG 2A,2X,6F10.5)	*****
IF(ISECT.LE.3)GO TO 21000	*****
WRITE(6,6003) (STwG2A(I),I=LJ,JJ),SFLO2A	*****
6003 FORMAT (10H wG 2A,2X,6F10.5,2X,11HTOTAL FLOW ,F10.5)	*****
21000 IF(SHFLAG) WRITE(6,20000)	*****
20000 FORMAT(1H1,45H AN EXI! HAS BEEN MADE FROM SUBROUTINE WOUT)	*****
RETURN	WOUT 162
END	WOUT 163

Listing of Code (continued)

	SUBROUTINE PHIM(EXI,E[A,TR,PR)	PHIM 001
CPHIM	LOGICAL PHEVER,SRFLAG	PHIM 002
	COMMON SRFLAG	*****
	IF(SRFLAG) WRITE(6,10000)	*****
10000	FORMAT(44H AN ENTRY HAS BEEN MADE IN SUBROUTINE PHIM)	*****
	A = EXI-.5	PHIM 003
	B = -(EXI+(1.-ETA)/2.)	PHIM 004
	C = ETA/2.	PHIM 005
	X = (-B -SQRT(B**2 -4.*A*C))/(2.*A)	PHIM 006
	TR = ETA/(ETA-X)	PHIM 007
	PH = TR*EXI	PHIM 008
	IF(SRFLAG) WRITE(6,20000)	*****
20000	FORMAT(45H AN EXIT HAS BEEN MADE FROM SUBROUTINE PHIM)	*****
	RETURN	PHIM 009
	END	PHIM 010

Keywords: *work engagement, organizational commitment, turnover intentions, organizational citizenship behaviors, organizational identification*

A. Control Cards for FORTRAN Deck Setup

Example

JØB Card	JØB , 10.
Account Number Card	AS77987.
ID Card	ASD1097, TURBIN, 120, 75000, 01.
RUN Card	RUN (P,,,,,, 14000)
LØC Card*	LØC, 75000.
LGØ Card	LGØ .
End-of-Record Card	7/8/9
FORTRAN Deck	PRØGRAM JIM
/ /	/ /
/ /	/ /
/ /	/ /
/ /	/ /
/ /	/ /
/ /	/ /
/ /	/ /
End-of-Record Card	END
End-of-Record Card	7/8/9
End-of-Record Card	7/8/9
Data Deck	FALSE
/	/
/	/
/	/
/	/
/	/
/	/
/	/
End-of-File Card	ENDJØB = 1.0 \$
	6/7/8/9

* The LØC card is required to initialize the core to zero before compilation and execution.

B. Control Cards for Binary Deck Setup

	<u>Example</u>
JØB Card	JØB, 10.
Account Number Card	AS77987.
ID Card	ASD1097, TURBIN, 120, 7500, 01.
LØC Card	LØC, 75000.
LØAD Card	LØAD (INPUT)
EXECUTE Card	EXECUTE.
End-of-Record Card	7/8/9
End-of-Record Card	7/8/9
Binary Deck	Binary Cards
/	/
/	/
/	/
/	/
/	/
End-of-Record Card	7/8/9
End-of-Record Card	7/8/9
Data Deck	Data Cards
/	/
/	/
/	/
/	/
/	/
	ENDJØB = 1.0 \$
End-of-File Card	6/7/8/9